

Control Systems I

Prof. Colin Jones

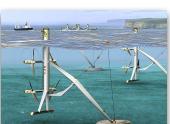


Make things that **change with time** do what we want them to do

Make things that **change with time** do what we want them to do



Most engineered systems
**require controllers to
function**



Controllers can provide
optimal performance



Analysis and understanding
of dynamic systems

A controller is anything that **senses** the environment, takes **decisions**, and **modifies** the environment in order to meet some **objective**.

Robot Quadrotors Perform James Bond Theme

GRASP Lab, University of Pennsylvania

Components of a Control System

Sensor Measure the world

Actuator Effect the world

System The object we're trying to control

Controller Takes decisions based on

- Measurements
- Knowledge of how the system works

A controller is anything that **senses** the environment, takes **decisions**, and **modifies the environment** in order meet some **objective**.

Note: Controller doesn't have to be a 'computer', or an electronic circuit

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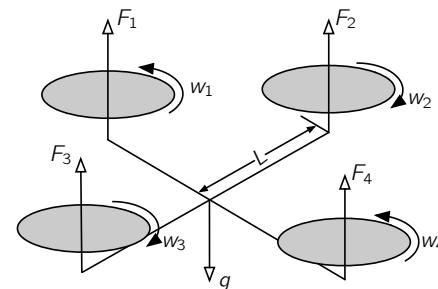
Example: Autonomous Quadrocopter flight

- Highly agile due to fast rotational dynamics
- High thrust-to-weight ratio allows for large translational accelerations
- Motion control by altering rotation rate and/or pitch of the rotors
- High thrust motors enable high performance control



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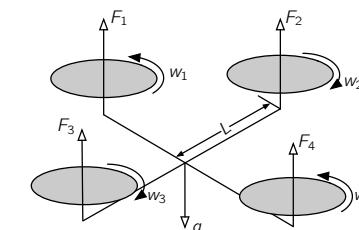
How a Quad Works



- We can set the speed of the propellers (our inputs)
- Our goal is to control the pitch, roll and altitude

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How a Quad Works



Force is quadratic in propeller speed:

$$F_i(t) = k_F w_i(t)^2$$

Moment is quadratic in prop speed:

$$M_i(t) = k_M w_i(t)^2$$

Vertical force:

$$F(t) = F_1(t) + F_2(t) + F_3(t) + F_4(t)$$

Roll moment:

$$M_\alpha(t) = L(F_1(t) - F_4(t))$$

Pitch moment:

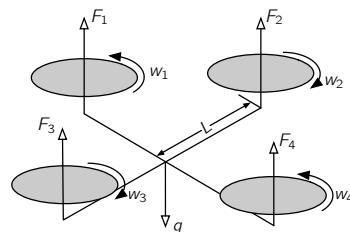
$$M_\beta(t) = L(F_2(t) - F_3(t))$$

Rotation:

$$M_\gamma(t) = M_1(t) + M_2(t) + M_3(t) + M_4(t)$$

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How a Quad Works



Force is quadratic in propeller speed:

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Moment is quadratic in prop speed:

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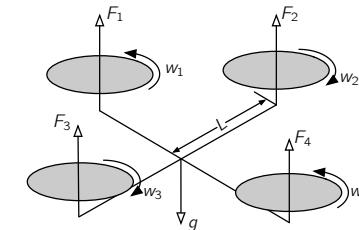
Vertical force:

$$\begin{aligned} \text{Roll moment: } & \begin{pmatrix} F(t) \\ M_\alpha(t) \end{pmatrix} = \begin{bmatrix} k_F & k_F & k_F & k_F \\ Lk_F & 0 & 0 & -Lk_F \end{bmatrix} \begin{pmatrix} w_1(t)^2 \\ w_2(t)^2 \\ w_3(t)^2 \\ w_4(t)^2 \end{pmatrix} \\ \text{Pitch moment: } & \begin{pmatrix} M_\beta(t) \\ M_\gamma(t) \end{pmatrix} = \begin{bmatrix} 0 & Lk_F & -Lk_F & 0 \\ k_M & k_M & k_M & k_M \end{bmatrix} \begin{pmatrix} w_1(t)^2 \\ w_2(t)^2 \\ w_3(t)^2 \\ w_4(t)^2 \end{pmatrix} \\ \text{Rotation: } & \end{aligned}$$

