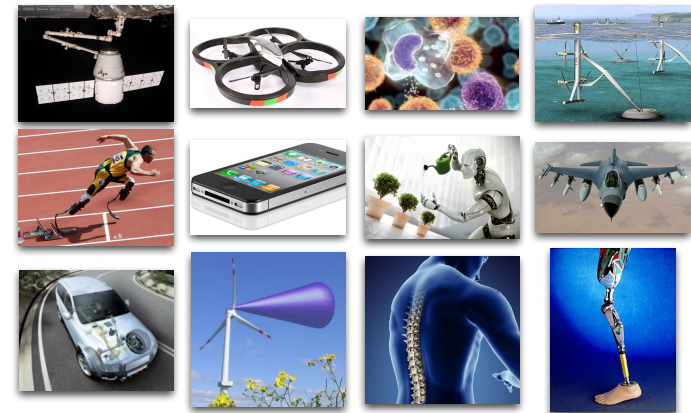


# Control Systems I

Prof. Colin Jones



Make things that **change with time** do what we want them to do

## Control Systems I

### Introduction

Colin Jones

Laboratoire d'Automatique

Make things that **change with time** do what we want them to do



Most engineered systems  
**require controllers to  
function**

Controllers can provide  
**optimal performance**

**Analysis and understanding  
of dynamic systems**

A controller is anything that **senses** the environment, takes **decisions**, and **modifies the environment** in order meet some **objective**.

Robot Quadrotors Perform James Bond Theme

GRASP Lab, University of Pennsylvania

## Components of a Control System

**Sensor** Measure the world

**Actuator** Effect the world

**System** The object we're trying to control

**Controller** Takes decisions based on

- Measurements
- Knowledge of how the system works

A controller is anything that **senses** the environment, takes **decisions**, and **modifies the environment** in order to meet some **objective**.

Note: Controller doesn't have to be a 'computer', or an electronic circuit

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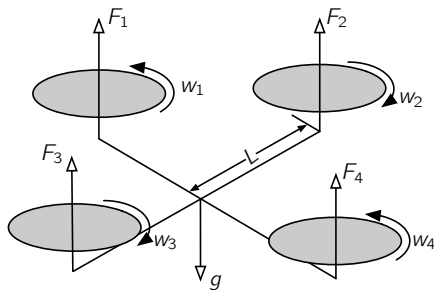
## Example: Autonomous Quadcopter flight

- Highly agile due to fast rotational dynamics
- High thrust-to-weight ratio allows for large translational accelerations
- Motion control by altering rotation rate and/or pitch of the rotors
- High thrust motors enable high performance control



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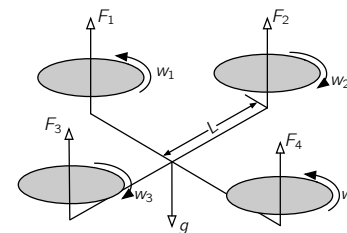
## How a Quad Works



- We can set the speed of the propellers (our inputs)
- Our goal is to control the pitch, roll and altitude

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## How a Quad Works



Force is quadratic in propeller speed:

$$F_i(t) = k_F w_i(t)^2$$

Moment is quadratic in prop speed:

$$M_i(t) = k_M w_i(t)^2$$

Vertical force:

$$F(t) = F_1(t) + F_2(t) + F_3(t) + F_4(t)$$

Roll moment:

$$M_\alpha(t) = L(F_1(t) - F_4(t))$$

Pitch moment:

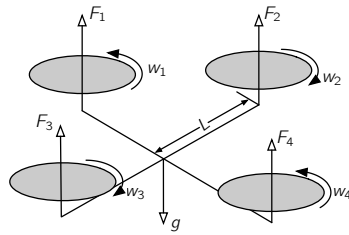
$$M_\beta(t) = L(F_2(t) - F_3(t))$$

Rotation:

$$M_\gamma(t) = M_1(t) + M_2(t) + M_3(t) + M_4(t)$$

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## How a Quad Works



Force is quadratic in propeller speed:

$$F_i(t) = k_F w_i(t)^2$$

Moment is quadratic in prop speed:

