

## Solutions of Exercises of Chapter 7

### 7. Solution:

Assume the original system is,

$$\begin{aligned}\dot{\mathbf{x}} &= \mathbf{A}\mathbf{x} + \mathbf{B}u, \\ y &= \mathbf{C}\mathbf{x} + Du, \\ G(s) &= \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + D.\end{aligned}$$

Assume a change of state from  $\mathbf{x}$  to  $\mathbf{z}$  using the nonsingular transformation  $\mathbf{T}$ ,

$$\mathbf{x} = \mathbf{T}\mathbf{z}.$$

The new system matrices are,

$$\bar{\mathbf{A}} = \mathbf{T}^{-1}\mathbf{A}\mathbf{T}, \quad \bar{\mathbf{B}} = \mathbf{T}^{-1}\mathbf{B}, \quad \bar{\mathbf{C}} = \mathbf{C}\mathbf{T}, \quad \bar{D} = D.$$

The transfer function is,

$$\begin{aligned}G_z(s) &= \bar{\mathbf{C}}(s\mathbf{I} - \bar{\mathbf{A}})^{-1}\bar{\mathbf{B}} + \bar{D} \\ &= \mathbf{C}\mathbf{T}(s\mathbf{I} - \mathbf{T}^{-1}\mathbf{A}\mathbf{T})^{-1}\mathbf{T}^{-1}\mathbf{B} + D.\end{aligned}$$

If we factor  $\mathbf{T}^{-1}$  from the left and  $\mathbf{T}$  on the right of the  $(s\mathbf{T}^{-1}\mathbf{T} - \mathbf{T}^{-1}\mathbf{A}\mathbf{T})^{-1}$  term, we obtain:

$$\begin{aligned}G_z(s) &= \mathbf{C}\mathbf{T}[\mathbf{T}^{-1}(s\mathbf{I} - \mathbf{A})\mathbf{T}]^{-1}\mathbf{T}^{-1}\mathbf{B} + D \\ &= \mathbf{C}\mathbf{T}\mathbf{T}^{-1}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{T}\mathbf{T}^{-1}\mathbf{B} + D = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + D = G(s)\end{aligned}$$

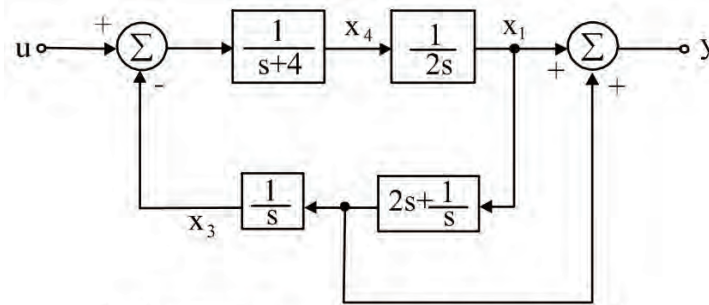
### 15. Solution

We are given  $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$ . Steady-state means that  $\dot{\mathbf{x}} = 0$  and a step input (or unit step) means  $u = 1(t)$ . Thus, assuming that the system is stable and  $\mathbf{A}$  is invertible (which you can check), we have,

$$\mathbf{0} = \mathbf{A}\mathbf{x}_{ss} + \mathbf{B} \implies \mathbf{x}_{ss} = -\mathbf{A}^{-1}\mathbf{B} = - \begin{bmatrix} -5 & 1 \\ -2 & -1 \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ 1 \end{bmatrix} = \begin{bmatrix} 1/7 \\ 5/7 \end{bmatrix}.$$

## 16. Solution

- (a) The block diagram can be simplified by moving the pick up point at  $x_4$  to  $x_1$ . This way,  $H_1$  will be changed to  $2s + 1/s$  and we obtain the following block diagram:

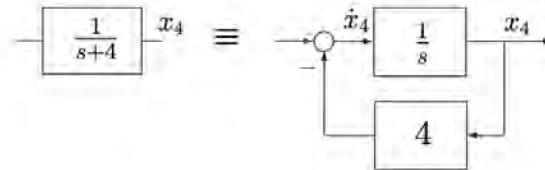


Block diagram for solution of Problem 7.16 (a).