- We have four degrees of freedom and four forces / moments
- Can set the forces / moments as we like - these are our inputs

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Quad Control

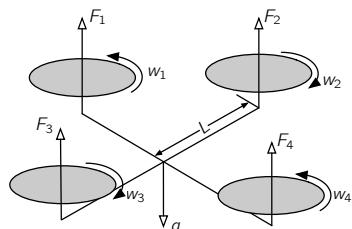


$$\text{Altitude: } m\ddot{z}(t) = \underbrace{-mg}_{\text{Gravity}} + \underbrace{F(t)}_{\text{Thrust of propellers}}$$

$$\text{Hold altitude at } z_c: F(t) = K(z_c - z(t))$$

$$\text{Resulting system: } m\ddot{z}(t) = -mg + K(z_c - z(t))$$

Quad Control



Roll and pitch:

$$I_\alpha \ddot{\alpha}(t) = M_\alpha(t)$$

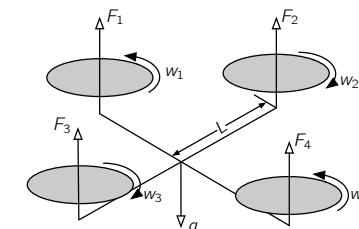
Hold attitude at α_c, β_c :

$$M_\alpha(t) = K_\alpha(\alpha_c - \alpha(t))$$

Resulting system

$$I_\alpha \ddot{\alpha}(t) = K_\alpha(\alpha_c - \alpha(t))$$

Quad Control



Yaw:

$$I_\gamma \ddot{\gamma}(t) = M_\gamma(t)$$

Keep yaw at zero:

$$M_\gamma(t) = -K_\gamma \gamma(t) - D_\gamma \dot{\gamma}(t)$$

Resulting system

$$I_\gamma \ddot{\gamma}(t) = -K_\gamma \gamma(t) - D_\gamma \dot{\gamma}(t)$$

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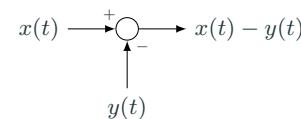
Example: Autonomous Quadrocopter flight

Demo movie

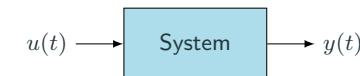
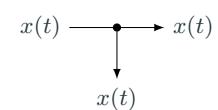
Lexus & Kmel robotics

Block Diagrams - Basic Building Blocks

Summation



Bifurcation

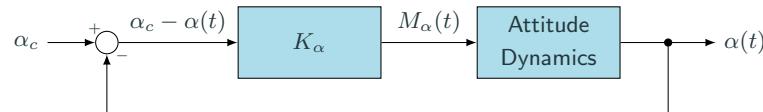


Enforces a dynamic constraint between the output $y(t)$ and the input $u(t)$
e.g. $\ddot{y}(t) + \alpha\dot{y}(t) - \ddot{u}(t) + u(t) = 0$

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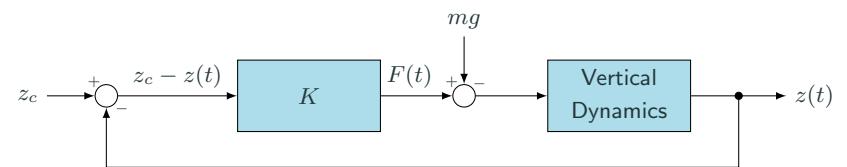
Block Diagram of Attitude Controllers



- Reference α_c
- Error $\alpha_c - \alpha(t)$
- Input $M_\alpha(t)$
- Output $\alpha(t)$
- Controller K_α
- System $I_\alpha \ddot{\alpha}(t) = M_\alpha(t)$

