

# Introduction to the Design of Space Mechanisms

Theme 6 part 1:  
Components  
Ball-bearing

Gilles Feusier

# Components

- Existing and, if possible, **qualified components** used in mechanisms:
  - Ball bearings, ball screws
  - Flex-pivots
  - Actuators, motors
  - Electrical, Electronic and Electromechanical (EEE) components:
    - Connectors,
    - Cables,
    - Switches,
    - Sensors (e.g. position sensors) and gauges,
    - Electrical and electronic components (including thermistors and heaters)
- **Use of Commercial Off-The-Shelf (COTS) components**
  - + Low cost, short lead time, large series (well known in defined constraints)
  - Batch variabilities, change of specification/manufacturing processes, obsolescence, requirements outside of the design specifications

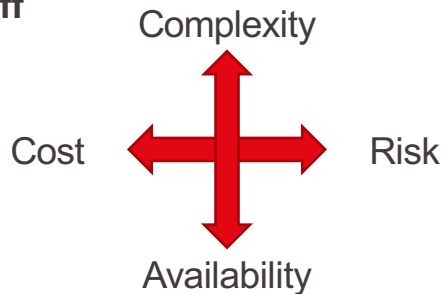
# EEE Components

- Acc. to **ECSS-Q-ST-60C Rev.3** [6.1]
- This standard defines the requirements of EEE components for space projects for
  - selection,
  - control, *EEE: Electrical, Electronic and Electromechanical components*
  - procurement,
  - usage.
- Components are divided into 3 classes:
  - Class 1 Highest assurance, lowest risk, highest cost
  - Class 2
  - Class 3 Lowest assurance, highest risk, lowest cost
- The project objectives, definition and constraints determine which class or classes of components are appropriate to be utilized within the system and subsystems.



- EEE components maybe
  - Available and space qualified from a catalogue (should be used in priority for ESA projects or depending on requirements).
  - Available from a catalogue but requiring a space qualification or an extension of their qualification (e.g. COTS).
  - Specifically developed for the purpose of the project. This later case requiring a full qualification program.

## Trade-Off



*Availability includes lead time and export restrictions*

# EEE Components

- Each class of EEE components requires a specific associated documentation (acc. to **ECSS-Q-ST-60C Rev.3**):
  - Tables 4-1, 5-1, 6-1 of ECSS-Q-ST-60C Rev.3
  - Main normative documents associated to EEE components, according to the Document Requirement Definition (DRD) in annexes A to D of ECSS-Q-ST-60C Rev.2
    - Component Control Plan (CCP). Class 1 components only.
    - Declared Component List (DCL)
    - Procurement Specification
    - Part Approval Document (PAD)
    - ...
- ESCC system for the specification, qualification and procurement of EEE parts for use in Space programs ([European Space Components Coordination](https://spacecomponents.org) / <https://spacecomponents.org>).
- For commercial components: ECSS-Q-ST-60-13C Rev.1 “Space product assurance - Commercial electrical, electronic and electromechanical (EEE) components”

# What is your experience with ball-bearings?

- Which type of ball-bearings do you know?
  - What type of configuration do you know?
  - Where are ball-bearings used in a spacecraft?
- 
- Turn to your neighbors (3-5 people teams)
  - **5 minutes**
  - Present the results of your discussion