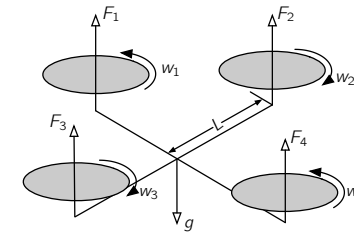
$$M_i(t) = k_M w_i(t)^2$$

$$\begin{array}{l} \text{Vertical force:} \\ \text{Roll moment:} \\ \text{Pitch moment:} \\ \text{Rotation:} \end{array} \begin{pmatrix} F(t) \\ M_\alpha(t) \\ M_\beta(t) \\ M_\gamma(t) \end{pmatrix} = \begin{bmatrix} k_F & k_F & k_F & k_F \\ Lk_F & 0 & 0 & -Lk_F \\ 0 & Lk_F & -Lk_F & 0 \\ k_M & k_M & k_M & k_M \end{bmatrix} \begin{pmatrix} w_1(t)^2 \\ w_2(t)^2 \\ w_3(t)^2 \\ w_4(t)^2 \end{pmatrix}$$

- We have four degrees of freedom and four forces / moments
- Can set the forces / moments as we like - these are our inputs

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## Quad Control



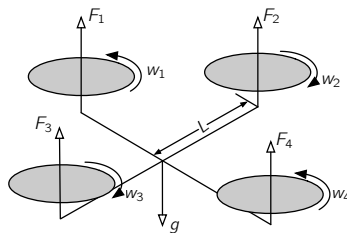
$$\text{Altitude: } m\ddot{z}(t) = \underbrace{-mg}_{\text{Gravity}} + \underbrace{F(t)}_{\text{Thrust of propellers}}$$

$$\text{Hold altitude at } z_c: F(t) = K(z_c - z(t))$$

$$\text{Resulting system: } m\ddot{z}(t) = -mg + K(z_c - z(t))$$

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## Quad Control



Roll and pitch:

$$I_\alpha \ddot{\alpha}(t) = M_\alpha(t)$$

Hold attitude at  $\alpha_c, \beta_c$ :

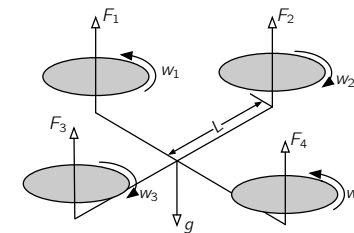
$$M_\alpha(t) = K_\alpha(\alpha_c - \alpha(t))$$

Resulting system

$$I_\alpha \ddot{\alpha}(t) = K_\alpha(\alpha_c - \alpha(t))$$

8

## Quad Control



Yaw:

$$I_\gamma \ddot{\gamma}(t) = M_\gamma(t)$$

Keep yaw at zero:

$$M_\gamma(t) = -K_\gamma \gamma(t) - D_\gamma \dot{\gamma}(t)$$

Resulting system

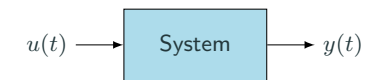
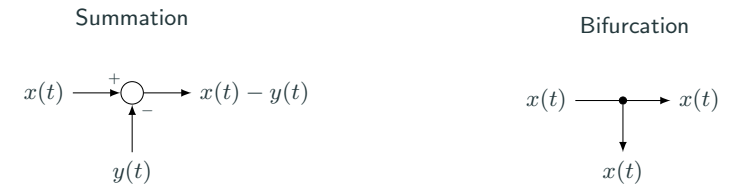
$$I_\gamma \ddot{\gamma}(t) = -K_\gamma \gamma(t) - D_\gamma \dot{\gamma}(t)$$

8

## Example: Autonomous Quadrocopter flight

Demo movie  
Lexus & Kmel robotics

## Block Diagrams - Basic Building Blocks

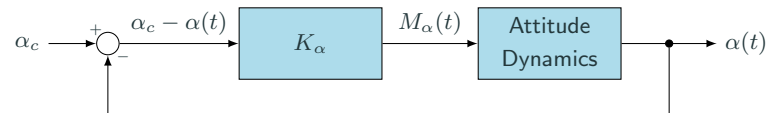


Enforces a dynamic constraint between the output  $y(t)$  and the input  $u(t)$   
e.g.  $\ddot{y}(t) + \alpha \dot{y}(t) - \ddot{u}(t) + u(t) = 0$

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## Block Diagram of Attitude Controllers