Goal: **Track** reference α_c

Block Diagram of Altitude Controller



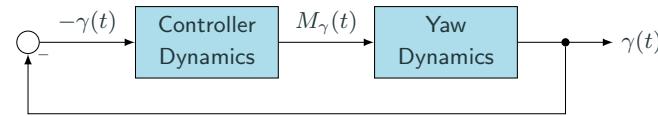
- Disturbance g

Goal: **Track** reference z_c and **reject** disturbance mg

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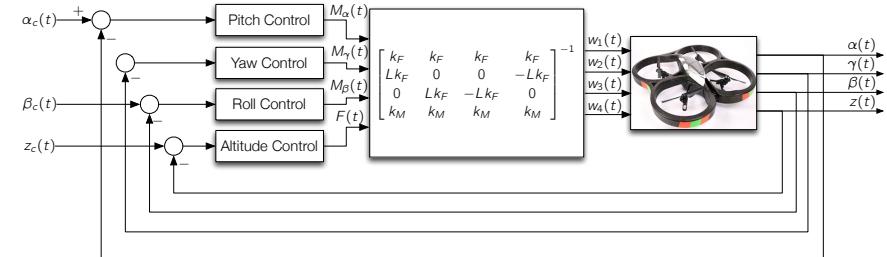
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Block Diagram of Yaw Controller



- Controller dynamics: $M_\gamma(t) = -K_\gamma\gamma(t) - D_\gamma\dot{\gamma}(t)$

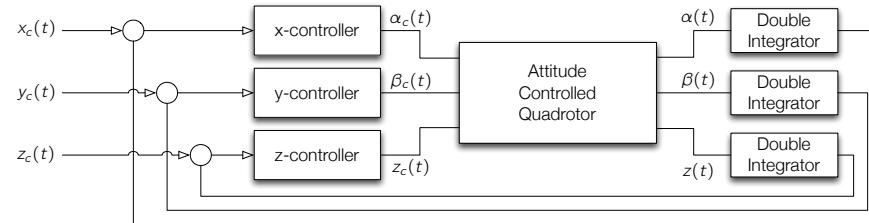
Cascade Control



Goal: **Regulate** the yaw

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Cascade Control



Possibly lots more loops

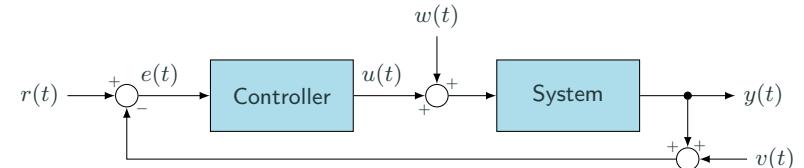
- Collision avoidance
- Trajectory planning
- Mission planning
- etc

Why?

- Inner loops make the system **predictable** and **simple**
- Conceptually simpler

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Canonical Block Diagram



Goal: Make $y(t) = r(t)$, no matter what $w(t)$, or $v(t)$ are

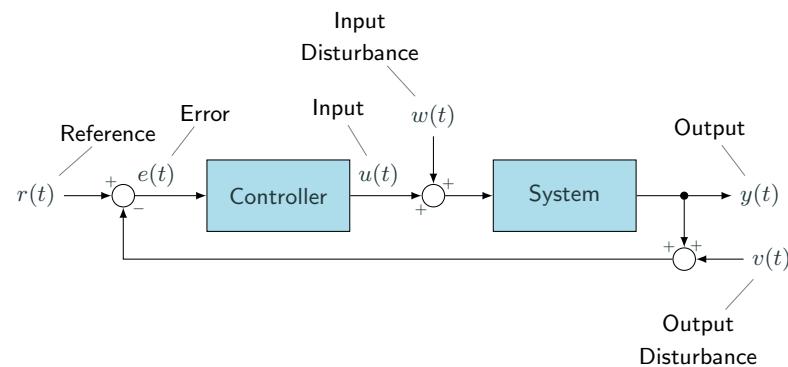
If $r(t)$ is...