# Ball Bearing Applications in Space

©Collins Aerospace



Attitude control  
system actuators

©TAS-CH



Sensor pointing  
mechanisms

©Sener



Deployment  
mechanisms

©ESARUAG



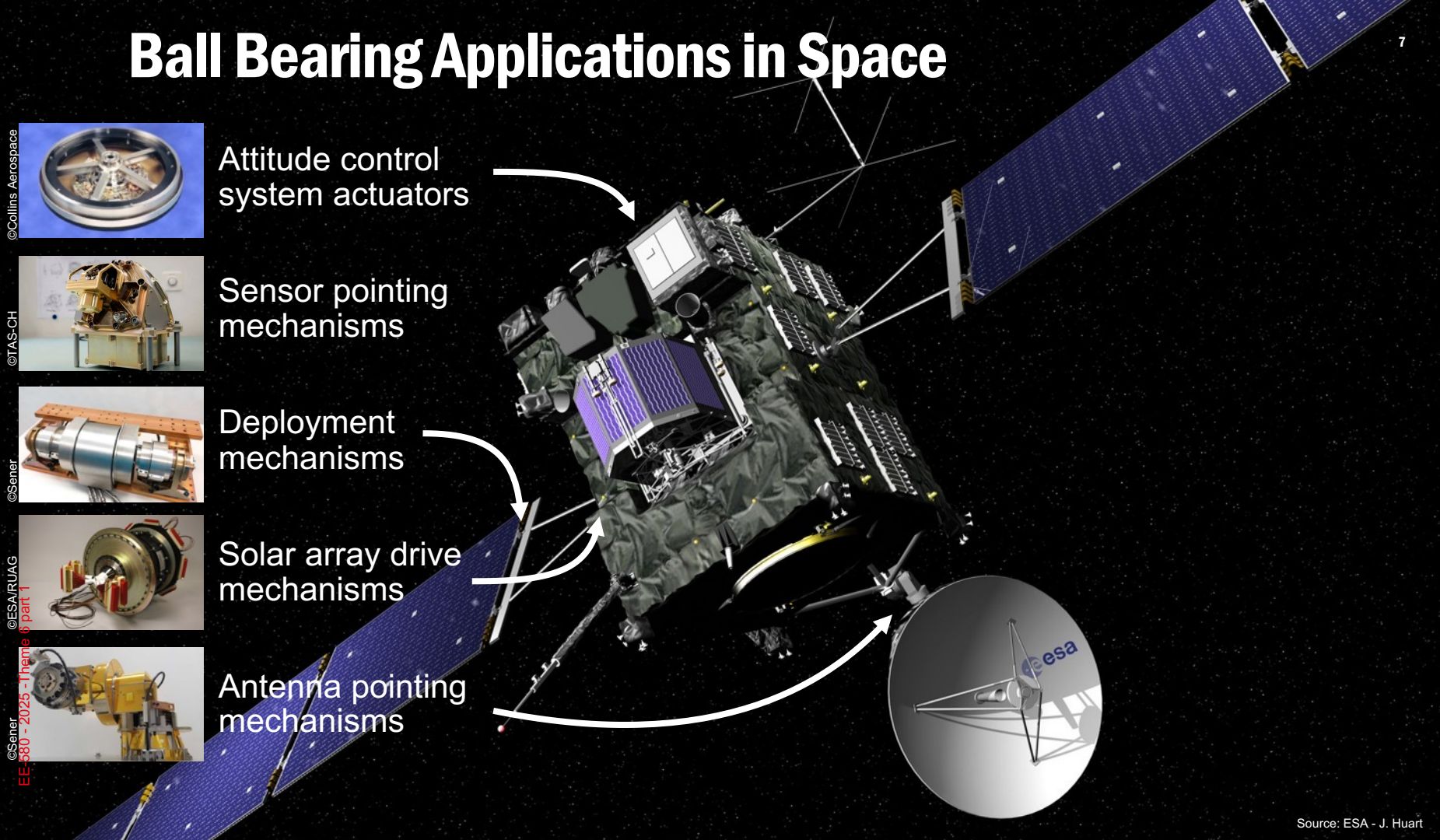
Solar array drive  
mechanisms

©Sener

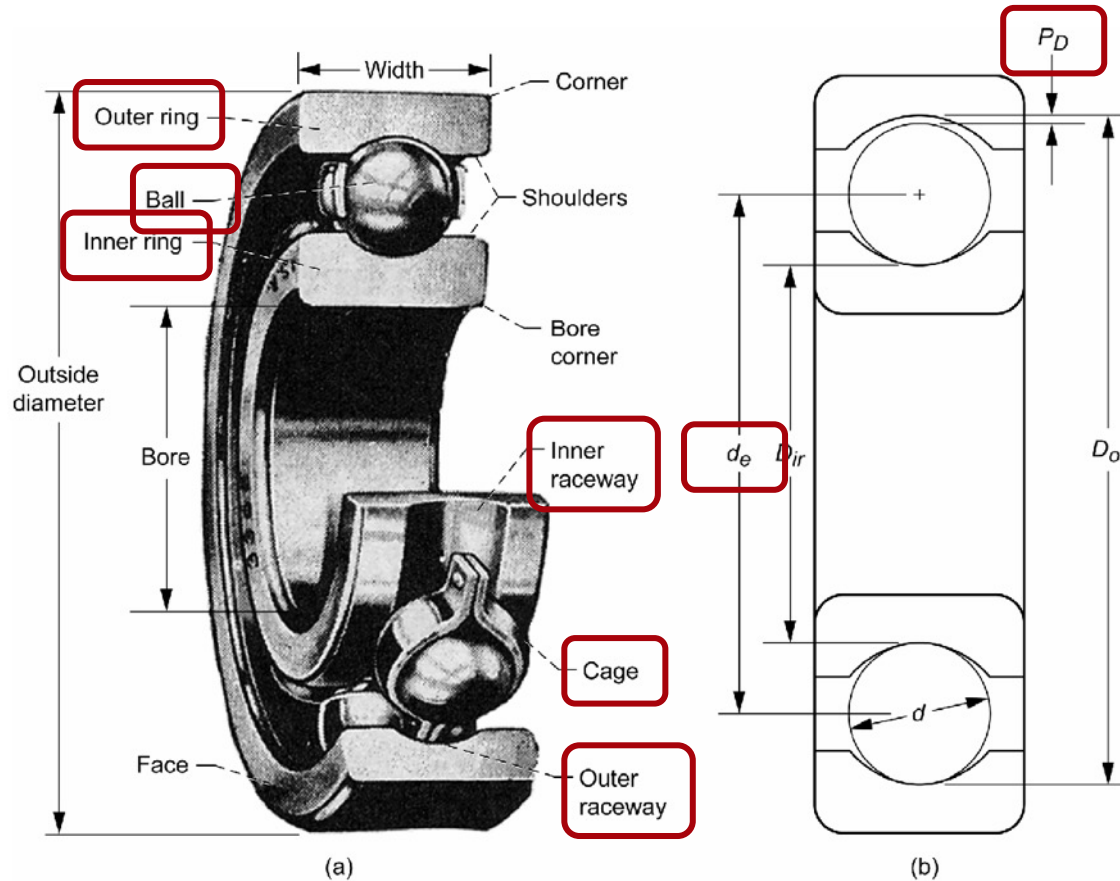


Antenna pointing  
mechanisms

EE-580 - Theme 6 part 1



# Ball Bearing Parts



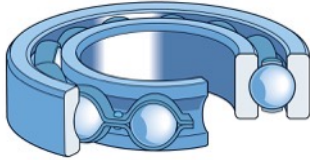
$P_D$ : diametral clearance  
 $D_{ir}$ : raceway diameter at ball-race contact of inner race  
 $D_{or}$ : raceway diameter at ball-race contact of outer race

Pitch diameter:

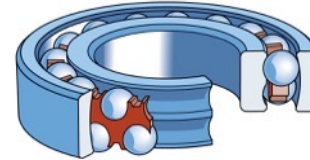
$$d_e = \frac{D_{or} + D_{ir}}{2}$$



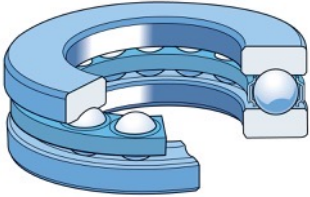
# Bearing Types and Geometry



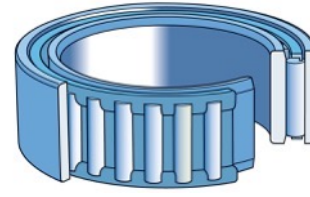
Deep groove bearing



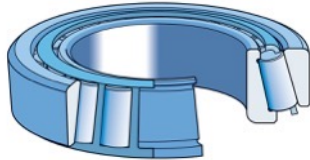
Self-aligning ball bearing



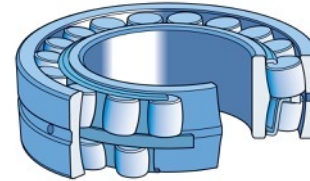
Thrust ball bearing



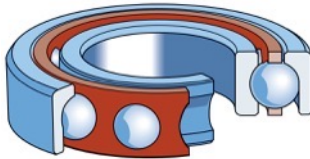
Needle roller bearing



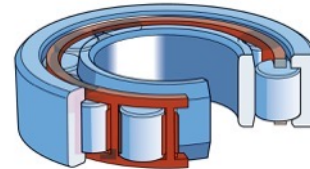
Tapered roller bearing



Spherical roller bearing

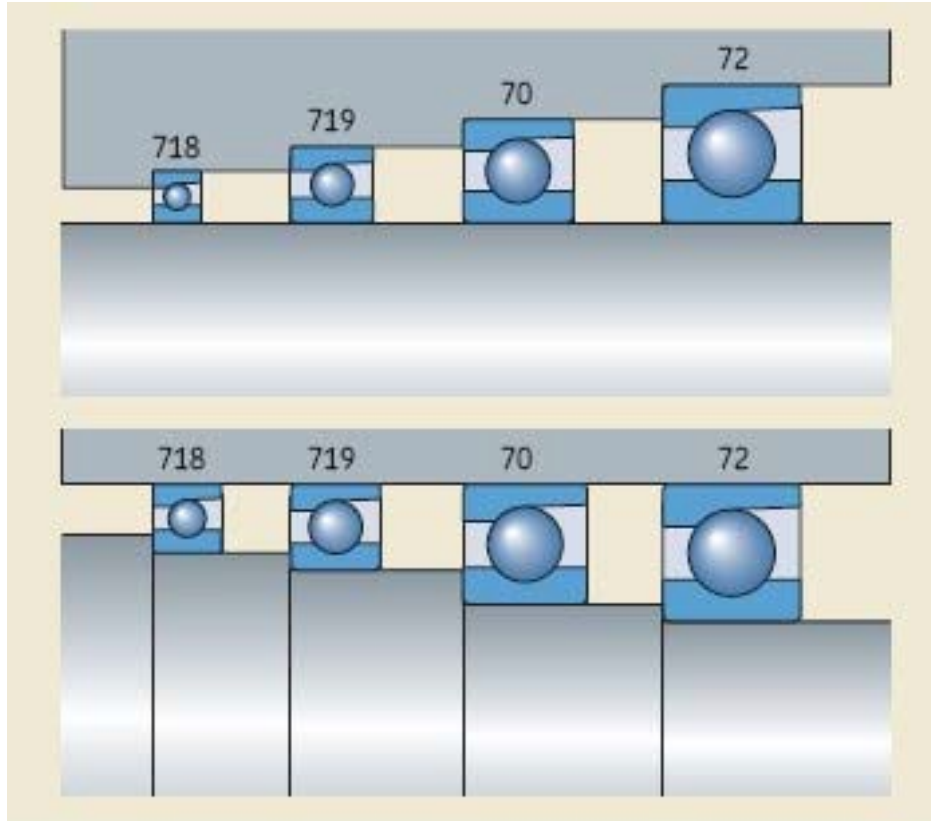


Angular contact ball bearing



Cylindrical roller bearing

# Bearing Types and Geometry - Sizes

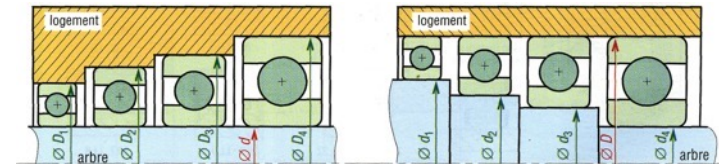


Use of coding.

In general a code for the series of ball-bearing (type) followed by the dimension codes (ISO).