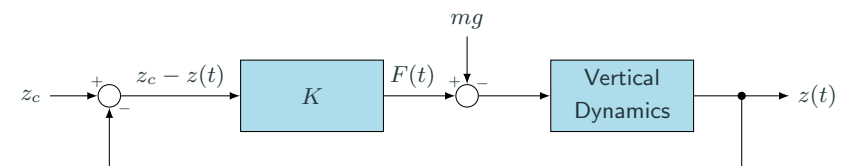


- Reference  $\alpha_c$
- Error  $\alpha_c - \alpha(t)$
- Input  $M_\alpha(t)$
- Output  $\alpha(t)$
- Controller  $K_\alpha$
- System  $I_\alpha \ddot{\alpha}(t) = M_\alpha(t)$

Goal: **Track** reference  $\alpha_c$

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## Block Diagram of Altitude Controller

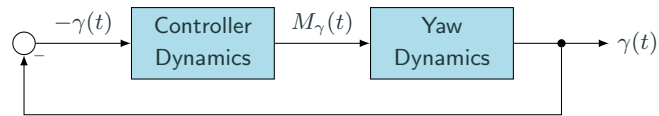


- Disturbance  $g$

Goal: **Track** reference  $z_c$  and **reject** disturbance  $mg$

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## Block Diagram of Yaw Controller

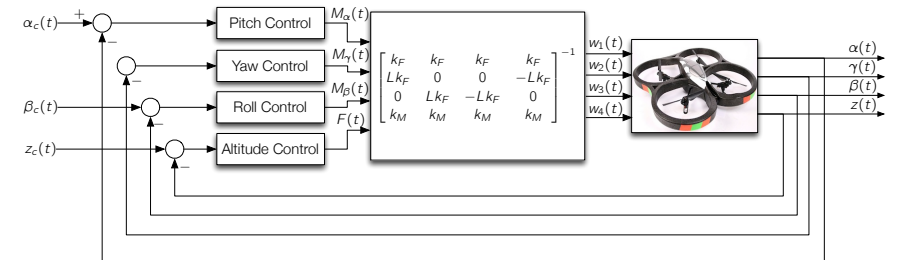


- Controller dynamics:  $M_\gamma(t) = -K_\gamma \gamma(t) - D_\gamma \dot{\gamma}(t)$

Goal: **Regulate** the yaw

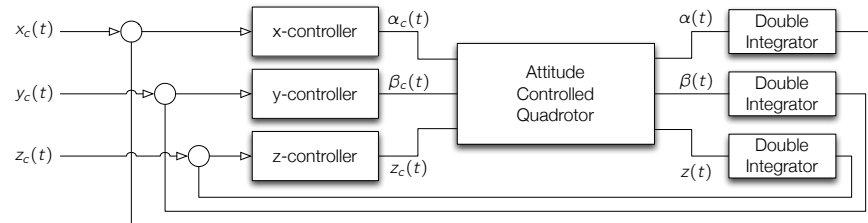
13

## Cascade Control



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## Cascade Control



Possibly lots more loops

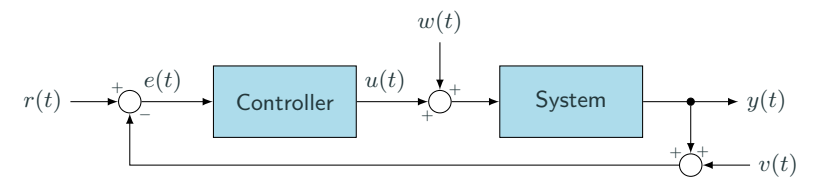
- Collision avoidance
- Trajectory planning
- Mission planning
- etc

Why?

- Inner loops make the system **predictable and simple**
- Conceptually simpler

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## Canonical Block Diagram



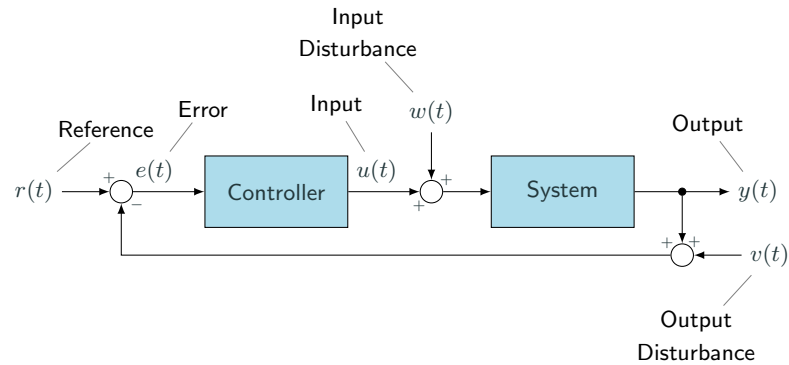
**Goal:** Make  $y(t) = r(t)$ , no matter what  $w(t)$ , or  $v(t)$  are

If  $r(t)$  is...