- zero, we're doing **regulation**
- time-varying, we're doing **servoing / tracking**

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Canonical Block Diagram



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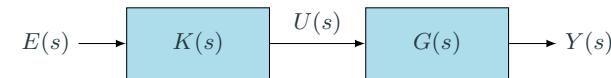
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Nomenclature

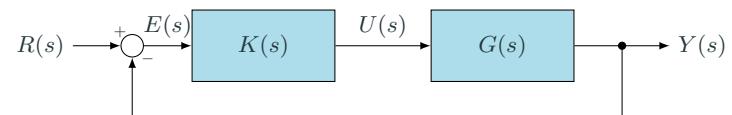
The **system**:



The **open-loop system or loop gain**:



The **closed-loop system**:



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What is a System?

A dynamic system transforms an input signal $u(t)$ into an output signal $y(t)$.

$$y = \mathcal{G}(u)$$



Quick Review of Systèmes Dynamique

More complete review on Moodle

We care about LTI systems

Linear $\mathcal{G}(au_1 + bu_2) = a\mathcal{G}(u_1) + b\mathcal{G}(u_2)$

Causal $u(t) = 0$ for $t < 0$ implies $y(t) = 0$ for $t < 0$

Time-invariant $y(t) = \mathcal{G}(u(t))$ implies that $\mathcal{G}(u(t+T)) = y(t+T)$

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Why are these types of systems important?

1. We can predict their behaviour from data easily

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1. We can predict their behaviour from data easily

Impulse Response

The impulse response $g(t)$ is defined as the output of the system in response to a dirac delta function at time $t = 0$:

$$g(t) := \mathcal{G}(\delta(t))$$

1. We can predict their behaviour from data easily

Impulse Response

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Theorem : Response of an LTI System

The output of an LTI system in response to an input signal $u(t)$ is

$$\mathcal{G}(u) = g * u$$

where $y = g * u$ if

$$y(t) = \int_0^t u(\tau)g(t - \tau)d\tau$$

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20

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If \mathcal{G} is an LTI system, then the impulse response completely characterizes it.

Key limitation: Most systems have an infinitely-long impulse response.

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2. We can store and manipulate complex systems

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Transfer Function

The **transfer function** of a system is the Laplace transform of its impulse response.

$$\mathcal{L}\{g(t)\} = G(s)$$

For LTI systems $G(s)$ is a rational polynomial function

The point: Convolution becomes multiplication

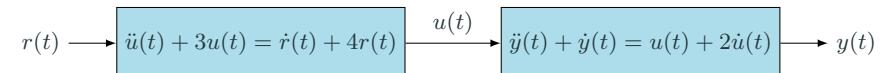
$$y = g * u$$

\Leftrightarrow

$$Y(s) = G(s)U(s)$$

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Manipulation of Simple Block Diagrams

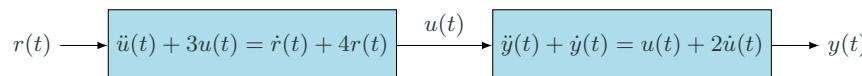


If we're given the reference function $r(t)$, what is $y(t)$?

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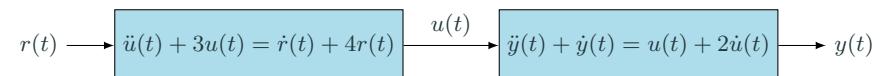
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Manipulation of Simple Block Diagrams



$$\begin{aligned} \ddot{u}(t) + 3u(t) = \dot{r}(t) + 4r(t) &\Rightarrow s^2U(s) + 3U(s) = sR(s) + 4R(s) \\ \ddot{y}(t) + \dot{y}(t) = u(t) + 2\dot{u}(t) &\Rightarrow s^2Y(s) + sY(s) = U(s) + 2sU(s) \end{aligned}$$

Manipulation of Simple Block Diagrams



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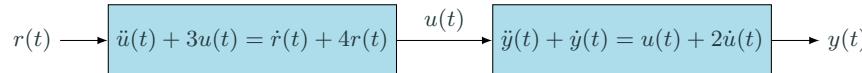
Re-arranging gives:

$$U(s) = \frac{s+4}{s^2+3} R(s) \quad Y(s) = \frac{1+2s}{s^2+s} U(s)$$

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Manipulation of Simple Block Diagrams



$$\begin{aligned} \ddot{u}(t) + 3u(t) = \dot{r}(t) + 4r(t) &\Rightarrow s^2U(s) + 3U(s) = sR(s) + 4R(s) \\ \ddot{y}(t) + \dot{y}(t) = u(t) + 2\dot{u}(t) &\Rightarrow s^2Y(s) + sY(s) = U(s) + 2sU(s) \end{aligned}$$