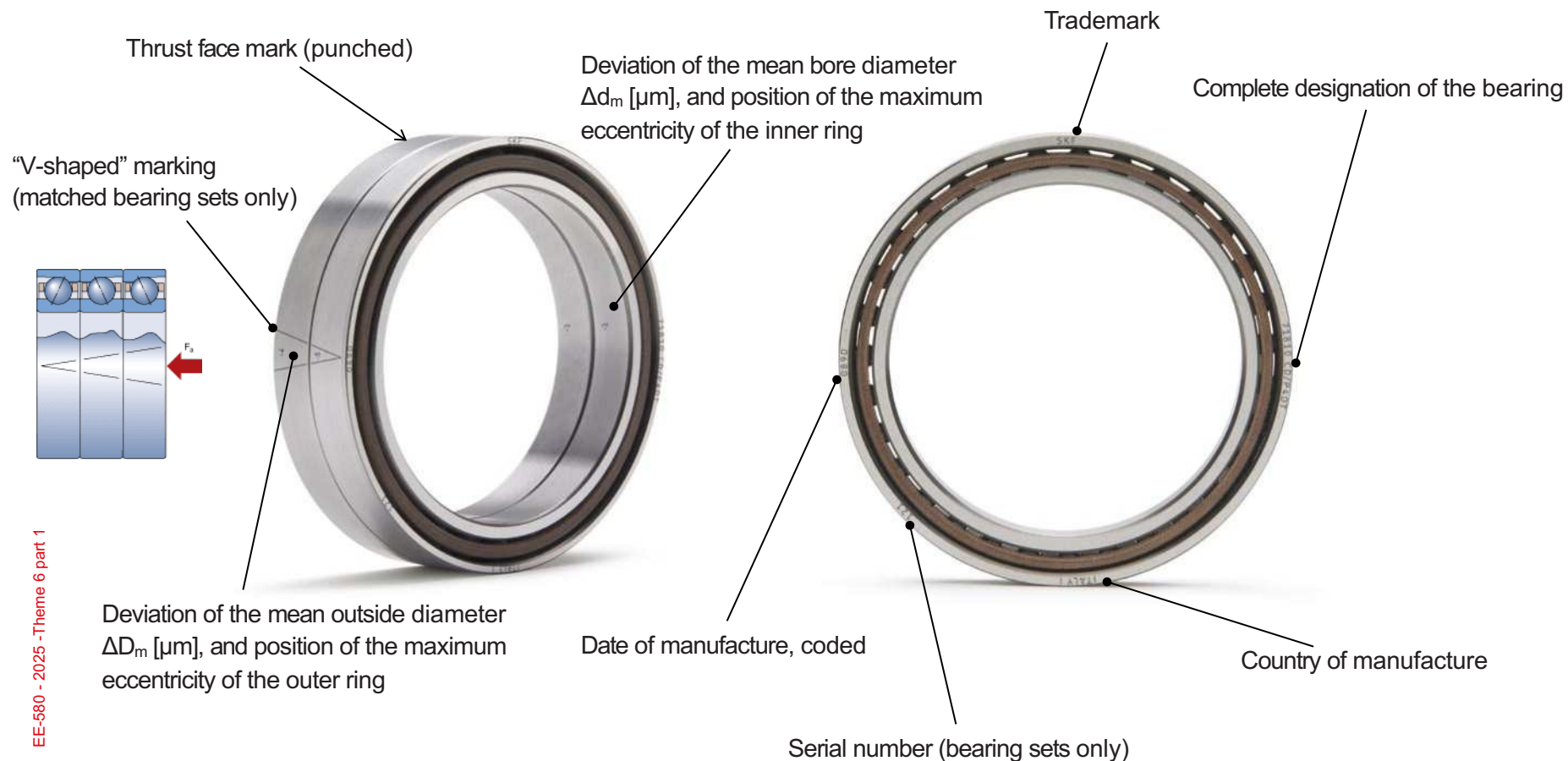
7 = angular contact ball bearing

Diameter of axis and load will permit to define the size of the bearing.



Source: [http://www.zpag.net/Tecnologies\\_Industrielles/Roulements\\_Etude.htm](http://www.zpag.net/Tecnologies_Industrielles/Roulements_Etude.htm)

# Bearing Marking Example



# Bearing Types and Geometry - Cage



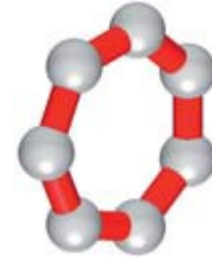
Pressed metal  
(Steel, copper)



Massive ring  
(organic, composite,  
bronze)



Crown type  
(organic, composite ...)



Spacers (cylindres)  
(PTFE typ.)



Spacers (rings)  
(PTFE typ.)



Full balls

## Trade-Off

Mechanical  
strength



Friction  
torque

# Materials for Ball Bearings for Space Applications

EN standard (chemical composition)	AISI	Standard	Remarks
X105CrMo17	440	AMS 5630, 5880, 5618	Z100CD17
X40CrMoVN16.2	-	AMS 5925	XD15NW™
X30CrMoN15.1	-	AMS 5898	CRONIDUR® 30
100Cr6	SAE 52100	AMS 6440, 6444	100C6
HS 18-0-1	T1	AMS 5626	High-speed steel
80MoCrV40	M50	AMS 6490, 6491	Semi high-speed steel
X115CrMoV14.4.1	-	AMS 5749	BG42®
CoCr30W8	-	-	ALACRITE 554
CoCr32W13	-	-	ALACRITE 505
Ti 6Al-4V	-	AMS 4911, 4928, 4935, 4965, 4967	Titanium alloy TA6-V
Si <sub>3</sub> N <sub>4</sub>	-	-	Silicon nitride (ceramic)

Source: ADR catalogue. List is not exhaustive

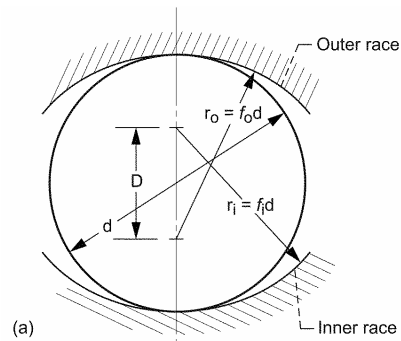
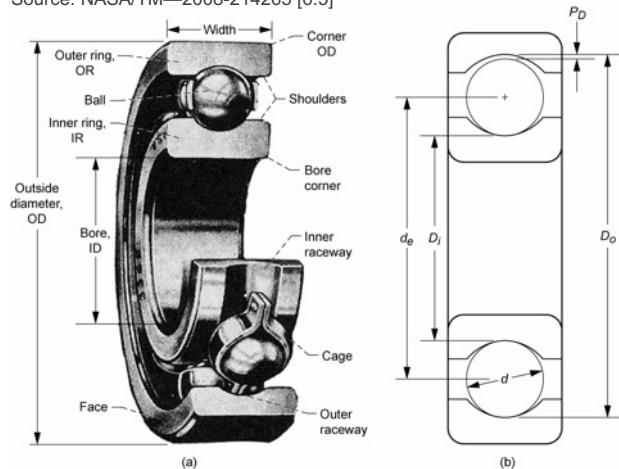
Stainless steels: suitable for Space applications

Standard steels: Corrosion (not suitable for Space)

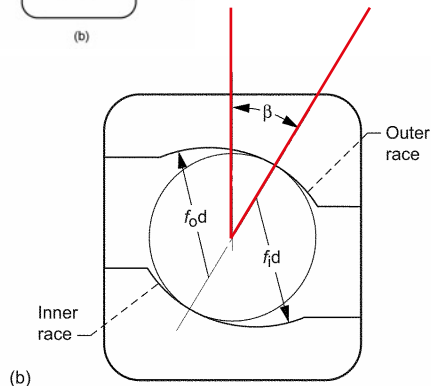
Light weight: Low load capacity

Ceramics: Fragile (suitable for Space with precautions)

Source: NASA/TM—2008-214265 [6.5]



Deep-groove ball bearing



Angular contact ball bearing

**Conformity:**  $f = \frac{r}{d} \quad (\approx 0.52)$

**Pitch diameter:**  $d_e = \frac{D_o + D_i}{2}$

**Contact angle:**  $\beta$  between  $10^\circ$  and  $30^\circ$   
The choice depends on the load case

With:

$d$ : diameter of the ball

$r$ : race-groove radius ( $r_o$  or  $r_i$ )

$f$ : shall be  $> 0.5$  for geometric reasons ( $f_o$  or  $f_i$ )