- zero, we're doing **regulation**
- time-varying, we're doing **servoing / tracking**

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## Canonical Block Diagram



**Goal:** Make  $y(t) = r(t)$ , no matter what  $w(t)$ , or  $v(t)$  are

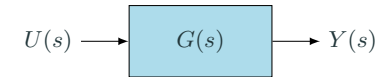
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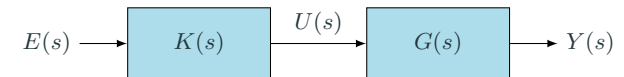
16

## Nomenclature

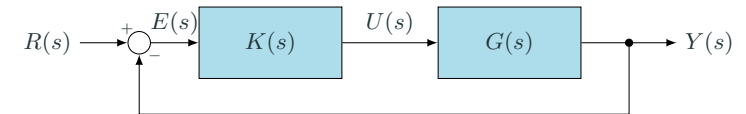
The **system**:



The **open-loop system** or **loop gain**:



The **closed-loop system**:

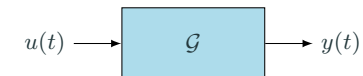


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## What is a System?

A dynamic system transforms an input signal  $u(t)$  into an output signal  $y(t)$ .

$$y = \mathcal{G}(u)$$



We care about LTI systems

**Linear**  $\mathcal{G}(au_1 + bu_2) = a\mathcal{G}(u_1) + b\mathcal{G}(u_2)$

**Causal**  $u(t) = 0$  for  $t < 0$  implies  $y(t) = 0$  for  $t < 0$

**Time-invariant**  $y(t) = \mathcal{G}(u(t))$  implies that  $\mathcal{G}(u(t+T)) = y(t+T)$

## Quick Review of Systèmes Dynamique

More complete review on Moodle

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## Why are these types of systems important?

1. We can predict their behaviour from data easily

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## 1. We can predict their behaviour from data easily

### Impulse Response

The impulse response  $g(t)$  is defined as the output of the system in response to a dirac delta function at time  $t = 0$ :

$$g(t) := \mathcal{G}(\delta(t))$$

20

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The impulse response  $g(t)$  is defined as the output of the system in response to a dirac delta function at time  $t = 0$ :

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### Theorem : Response of an LTI System

The output of an LTI system in response to an input signal  $u(t)$  is

$$\mathcal{G}(u) = g * u$$

where  $y = g * u$  if

$$y(t) = \int_0^t u(\tau)g(t - \tau)d\tau$$

20

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If  $\mathcal{G}$  is an LTI system, then the impulse response completely characterizes it.

Key limitation: Most systems have an infinitely-long impulse response.

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## Why are these types of systems important?

1. We can predict their behaviour from data easily
2. We can store and manipulate complex systems

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## 2. We can store and manipulate complex systems

### Transfer Function

The **transfer function** of a system is the Laplace transform of its impulse response.

$$\mathcal{L}\{g(t)\} = G(s)$$

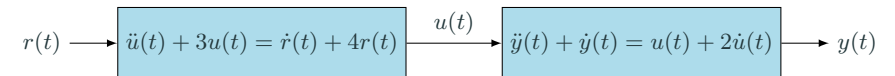
For LTI systems  $G(s)$  is a rational polynomial function

**The point:** Convolution becomes multiplication

$$y = g * u \quad \Leftrightarrow \quad Y(s) = G(s)U(s)$$

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## Manipulation of Simple Block Diagrams

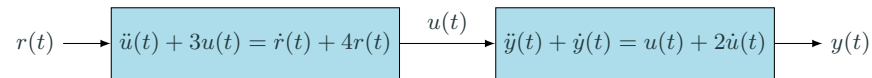


If we're given the reference function  $r(t)$ , what is  $y(t)$ ?