Re-arranging gives:

$$U(s) = \frac{s+4}{s^2+3} R(s) \quad Y(s) = \frac{1+2s}{s^2+s} U(s)$$

... and we can compute the impact of $r(t)$ on $y(t)$

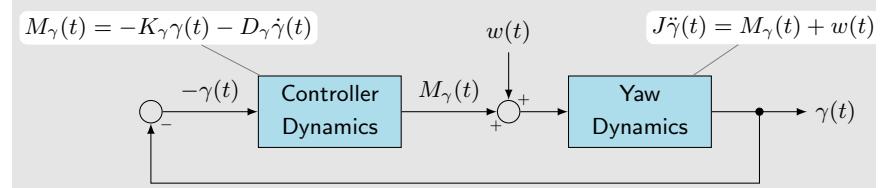
$$Y(s) = \frac{1+2s}{s^2+s} \cdot \frac{s+4}{s^2+3} R(s)$$

Series connection of blocks (convolution) becomes multiplication!

$$R(s) \longrightarrow \frac{1+2s}{s^2+s} \cdot \frac{s+4}{s^2+3} \longrightarrow Y(s)$$

Example: System Response

Compute response to a impulsive disturbance acting on the yaw system

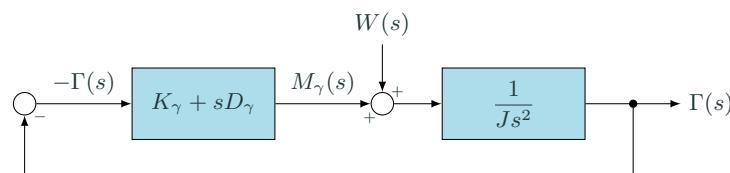
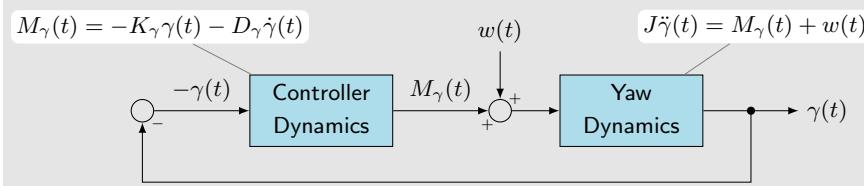


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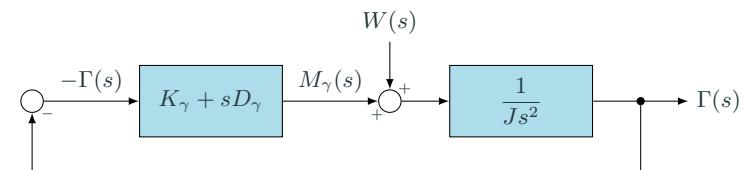
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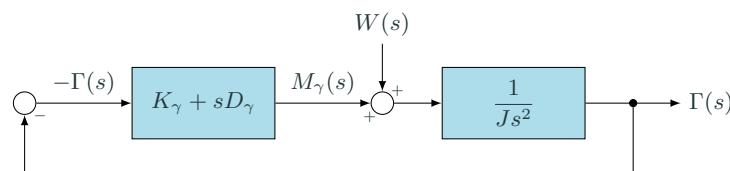
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Example: System Response



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Example: System Response

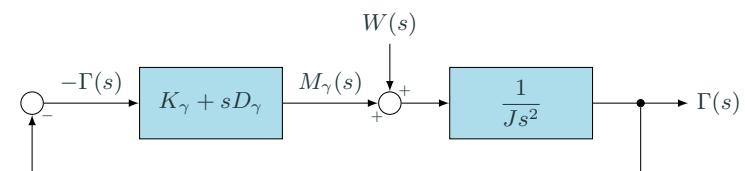


Start at the output and work backwards against the arrows

$$\begin{aligned} \Gamma &= \frac{1}{Js^2}(W - (K_\gamma + sD_\gamma)\Gamma) \\ (Js^2 + sD_\gamma + K_\gamma)\Gamma &= W \end{aligned}$$