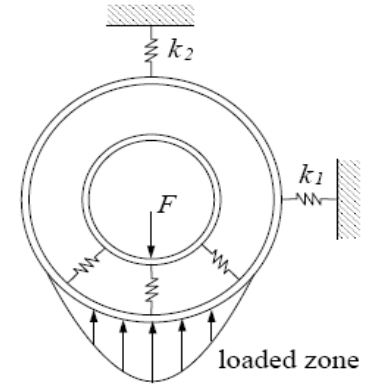
$P_D$ : diametral clearance



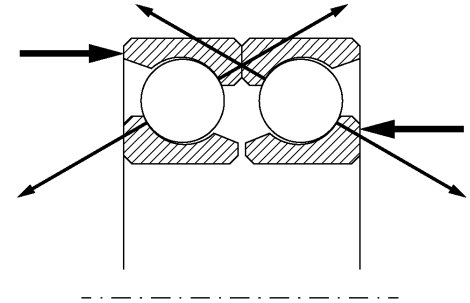
- The forces are distributed on the ball as a function of the load direction
  - **Radial load** (*distributed*)
  - **Axial load** (*uniformly*)
  - **Combined load**
    - Combination of radial and axial loads
    - Calculation of an equivalent load  $P_{eq}$  producing a stress equivalent to the combined load.
- Standard mathematical model for predicting ball bearing lifetime:
  - **Lundberg-Palmgren theory** (1947/1952)
    - Takes into account stresses and fatigue theories
    - Use of many experimental measures
    - Statistic analysis based on Weibull method



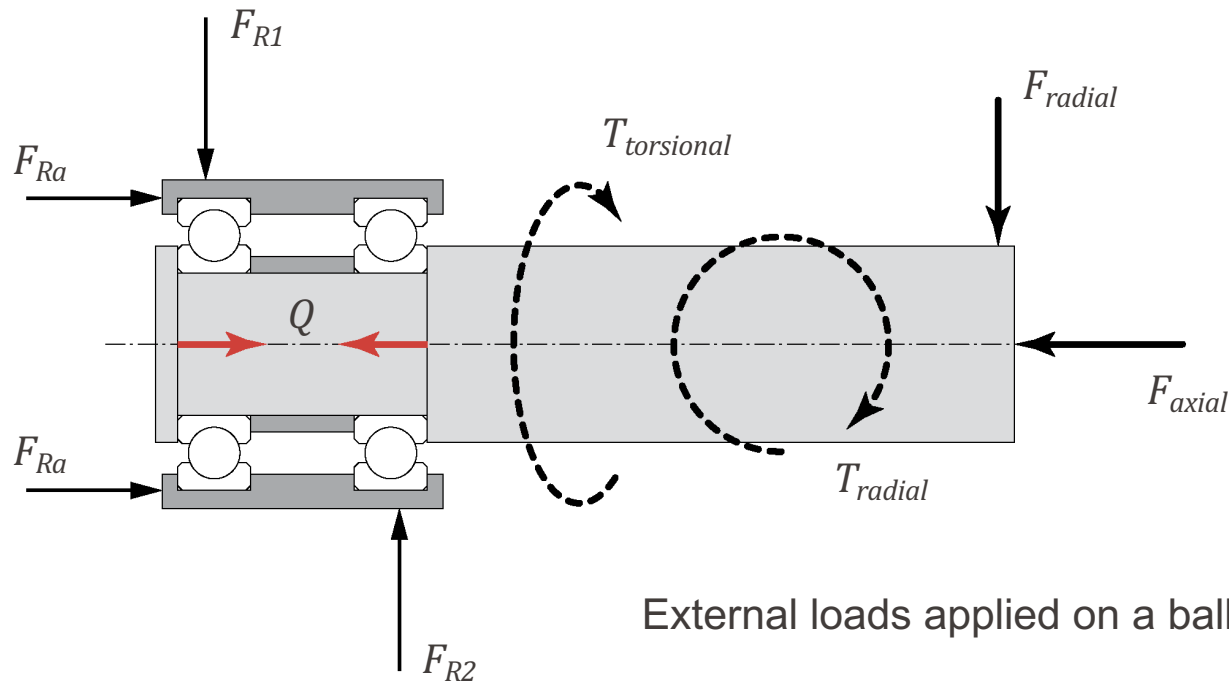
**Radial load:**  
only few balls  
are loaded not  
uniformly



**Axial load:**  
all balls are uniformly loaded



# Functional Parameters



External loads applied on a ball bearing system

- $F_{axial}$ ,  $F_{radial}$ ,  $T_{torsional}$ ,  $T_{radial}$ : Functional applied forces and torques
- $F_{R1}$ ,  $F_{R2}$ ,  $F_{Ra}$ : Reaction forces acting on the mechanism
- $Q$ : Preload (also participates to the ball bearing stressing loads)

# References for the Lundberg-Palmgren Theory

- NASA Technical Memorandum 107440 [6.2]

## **A. Palmgren Revisited - A Basis for Bearing Life Prediction**

Erwin V. Zaretsky

*Lewis Research Center*

*Cleveland, Ohio*

Prepared for the

STLE Annual Meeting

sponsored by the Society of Tribologists and Lubrication Engineers

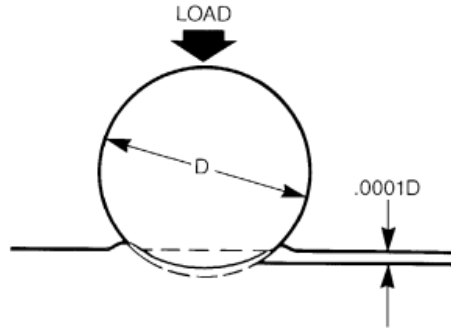
Kansas City, Missouri, May 18–22, 1997

- See also ADR catalogue for practical use of the theory  
Cf. [6.3] ADR-catalog.pdf on the MOODLE

- Some important definitions
  - Basic static radial load rating  $C_o$ :
    - value of the static radial load which will cause a total permanent deformation (ball and raceway) on the most heavily stressed ball/raceway contact (4'200 MPa for ground applications) of approximately 0.0001 of the ball diameter.
  - Basic dynamic radial load rating  $C$ :
    - value of the radial load of constant strength and direction that can be theoretically sustained for a nominal bearing life of 1 million revolutions.
  - Equivalent static radial load  $P_o$ :
    - a static radial load that would cause the same total permanent deformation on the most loaded contact as that obtained under effectively applied loads.
  - Equivalent dynamic radial load  $P$ :
    - a dynamic radial load that is constant in magnitude and direction under which the reached life would be the same as that with effectively applied loads.

# Ball Bearing Life Evaluation

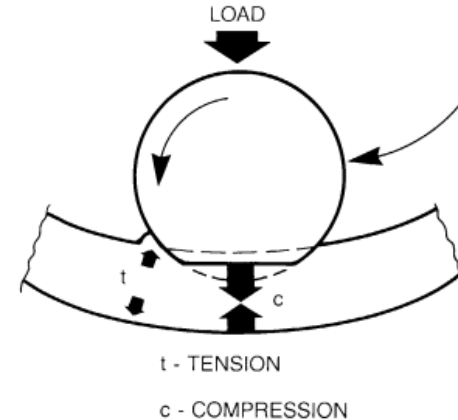
- Definition of the load ratings



## Basic static radial load $C_0$

- No or very slow rotation
- Very slow oscillations
- Shocks

Permanent deformation of rolling element and raceway ( $0.0001 \times D$ )



## Basic dynamic radial load $C$

- Rotation under load
- Cyclic stresses (fatigue)

Basic rating life of  $10^6$  revolutions

# Ball Bearing Life Evaluation

- Equivalent load ratings

$$P = X \cdot F_{radial} + Y \cdot F_{axial} \quad [N]$$

where:  $X$  radial coefficient of the bearing  
 $Y$  axial coefficient of the bearing

Cf. tables given by manufacturer  
 (e.g. ADR catalogue)