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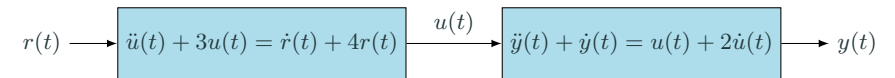
## Manipulation of Simple Block Diagrams



$$\begin{aligned}\ddot{u}(t) + 3u(t) &= \dot{r}(t) + 4r(t) &\Rightarrow & s^2U(s) + 3U(s) = sR(s) + 4R(s) \\ \ddot{y}(t) + \dot{y}(t) &= u(t) + 2\dot{u}(t) &\Rightarrow & s^2Y(s) + sY(s) = U(s) + 2sU(s)\end{aligned}$$

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## Manipulation of Simple Block Diagrams



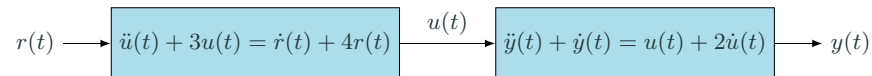
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Re-arranging gives:

$$U(s) = \frac{s+4}{s^2+3}R(s) \qquad Y(s) = \frac{1+2s}{s^2+s}U(s)$$

23

## Manipulation of Simple Block Diagrams



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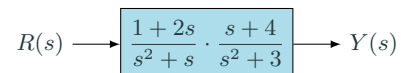
Re-arranging gives:

$$U(s) = \frac{s+4}{s^2+3}R(s) \qquad Y(s) = \frac{1+2s}{s^2+s}U(s)$$

... and we can compute the impact of  $r(t)$  on  $y(t)$

$$Y(s) = \frac{1+2s}{s^2+s} \cdot \frac{s+4}{s^2+3}R(s)$$

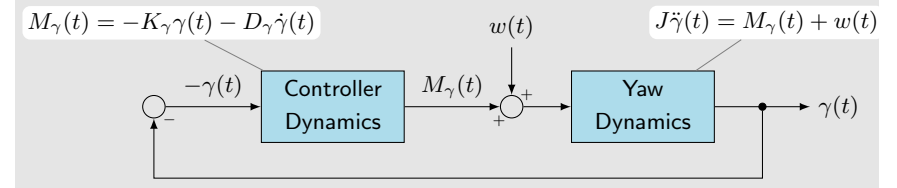
Series connection of blocks (convolution) becomes multiplication!



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## Example: System Response

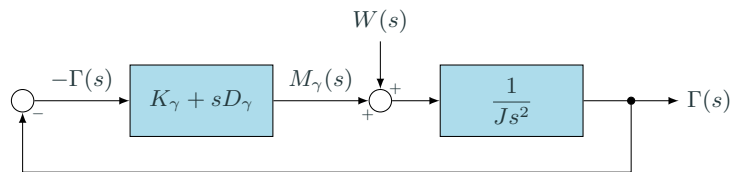
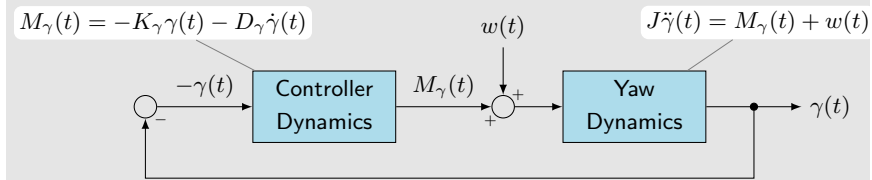
Compute response to a impulsive disturbance acting on the yaw system



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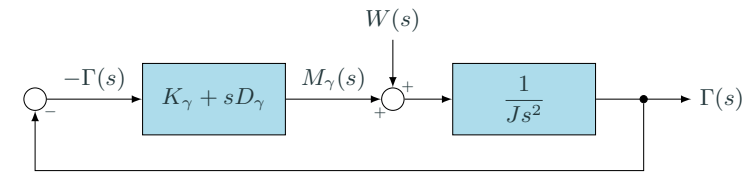
## Example: System Response