In the next step the pick up point in the feedback loop (before  $1/s$  block) will be moved to  $x_1$ . This will create a new block  $1 + 2s + 1/s$  between  $x_1$  and  $y$  and eliminate the summation. The final transfer function will be:

$$\frac{Y(s)}{U(s)} = \frac{\frac{1}{s+4} \frac{1}{2s}}{1 + \frac{1}{s+4} \frac{1}{2s} \frac{2s^2+1}{s^2}} \frac{2s^2 + s + 1}{s} = \frac{s(2s^2 + s + 1)}{2s^4 + 8s^3 + 2s^2 + 1}$$

- (b) The block diagram includes essentially the integrators. The first order model  $G_1$  can be written in an equivalent form to emphasise the integrator as follows:



The state and output equations can be written from the block diagram of Fig. 7.85 as follows :

$$\dot{x}_1 = 0.5x_4 \quad ; \quad \dot{x}_2 = x_1 \quad ; \quad \dot{x}_3 = x_2 + x_4 \quad ; \quad \dot{x}_4 = u - x_3 - 4x_4$$

and  $y = x_1 + x_2 + x_4$ . This leads to the following state-space model:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 0 & \frac{1}{2} \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 1 \\ 0 & 0 & -1 & -4 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1 \end{bmatrix} u,$$

$$y = [ 1 \quad 1 \quad 0 \quad 1 ] \mathbf{x}.$$

The results can be checked for consistency using Matlab's command `ss2tf`.

## 22. Solution:

The natural frequency for a second-order system is related to the peak-time by the following relation (Chapter 3, Slide 41):

$$\omega_n = \frac{\pi}{t_p \sqrt{1 - \zeta^2}} = \frac{1}{\sqrt{1 - (0.707)^2}} = 1.414 \quad \text{rad/s}$$

Using full state feedback, we would like the a characteristic equation to be,

$$s^2 + 2\zeta\omega_n s + \omega_n^2 = s^2 + 2s + 2 = 0.$$

Using state feedback  $u = -\mathbf{K}\mathbf{x}$ , we get,

$$\dot{\mathbf{x}} = (\mathbf{A} - \mathbf{BK})\mathbf{x} = \begin{bmatrix} 0 & 1 \\ -6 - k_1 & -5 - k_2 \end{bmatrix} \mathbf{x}.$$

Hence the closed-loop characteristic equation is,

$$s^2 + (5 + k_2)s + (6 + k_1) = 0.$$

Comparing coefficients,  $k_1 = -4$  and  $k_2 = -3$ . The MATLAB command `place` can also be used.

## 25. Solution:

(a) Let's write this system in the control canonical form

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & -4 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u, \\ y = [1 \quad 0]\mathbf{x}$$

(b) If  $u = -[K_1 \quad K_2]\mathbf{x}$ , the poles of the closed-loop system satisfy  $\det(s\mathbf{I} - \mathbf{A} + \mathbf{BK}) = 0$ . Thus,

$$\det(s\mathbf{I} - \mathbf{A} + \mathbf{BK}) = \begin{bmatrix} s + K_1 & 4 + K_2 \\ -1 & s \end{bmatrix} = s^2 + K_1s + 4 + K_2$$

The closed-loop characteristic equation is:

$$(s + 2 - 2j)(s + 2 + 2j) = s^2 + 4s + 8$$

Comparing coefficients, we have  $K_1 = 4$  and  $K_2 = 4$ .

**34. Solution:**

(a)

$$\mathcal{O} = \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \end{bmatrix} = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix},$$

is nonsingular. Therefore,  $(\mathbf{A}, \mathbf{C})$  is observable.