Contact angle	$\frac{F_a}{C_o}$	Single bearing or DT pair						DO or DX pairs			
		$\frac{F_a}{F_r} \leq e$		$\frac{F_a}{F_r} > e$		$X_o$	$Y_o$	$\frac{F_a}{F_r} \leq e$		$\frac{F_a}{F_r} > e$	
		X	Y	X	Y			X	Y	X	Y
5°	0.014	0.23			2.30				2.78		3.74
	0.028	0.26			1.99				2.40		3.23
	0.056	0.30			1.71				2.07		2.78
	0.085	0.34			1.55				1.87		2.52
	0.110	0.36			1.45				1.75	0.78	2.36
	0.170	0.40			1.31	0.6	0.5	1	1.58		2.13
	0.280	0.45			1.15				1.39		1.87
	0.420	0.50			1.04				1.26		1.69
10°	0.560	0.52			1.00				1.21		1.63
	0.014	0.29			1.88				2.18		3.06
	0.029	0.32			1.71				1.98		2.78
	0.057	0.36			1.52				1.76		2.47
	0.086	0.38			1.41				1.63		2.29
	0.110	0.40			1.34				1.55	0.75	2.18
	0.170	0.44			1.23	0.6	0.5	1	1.42		2.00
	0.290	0.49			1.10				1.27		1.79
15°	0.430	0.54			1.01				1.17		1.64
	0.570	0.54			1.00				1.16		1.63
	0.015	0.38			1.47				1.65		2.39
	0.029	0.40			1.40				1.57		2.28
	0.058	0.43			1.30				1.46		2.11
	0.087	0.46			1.23				1.38		2.00
	0.120	0.47			1.19	0.5	0.46	1	1.34	0.72	1.93
	0.170	0.50			1.12				1.26		1.82
20°	0.290	0.55			1.02				1.14		1.66
	0.440	0.56			1.00				1.12		1.63
	0.580	0.56			1.00				1.12		1.63
	—	0.57		0.43	1.00		0.42		1.09	0.70	1.63
	—	0.68		0.41	0.87		0.38	1	0.92	0.67	1.41
25°	—	0.80	1	0	0.39	0.5	0.33		0.78	0.63	1.24
30°	—	0.80			0.39				0.78	0.63	1.24
35°	—	0.95			0.37		0.29		0.66	0.60	1.07

Source: ADR catalogue

- Nominal life for individual bearings (Lundberg-Palmgren)  $L_{10}$ :

$$L_{10} = \left( \frac{C}{P} \right)^p \quad [10^6 \text{ rev.}] \quad \text{where } p = 3 \text{ for ball bearings}$$

The **life associated with a reliability of 90%** (90% probability of survival)



# Ball Bearing Life Evaluation

Reading: E. Zaretsky et al. “Relation Between Hertz Stress-Life Exponent, Ball-Race Conformity, and Ball Bearing Life”, NASA/TM-2008-214265 [6.5]

- The lifetime depends mainly on:
  - The Hertz pressure at ball-races contact points, function of:
    - Conformity  $f$
    - Ball diameter  $d$
    - Number of balls
    - Direction of the load
  - The materials
  - The travelled distance (number of revolutions)
  - The lubrication

# Ball Bearing Life Evaluation

- Hertz Pressure (in GPa)
  - Radial load (deep groove)

$$S_{max j} = K \cdot \frac{\left[ \frac{2 \cdot (-1)^{j+1}}{D_j} + \frac{4}{d} - \frac{1}{f_j d} \right]^{2/3}}{(\mu \cdot \nu)_j} P_{N_{max}}^{1/3}$$

- Axial load (angular-contact)

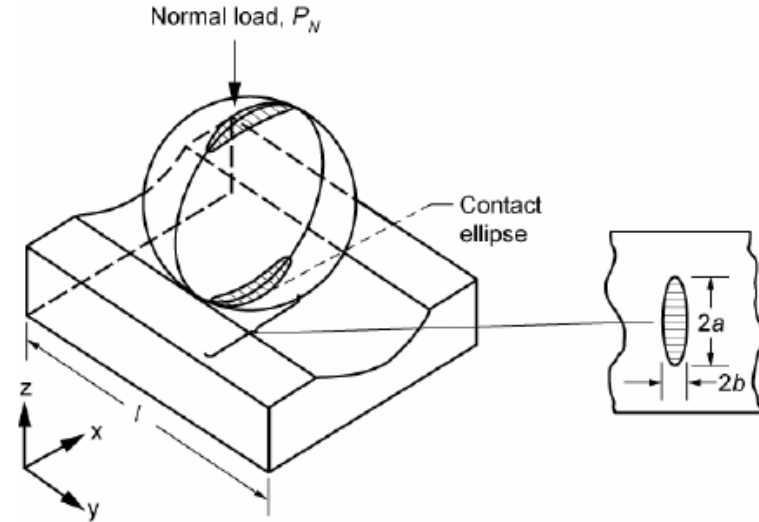
$$S_{max j} = K \cdot \frac{\left[ \frac{2 \cdot \cos(\beta)}{d_e + (-1)^j d \cdot \cos(\beta)} + \frac{4}{d} - \frac{1}{f_j d} \right]^{2/3}}{(\mu \cdot \nu)_j} P_N^{1/3}$$

with:  $D_j$ : diameter of the grooves  
 $j = 0$ : outer race  
 $j = 1$ : inner race

$d$ : diameter of the balls  
 $P_N$ : normal load on the balls  
 $P_{N_{max}}$ : normal load on the most loaded ball

$f_j$ : conformity  
 $\beta$ : contact angle  
 $d_e$ : pitch diameter

$K$ : geometry-load constant ( $K = 1.58 \cdot 10^{-3}$  for bearing steel on bearing steel,  $S_{max}$  in GPa)



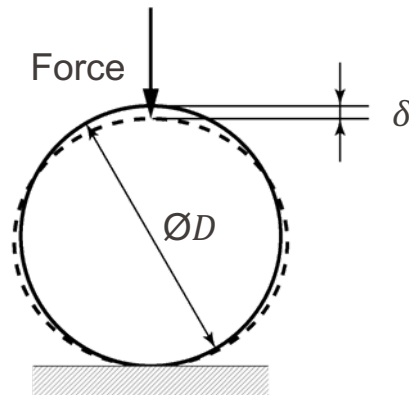
$\mu$  and  $\nu$ : transcendental functions from Hertz theory

# Ball Bearing Life Evaluation

- Allowable Hertz pressure

- **For terrestrial, lubricated applications (ISO 76)**<sup>1</sup>  $P_H \leq 4'200 \text{ MPa}$ 
  - Corresponds to a total permanent deformation of the rolling element of diameter  $D$  of:

$$\frac{\delta}{D} = 1/10'000$$



- **Space applications**

- According to ECSS-E-ST-33-01C<sup>1</sup>  $P_H \leq 3'360 \text{ MPa}$
- **Other applicable standards** (e.g. in-house standards):
  - Under vibrations (for sizing):  $P_H \leq 2'414 \text{ MPa}$
  - In orbit:  $P_H \leq 1'034 \text{ MPa}$
  - Permissible actual loads may reach  $1.5 \times P_H$  (safety factor)
  - ...

<sup>1</sup> for SAE 52100

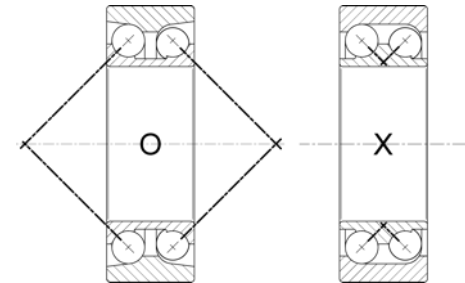
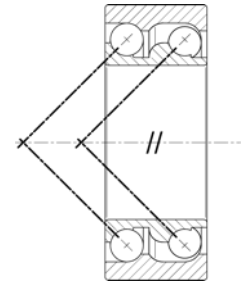
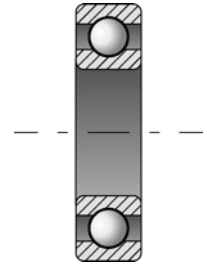
# Ball Bearing Preload

## ■ Preload

- Always required for precision mechanisms using ball bearings
  - Control of the stiffness of the bearing assembly
  - Assembly without play (axial and radial)
  - Stability in vibration (gapping allowed or not)
- **Drawbacks**
  - The permanent axial force is detrimental for:
    - Lifetime
    - Resistive torque
- Types of assembly (consequences will be discussed later on)
  - Without any spring: **Hard preload**
  - With a spring: **Soft preload**