Compute response to a impulsive disturbance acting on the yaw system



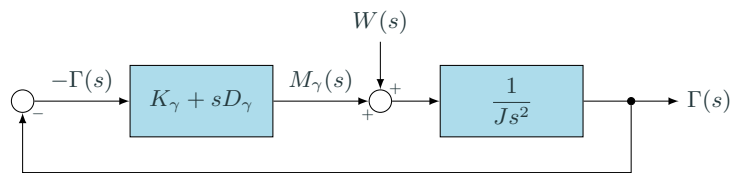
24

## Example: System Response



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## Example: System Response



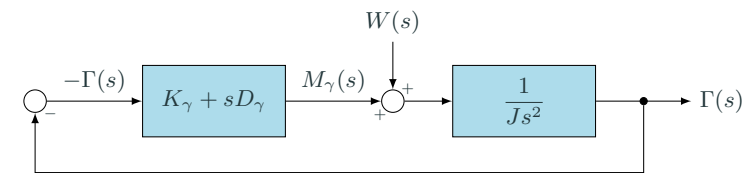
Start at the output and work backwards against the arrows

$$\Gamma = \frac{1}{Js^2}(W - (K_\gamma + sD_\gamma)\Gamma)$$

$$(Js^2 + sD_\gamma + K_\gamma)\Gamma = W$$

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## Example: System Response



Start at the output and work backwards against the arrows

$$\Gamma = \frac{1}{Js^2}(W - (K_\gamma + sD_\gamma)\Gamma)$$

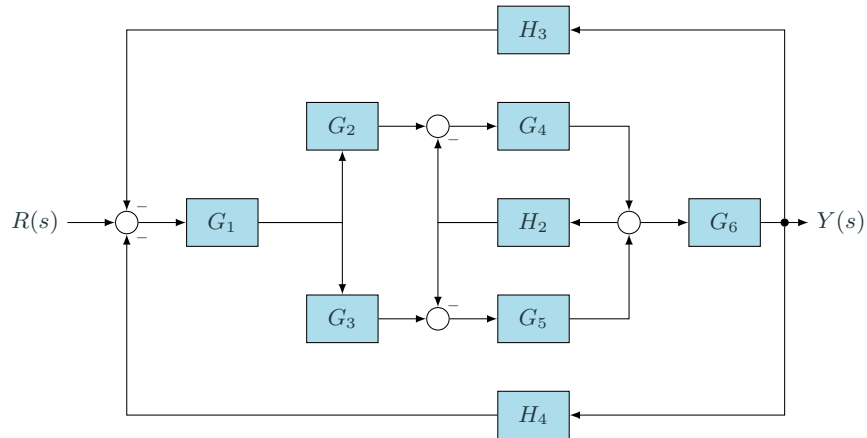
$$(Js^2 + sD_\gamma + K_\gamma)\Gamma = W$$

$$W(s) \longrightarrow \boxed{\frac{1}{Js^2 + Ds + K}} \longrightarrow \Gamma(s)$$

Where we recall that  $D$  sets the damping and  $K$  the response rate.

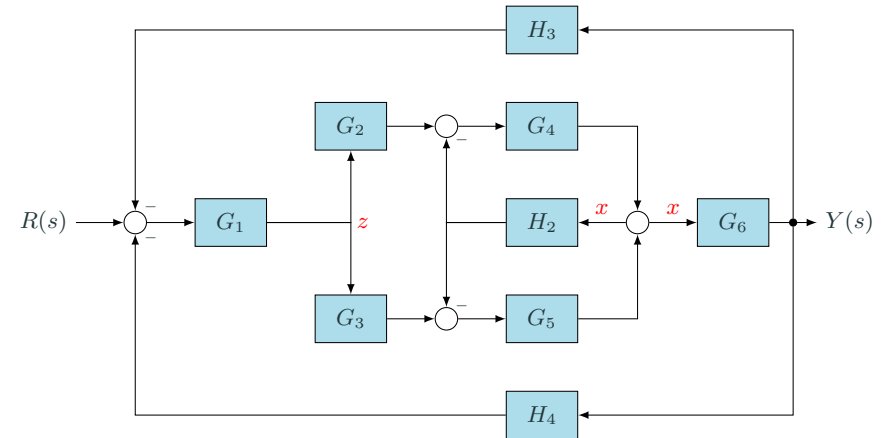
25

### Example: Complex System



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### Example: Complex System



Add auxiliary variables for internal loops, and wherever convenient to simplify.

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### Example: Complex System

Start at the output and work back **against** the arrows.