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Example: System Response



Start at the output and work backwards against the arrows

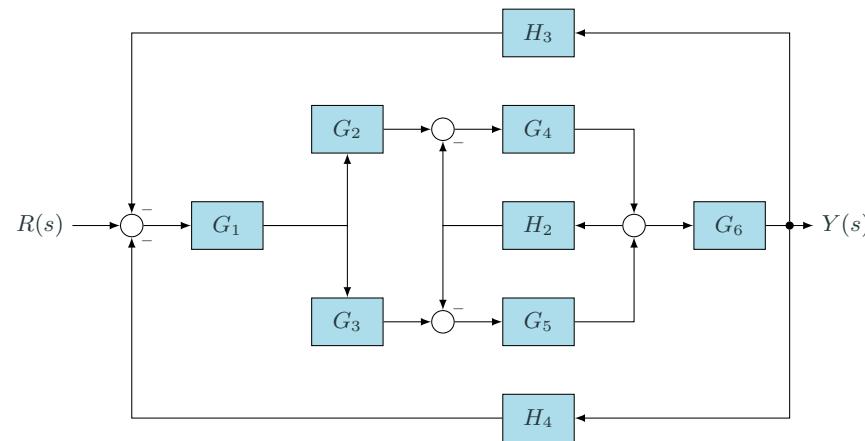
$$\begin{aligned} \Gamma &= \frac{1}{Js^2}(W - (K_\gamma + sD_\gamma)\Gamma) \\ (Js^2 + sD_\gamma + K_\gamma)\Gamma &= W \end{aligned}$$

$$W(s) \longrightarrow \frac{1}{Js^2 + Ds + K} \longrightarrow \Gamma(s)$$

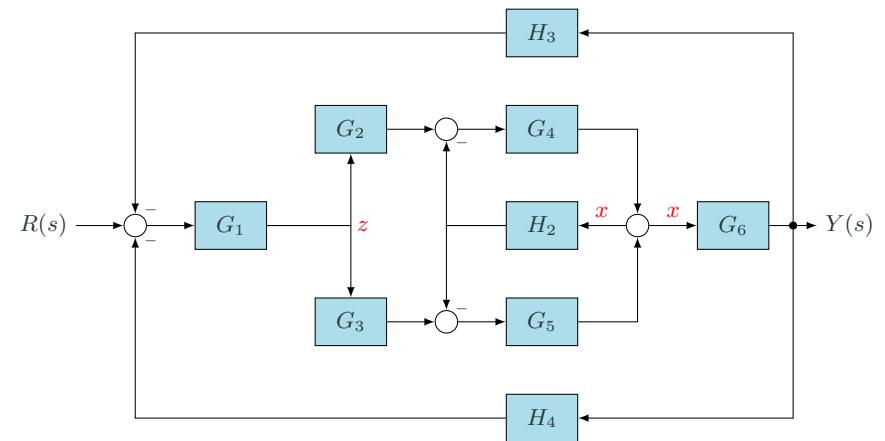
Where we recall that D sets the damping and K the response rate.

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Example: Complex System



Example: Complex System



Add auxiliary variables for internal loops, and wherever convenient to simplify.

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Example: Complex System

Start at the output and work back **against** the arrows.