(b) Let,

$$\mathcal{O}_{unobs} = \begin{bmatrix} \mathbf{C} \\ \mathbf{C}(\mathbf{A} - \mathbf{BK}) \end{bmatrix} = \begin{bmatrix} 1 & 2 \\ -K_1 & 1 - K_2 \end{bmatrix}.$$

So if  $\det(\mathcal{O}_{unobs}) = 1 - K_2 + 2K_1 = 0$ , then  $(\mathbf{A} - \mathbf{BK}, \mathbf{C})$  is unobservable.

(c)  $K_1 = 1 \implies 1 - K_2 + 2 = 0 \implies K_2 = 3$ .

(d)

$$G_{ol}(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} = \frac{s+2}{s^2+2s-1} = \frac{s+2}{(s-0.414)(s+2.414)},$$

$$G_{cl}(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A} + \mathbf{BK})^{-1}\mathbf{B} = \frac{s+2}{s^2+3s+2} = \frac{s+2}{(s+2)(s+1)} = \frac{1}{(s+1)}.$$

The computations can be carried out using MATLAB's `ss2tf` command. So the unobservability is due to a **cancellation** of one of the closed-loop poles with the zero of the system. In other words, this closed-loop mode is unobservable from the output.

**37. Solution:**

(a) Apply Kirchhoff's voltage and current laws, with  $x_1 = i_L$  and  $x_2 = v_c$ , we obtain,

$$\begin{aligned} L\dot{x}_1 + Rx_1 &= x_2 + RC\dot{x}_2, \\ C\dot{x}_2 &= u - x_1, \\ y &= (u - x_1)R \end{aligned}$$

Thus,

$$\begin{aligned} \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} &= \begin{bmatrix} -2R/L & 1/L \\ -1/C & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} R/L \\ 1/C \end{bmatrix} u, \\ y &= \begin{bmatrix} -R & 0 \end{bmatrix} \mathbf{x} + Ru. \end{aligned}$$

(b) The condition for the system to be uncontrollable is  $\det(\mathcal{C}) = 0$ .

$$\begin{aligned} \mathcal{C} &= \begin{bmatrix} \mathbf{B} & \mathbf{AB} \end{bmatrix} = \begin{bmatrix} R/L & -2R^2/L^2 + 1/LC \\ 1/C & -R/LC \end{bmatrix}, \\ \det(\mathcal{C}) &= R^2/L^2C - 1/LC^2. \end{aligned}$$

Thus, the system is controllable if  $R^2 \neq L/C$ .

(c) The condition for the system to be unobservable is,

$$\begin{aligned} \mathcal{O} &= \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \end{bmatrix} = \begin{bmatrix} -R & 0 \\ 2R^2/L & -R/L \end{bmatrix}, \\ \det(\mathcal{O}) &= R^2/L. \end{aligned}$$

Since  $\det(\mathcal{O}) \neq 0$  for any  $R, L, C$  except  $R = 0$  or  $L = \infty$ , the system is observable.

**46. Solution:**

(a) Defining  $x_1 = \theta$  and  $x_2 = \dot{\theta}$ , and anticipating that the measured variable in part (b) is  $\dot{\theta}$ , we have,

$$\begin{aligned} \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} &= \begin{bmatrix} 0 & 1 \\ -\omega^2 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u, \\ y &= \begin{bmatrix} 0 & 1 \end{bmatrix} \mathbf{x}. \end{aligned}$$

(b) From,

$$\begin{aligned} \det(s\mathbf{I} - \mathbf{A} + \mathbf{LC}) &= 0, \\ \det \left\{ \begin{bmatrix} s & 0 \\ 0 & s \end{bmatrix} - \begin{bmatrix} 0 & 1 \\ -\omega^2 & 0 \end{bmatrix} + \begin{bmatrix} l_1 \\ l_2 \end{bmatrix} \begin{bmatrix} 0 & 1 \end{bmatrix} \right\} &= s^2 + l_2s + \omega^2(-l_1 + 1) = 0. \end{aligned}$$

Using  $\omega = 5$  and the specified roots for the estimator, we calculate  $l_1 = -7$ , and  $l_2 = 20$ . This result can be verified using MATLAB's place command.