A block is a multiplication, a summation is addition.

$$Y = G_6 x$$

$$x = G_4(G_2 z - H_2 x) + G_5(G_3 z - H_2 x)$$

$$z = G_1(R - H_3 Y - H_4 Y)$$

Solve for  $Y$  as a function of  $R$

$$x = (G_4 G_2 + G_5 G_3)z - (G_4 H_2 + G_5 H_2)x$$

$$(1 + G_4 H_2 + G_5 H_2)x = (G_4 G_2 + G_5 G_3)z$$

$$x = \frac{G_4 G_2 + G_5 G_3}{1 + G_4 H_2 + G_5 H_2} z$$

$$Y = G_6 \frac{G_4 G_2 + G_5 G_3}{1 + G_4 H_2 + G_5 H_2} z$$

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### Example: Complex System

$$Y = G_6 \underbrace{\frac{G_4 G_2 + G_5 G_3}{1 + G_4 H_2 + G_5 H_2}}_Q z \quad z = G_1 R - (H_3 + H_4)Y$$

Solve to get the transfer function

$$\frac{Y}{R} = \frac{Q G_1}{1 + Q(H_3 + H_4)}$$

If we want to do more algebra, we can eliminate  $Q$

$$\frac{Y}{R} = \frac{G_1 G_2 G_4 G_6 + G_1 G_3 G_5 G_6}{(G_4 + G_5)H_2 + G_2 G_4 G_6 H_3 + G_2 G_4 G_6 H_4 + G_3 G_5 G_6 H_3 + G_3 G_5 G_6 H_4 + 1}$$

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### Why are these types of systems important?

1. We can predict their behaviour from data easily
2. We can store and manipulate complex systems
3. We can shape system behaviour

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### 3. We can shape system behaviours

Time domain

- PID
- Model predictive control
- ...

30

### 3. We can shape system behaviours

Time domain

- PID
- Model predictive control
- ...

Frequency domain

- Loopshaping controllers
- $\mathcal{H}_\infty$  - robust optimal control
- ...

30

### 3. We can shape system behaviours

Time domain

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Frequency domain

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- ...

Pole/zero domain

- Pole placement
- Linear quadratic regulation
- ...

30

### 3. We can shape system behaviours

Time domain

- PID
- Model predictive control
- ...

Frequency domain

- Loopshaping controllers
- $\mathcal{H}_\infty$  - robust optimal control
- ...

Pole/zero domain

- Pole placement
- Linear quadratic regulation
- ...

Many very well-established techniques that are proven and work well at large scales.

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### Key points to review

Please review:

- Computation of Laplace transforms
- Manipulation of block diagrams
- Inverse Laplace transforms
- System response to impulse, step, ramp, etc

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### Administration

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### Teachers



#### Professor

Colin Jones  
Laboratoire d'Automatique  
ME C2 405  
colin.jones@epfl.ch



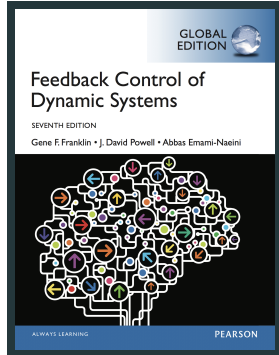
#### Travaux Pratique

Christophe Salzmann  
Laboratoire d'Automatique  
ME C2 426  
christophe.salzmann@epfl.ch

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## Reference Material

We will mostly follow the textbook:



- The sections of the text that we are covering will appear on Moodle
- Lecture notes and pre-recorded videos are on Moodle

You are responsible for the material in the text **and** in the lecture notes

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## Activities

1. Lectures
  - Two hours per week
  - Lectures are not recorded, but high-quality pre-recordings are on Moodle
2. TPs
  - Seven TPs done via a MOOC interface driving a physical device
  - Can do the TPs in-person **or** remotely
3. Exercises
  - Written / computer exercises
  - 13 exercise sets

Detailed schedule on Moodle

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## How to Get Help

**In person** During lectures, or during afternoon exercise / TP sessions

**Ed Discussion** Please post your questions publicly - others will benefit!

**Recorded videos** Lectures have been pre-recorded and are available on Moodle.

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## Grading and Exams

100% Final exam

- One question from the TPs (MOOC) worth 20%
- Questions based on the lectures / exercises worth 80%

