

## LAB 3 – Inertial Navigation in 2D / realistic signal (1 week)

### Objective:

Determine the position, velocity, and attitude (PVA) of a rigid body undergoing uniform circular motion using “2D” strapdown inertial navigation system with realistic inertial measurements.

### Task:

1. Simulate realistic measurements for a gyro and 2 orthogonal accelerometers by adding the following noise structure (stochastic values are listed in table below) to the nominal measurements (LAB 2) obtained earlier:

**Gyro errors:** *random-const. (RC) + 1<sup>st</sup> order Gauss-Markov (GM) + white noise (WN)*

**Accelerometer errors:** *random-const. (RC) + white noise (WN)*

2. Perform strapdown inertial navigation (**use trapezoidal integration at 100 Hz**) with the simulated realistic signal. Study the effect of the individual error sources on trajectory determination and identify the predominant influences.

**Hint:** apply error sources first separately then study their combined effect. Use your study to answer the questions and make synthesis in a report.

### Questions:

For each of the quantity being asked to be analyzed, make separate tables to answer the questions

- I. Which of the three error sources influence azimuth estimation *the least*?
- II. Which error source influences velocity estimation *the most*?
- III. Which error source influences position estimation *the most*?

### Numerical data:

- Circle radius: 500 m.
- Angular speed  $\omega = \pi/100$  rad/s
- Initial position: on North axis
- Initial azimuth:  $90^\circ$  (measured from North axis in clockwise direction)
- Initial velocity: north-axis: 0, east-axis:  $\omega \cdot \text{radius}$
- Acceleration due to gravity,  $g = 9.81 \text{ m/s}^2$

**!!! IMPORTANT!!!**

Simulated errors **must be re-scaled** to correct units with respect to sampling freq.

Error type	Notation	Stochastic value			Note
		Provided units	Needed (SI) units	Value (SI)	
<b>Gyro</b> bias (random constant)	$b_G$	150 deg/h	rad/s		$1\sigma$
<b>Gyro</b> correlated noise (1 <sup>st</sup> order Gauss-Markov)	$\sigma_{G_{GM1}}^{PSD}$	0.007 deg/s/√Hz	rad/s/sample		PSD level (scale for simulation!)
	$1/\beta_G$	100 s	s	100 s	correlation time
<b>Gyro</b> white noise	$\sigma_{G_{WN}}^{PSD}$	0.10 deg/√h	rad/s/sample		PSD level
<b>Accelerometer</b> bias (random constant)	$b_A$	1.3 mg	m/s <sup>2</sup>		$1\sigma$
<b>Accelerometer</b> noise (white)	$\sigma_{A_{WN}}^{PSD}$	57 μg/√Hz	m/s <sup>2</sup> /sample		PSD level

Table 1

**Deliverables:**

**A REPORT SHOULD CONSIST OF ONLY THE FOLLOWING ELEMENTS. PLEASE DO NOT PROVIDE DETAILS THAT ARE NOT ASKED!**

1. **Table 1** filled with converted units for stochastic error values.
2. **Plot** trajectory **errors** (i.e. in azimuth; position – N, E; velocity – N, E) with respect to *time* along one revolution for the case when all noise sources are present.
3. **Answers** to the questions
4. Your **code**

Lab weight: 5%

**Deadline:** 06.04.2025