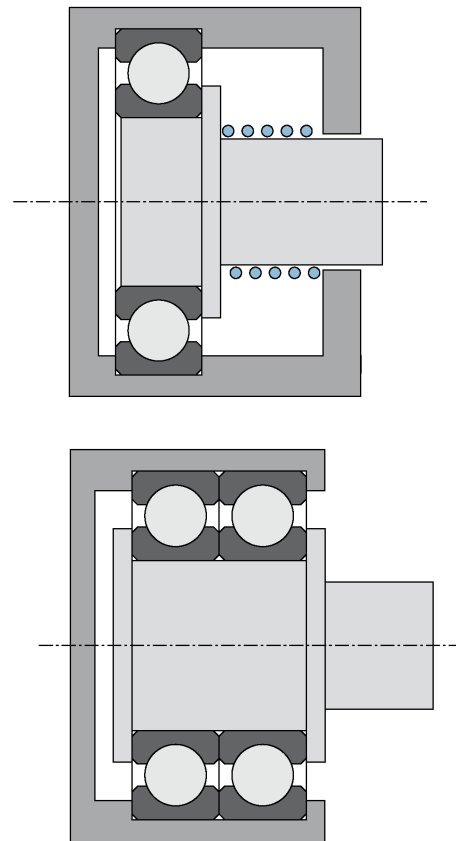
# Ball bearing Assembly

- One single ball bearing (deep groove) with radial contact
  - Barely used: axial and radial instability, problems in vibrations
  - Could be useful for laterally supporting a bearing, a pulley, for centering ...
- Tandem configuration
  - Used for doubling or multiplying the load capacity
  - Barely used for space mechanisms
  - Generally combined in a Tandem-Duplex configuration
- Duplex configuration
  - Assembly of a pair of ball bearings with a preload
  - Main assembly configuration for precision mechanisms



# Ball bearing Assembly

- Single ball bearing with elastic preload
  - Simple assembly
  - Low stiffness (not favorable in vibration)
- Ball bearing pair with rigid preload
  - Back to back assembly (**O-mount**)
  - Face to face assembly (**X-mount**)
  - Requires high mechanical accuracy
  - Very rigid (good behavior in vibration)
- Multiple ball bearings assembly
  - More than two ball bearing on the shaft
  - High loads capacity

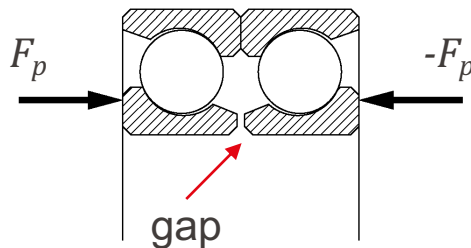




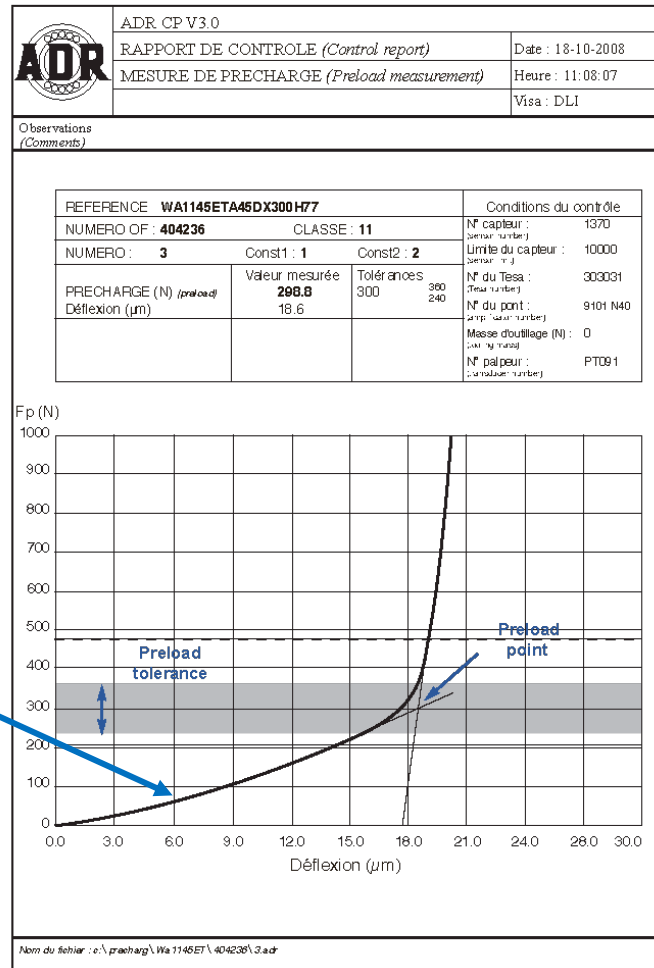
- Elastic preload ( $Q$ )
  - $k$ : spring stiffness
  - $\delta$ : spring deflection

$$Q = k \cdot \delta$$

- Hard preload
  - Non-linear stiffness
  - Sharp stiffness change when gap reduced to 0
  - Measured from the record:



$F_p$  vs.  $\delta$



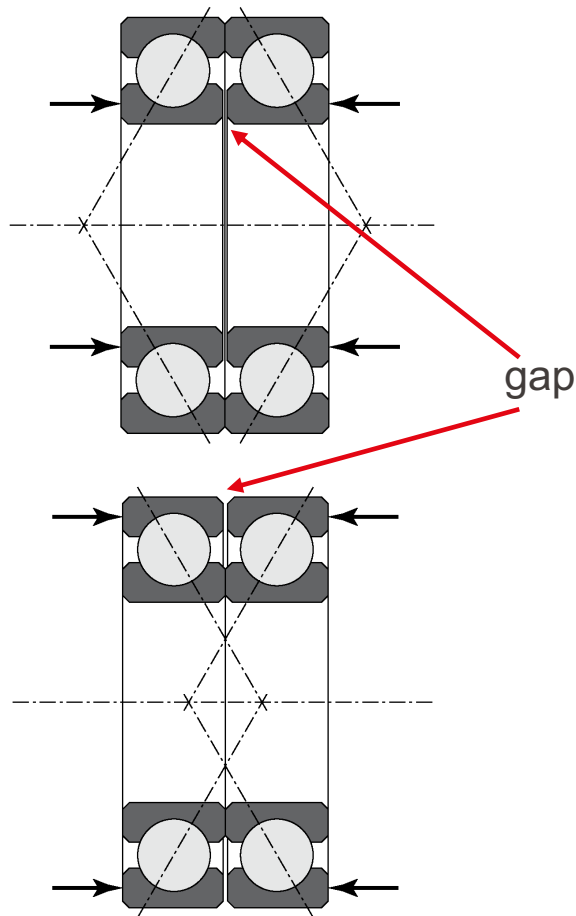
# Ball bearing Assembly (duplex)

## ▪ O-Mount (back to back)

- Very stiff
- Able to carry high moments
- Requires a very accurate alignment in order to avoid torque variations and wear