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## Employee Scheduling : The Challenge

Too few salespeople  
=  
Unhappy customers / less sales

Too many salespeople  
=  
Excessive wages

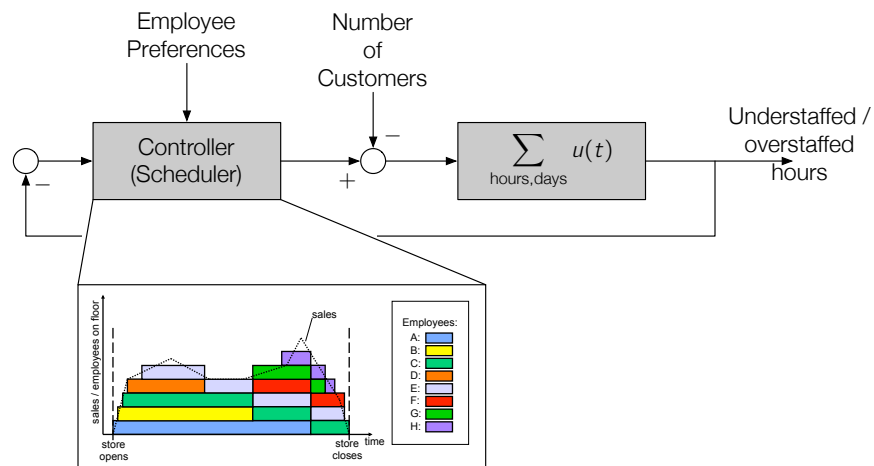


### Examples: Other Varieties of Control

What can control do?

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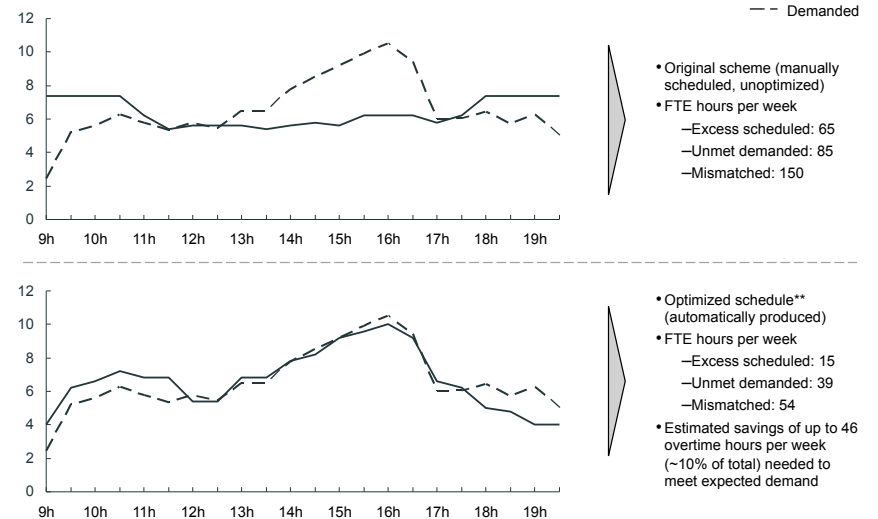
## The Control Problem



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### ... ACHIEVES SAVINGS BY MATCHING RESOURCES TO DEMAND

Average number of weekday staff\*



\* For retail store with 14 staff (11.5 FTEs)

\*\* Sample optimized schedule provided by Apex Optimization GmbH

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## Example: 'Fulfilment Centers'



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## Inerter

The Inerter in F1 Racing  
Slides from Prof. Malcom Smith

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## Demand Response

Demand Response Slides

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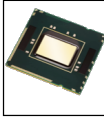





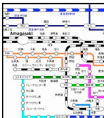
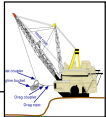
Kiva systems

Sold to Amazon in March, 2012 for \$775m USD

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## Control Applications at all Space and Time Scales

	Computer control	ns		
		$\mu$ s	Power systems	
	Traction control	ms		
		Seconds	Buildings	
	Refineries	Minutes		
		Hours	Nurse rostering	
	Train scheduling	Days		
		Weeks	Production planning	

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## Summary

- Feedback control is everywhere
- It is used to:
  - Stabilize unstable systems
  - Make behaviors repeatable / predictable
  - Maximize performance
  - Understand what complex systems are doing

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