A block is a multiplication, a summation is addition.

$$Y = G_6 x$$

$$x = G_4(G_2 z - H_2 x) + G_5(G_3 z - H_2 x)$$

$$z = G_1(R - H_3 Y - H_4 Y)$$

Solve for Y as a function of R

$$x = (G_4 G_2 + G_5 G_3)z - (G_4 H_2 + G_5 H_2)x$$

$$(1 + G_4 H_2 + G_5 H_2)x = (G_4 G_2 + G_5 G_3)z$$

$$x = \frac{G_4 G_2 + G_5 G_3}{1 + G_4 H_2 + G_5 H_2} z$$

$$Y = G_6 \frac{G_4 G_2 + G_5 G_3}{1 + G_4 H_2 + G_5 H_2} z$$

Example: Complex System

$$Y = G_6 \underbrace{\frac{G_4 G_2 + G_5 G_3}{1 + G_4 H_2 + G_5 H_2}}_Q z$$

$$z = G_1 R - (H_3 + H_4)Y$$

Solve to get the transfer function

$$\frac{Y}{R} = \frac{Q G_1}{1 + Q(H_3 + H_4)}$$

If we want to do more algebra, we can eliminate Q

$$\frac{Y}{R} = \frac{G_1 G_2 G_4 G_6 + G_1 G_3 G_5 G_6}{(G_4 + G_5)H_2 + G_2 G_4 G_6 H_3 + G_2 G_4 G_6 H_4 + G_3 G_5 G_6 H_3 + G_3 G_5 G_6 H_4 + 1}$$

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Why are these types of systems important?

1. We can predict their behaviour from data easily
2. We can store and manipulate complex systems
3. We can shape system behaviour

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Time domain

- PID
- Model predictive control
- ...

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3. We can shape system behaviours

Time domain

- PID
- Model predictive control
- ...

Frequency domain

- Loopshaping controllers
- \mathcal{H}_∞ - robust optimal control
- ...

3. We can shape system behaviours

Time domain

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Frequency domain

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Pole/zero domain

- Pole placement
- Linear quadratic regulation
- ...

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3. We can shape system behaviours

Time domain

- PID
- Model predictive control
- ...

Frequency domain

- Loopshaping controllers
- \mathcal{H}_∞ - robust optimal control
- ...

Pole/zero domain

- Pole placement
- Linear quadratic regulation
- ...

Many very well-established techniques that are proven and work well at large scales.

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Key points to review

Please review:

- Computation of Laplace transforms
- Manipulation of block diagrams
- Inverse Laplace transforms
- System response to impulse, step, ramp, etc

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Administration

Teachers



Professor
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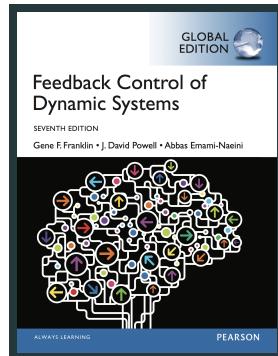


Travaux Pratique
Christophe Salzmann
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christophe.salzmann@epfl.ch

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Reference Material

We will mostly follow the textbook:



- The sections of the text that we are covering will appear on Moodle
- Lecture notes and pre-recorded videos are on Moodle

You are responsible for the material in the text **and** in the lecture notes

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Activities

1. Lectures
 - Two hours per week
 - Lectures are not recorded, but high-quality pre-recordings are on Moodle
2. TPs
 - Seven TPs done via a MOOC interface driving a physical device
 - Can do the TPs in-person **or** remotely
3. Exercises
 - Written / computer exercises
 - 13 exercise sets

[Detailed schedule on Moodle](#)

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How to Get Help

In person During lectures, or during afternoon exercise / TP sessions

Ed Discussion Please post your questions publicly - others will benefit!

Recorded videos Lectures have been pre-recorded and are available on Moodle.

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Grading and Exams

100% Final exam

- One question from the TPs (MOOC) worth 20%
- Questions based on the lectures / exercises worth 80%

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Employee Scheduling : The Challenge