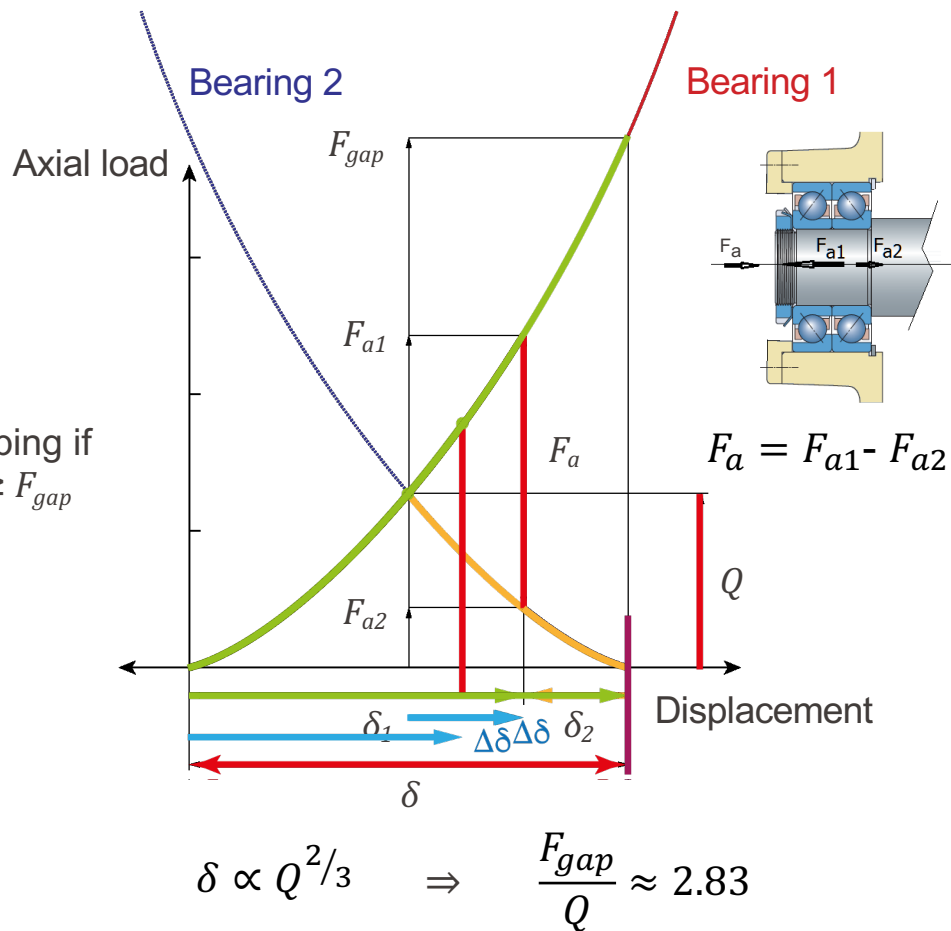
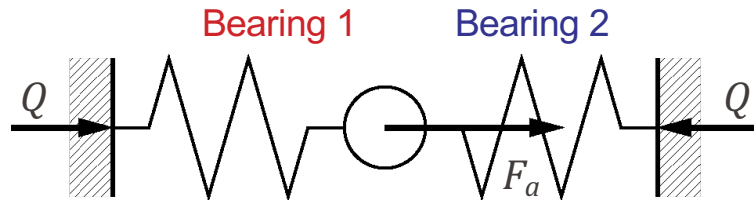
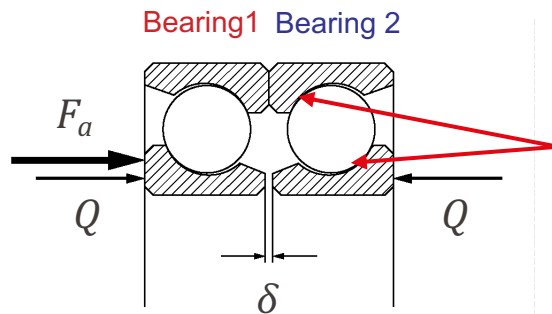
## ▪ X-Mount (face to face)

- More tolerant to misalignment
- Used for a regular torque
- Not appropriate for carrying high moments



# Ball bearing Assembly

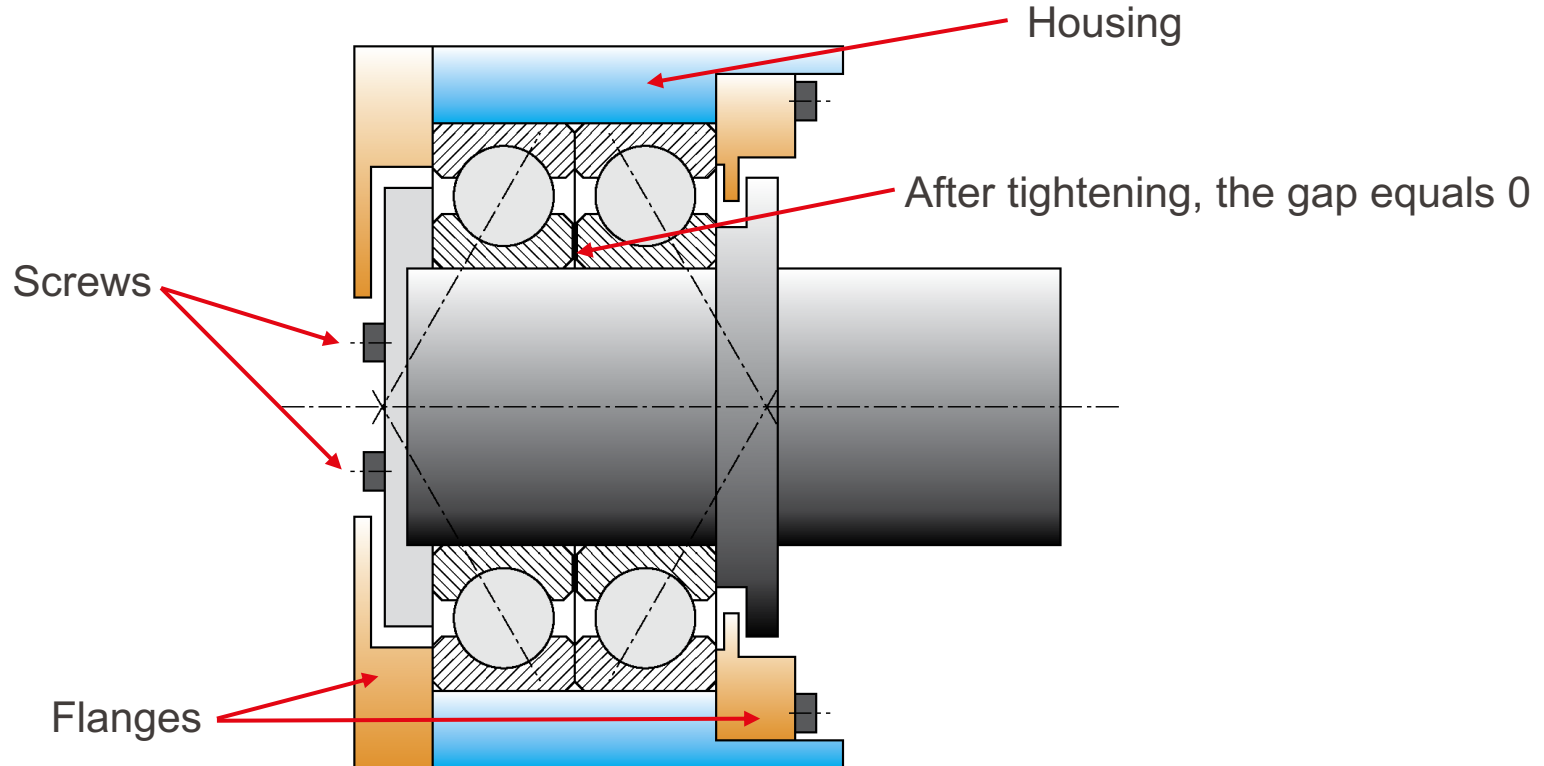
- Duplex ball bearing stiffness
  - Non-linear stiffness
    - ⇒ Better load capacity



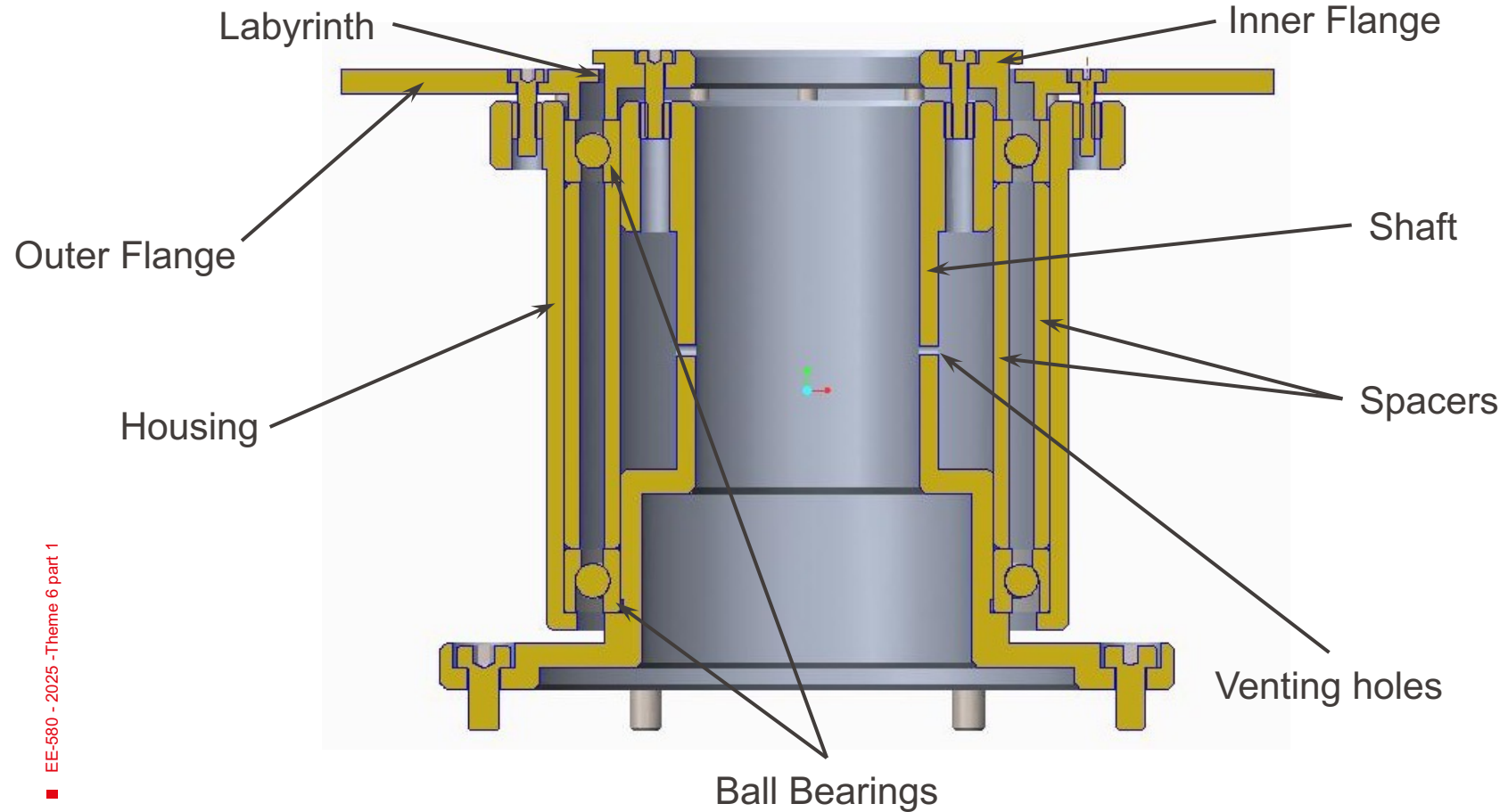
Reading: P. Guay et al. "Ball Bearing Stiffness. A New Approach Offering Analytical Expressions", Proceedings of the 16<sup>th</sup> European Space Mechanisms and Tribology Symposium (ESMATS 2015), September 2015. [6.6]

# Ball bearing Assembly

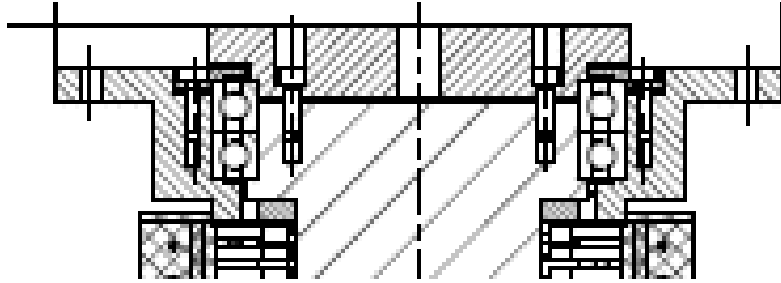
- Duplex Assembly (O-Mount, hard preload)



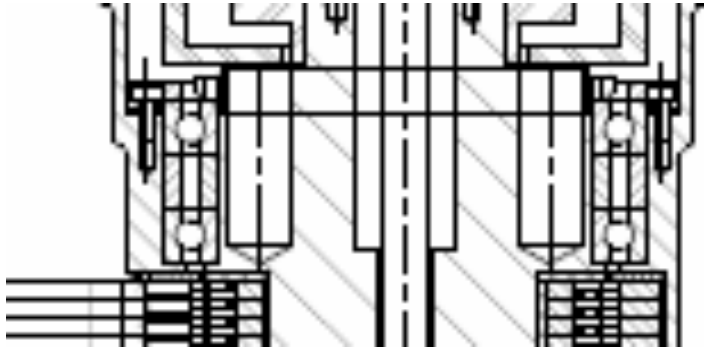
# Ball bearing – Example of Duplex O-Mount



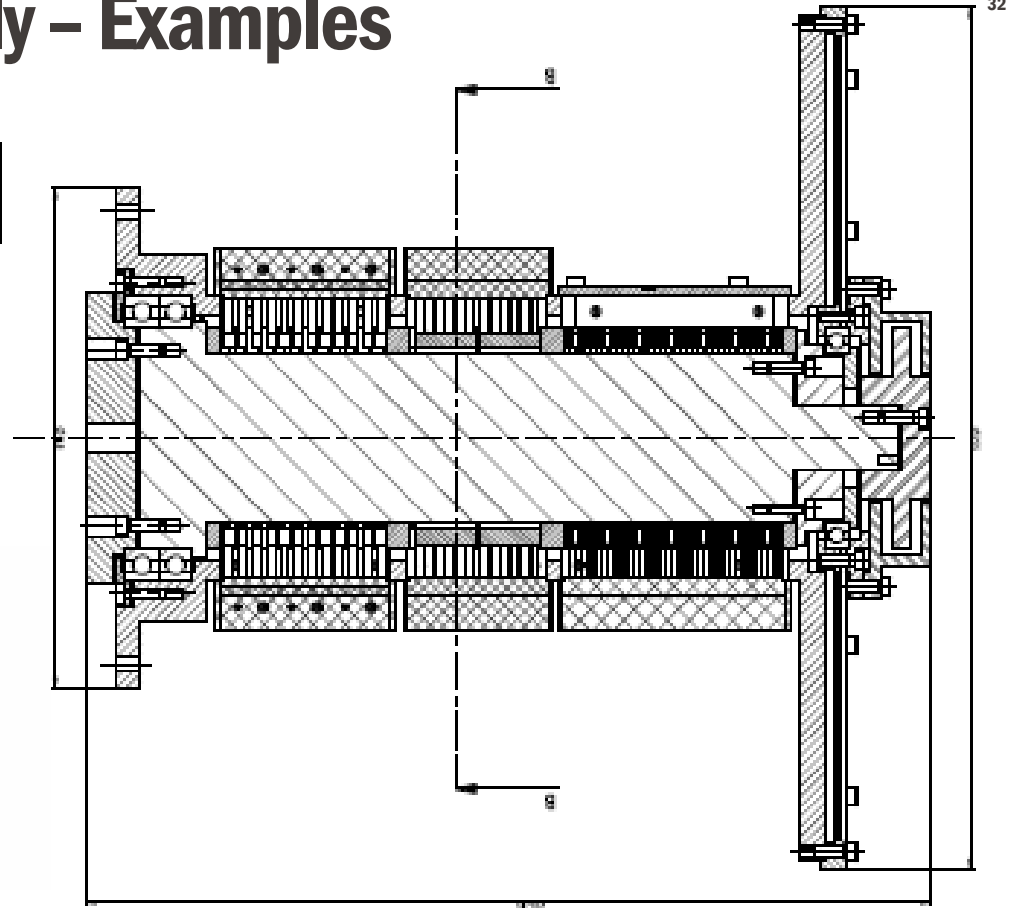
# Ball bearing Assembly – Examples



Hard mount



Hard mount with spacers



Hard mount on one side, supported by a membrane on the other side



# Theme 6 Part 1 Summary

- Various type of components
  - Mechanical components (bearings, actuators, ...)
  - EEE components (cables, connectors, sensors, ...)
    - According to ECSS-Q-ST-60C Rev.3 and ECSS-Q-ST-60-13C Rev.1
    - Classified: three classes
    - Use of COTS
- Ball-bearings
  - Usages
  - Types
  - Materials
  - Lundberg-Palmgren, life evaluation
  - Assembly
  - Preload

- Theme 6 – Components (continued): Ball-bearings, Motors and Actuators

Note:

- Mini Project part 3      Concept  
(cf. EE580\_MP3\_2025\_v1 Concept.pdf)  
Deadline **May 1<sup>2t</sup>, 16:00**