Examples: Other Varieties of Control

Too few salespeople

=
Unhappy customers / less sales



Too many salespeople

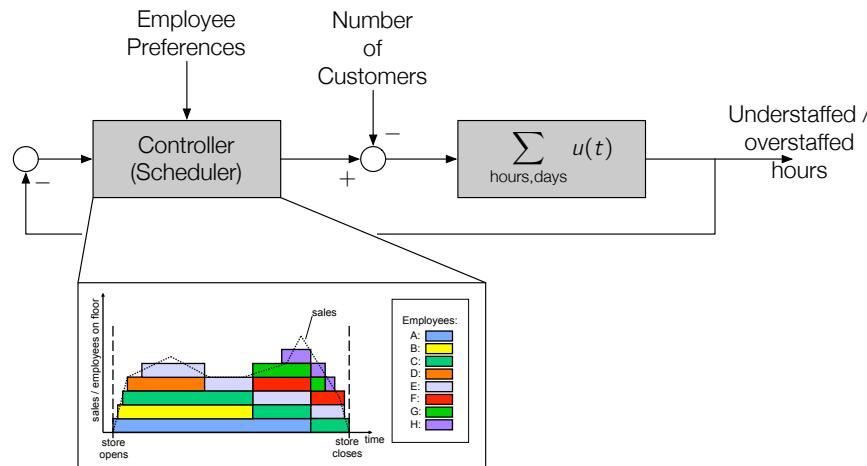
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Excessive wages



What can control do?

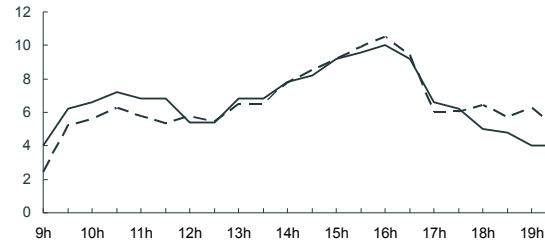
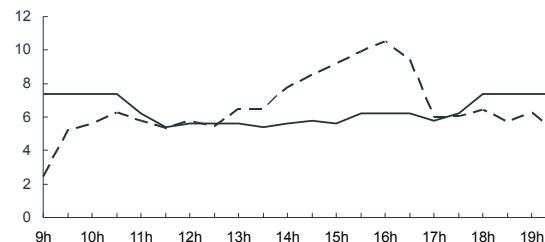
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The Control Problem



... ACHIEVES SAVINGS BY MATCHING RESOURCES TO DEMAND

Average number of weekday staff*



* For retail store with 14 staff (11.5 FTEs)

** Sample optimized schedule provided by Apex Optimization GmbH

Example: 'Fulfilment Centers'



Kiva systems

Sold to Amazon in March, 2012 for \$775m USD

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Inerter

The Inerter in F1 Racing
Slides from Prof. Malcom Smith

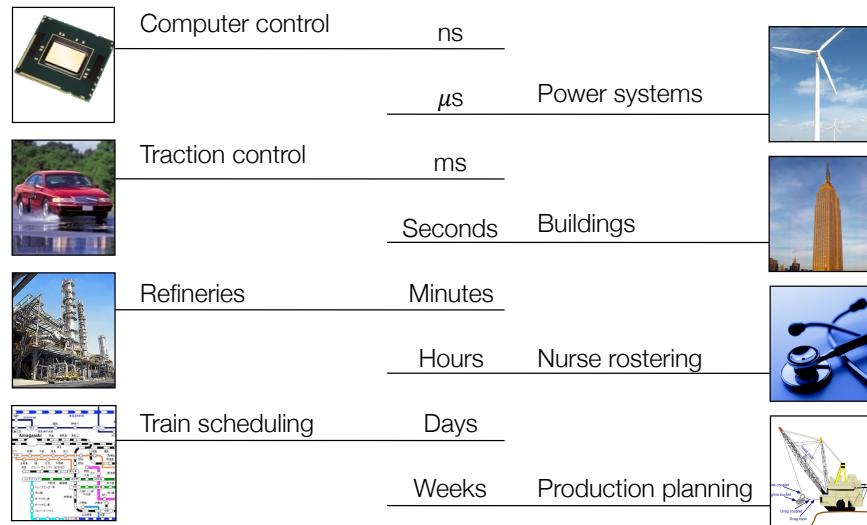
Demand Response

Demand Response Slides

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Control Applications at all Space and Time Scales



Summary

- Feedback control is everywhere
- It is used to:
 - Stabilize unstable systems
 - Make behaviors repeatable / predictable
 - Maximize performance
 - Understand what complex systems are doing