(c) To find the transfer function from the measured value of  $\dot{\theta}$ ,  $y$ , to the estimated value of  $\theta$ ,  $\hat{\theta}$ , we use the estimator equations,

$$\begin{aligned} \dot{\hat{\mathbf{x}}} &= \mathbf{A}\hat{\mathbf{x}} + \mathbf{B}u + \mathbf{L}(y - \mathbf{C}\hat{\mathbf{x}}) \\ &= (\mathbf{A} - \mathbf{LC})\hat{\mathbf{x}} + \mathbf{B}u + \mathbf{L}y. \end{aligned}$$

Since this is in state space form, we can now directly compute the transfer function from  $y$  to  $\hat{\theta}$ . It is simply,

$$\begin{aligned} \frac{\hat{\Theta}(s)}{Y(s)} &= \begin{bmatrix} 1 & 0 \end{bmatrix} (s\mathbf{I} - \mathbf{A} + \mathbf{LC})^{-1} \mathbf{L} \\ &= \frac{-7(s - 20/7)}{s^2 + 20s + 200}. \end{aligned}$$

(d) For controller gain  $\mathbf{K} = [k_1 \ k_2]$ , we require,

$$\det(s\mathbf{I} - \mathbf{A} + \mathbf{BK}) = 0 \implies s^2 + k_2s + \omega^2 + k_1 = 0.$$

Comparing this with the specified roots equation:

$$(s + 4 + j4)(s + 4 - j4) = s^2 + 8s + 32 = 0,$$

we obtain  $k_1 = 7$ , and  $k_2 = 8$ . This result can be verified using MATLAB's place command.

**Remark:** In part (c), when we look at the state equation of the estimator we note that if  $y$  is the input, then the input matrix will be  $\mathbf{L}$ . And when  $\hat{\theta}$  is the output, the output matrix will be  $\begin{bmatrix} 1 & 0 \end{bmatrix}$  because  $\hat{\theta}$  is the first state of  $\hat{\mathbf{x}}$ . The state matrix will be evidently  $\mathbf{A} - \mathbf{LC}$ .

**48. Solution:**

(a) From the transfer function, we can read off the elements that will give observer canonical form,

$$\begin{aligned}\dot{\mathbf{x}} &= \mathbf{A}_o \mathbf{x} + \mathbf{B}_o u, \\ y &= \mathbf{C}_o \mathbf{x}, \\ \mathbf{A}_o &= \begin{bmatrix} 0 & 1 \\ 4 & 0 \end{bmatrix}, \mathbf{B}_o = \begin{bmatrix} 0 \\ 4 \end{bmatrix}, \mathbf{C}_o = [ 1 \quad 0 ].\end{aligned}$$

(b) With  $u = -[k_1 \ k_2][x_1 \ x_2]^T$ , we want to achieve the following closed-loop characteristic equation:

$$\alpha_c(s) = (s + 2 + 2j)(s + 2 - 2j) = s^2 + 4s + 8 = 0.$$

From  $\det(s\mathbf{I} - \mathbf{A} + \mathbf{BK}) = 0$ , we obtain,

$$s^2 + 4k_2s + 4k_1 - 4 = 0.$$

Comparing the coefficients yields  $k_1 = 3$ , and  $k_2 = 1$ . This result can be verified using MATLAB's `place` command.

(c) The estimator roots are determined by the equation  $\alpha_e(s) = 0$ . We want to find  $l_1$  and  $l_2$  such that,

$$\alpha_e(s) = (s + 10 + 10j)(s + 10 - 10j) = s^2 + 20s + 200.$$

$$\begin{aligned}\alpha_e(s) &= \det(s\mathbf{I} - \mathbf{A} + \mathbf{LC}) \\ &= \det \left( \begin{bmatrix} s & -1 \\ -4 & s \end{bmatrix} + \begin{bmatrix} l_1 \\ l_2 \end{bmatrix} [ 1 \quad 0 ] \right) \\ &= \det \begin{bmatrix} s + l_1 & -1 \\ -4 + l_2 & s \end{bmatrix} = s^2 + l_1s + l_2 - 4.\end{aligned}$$

Comparing the coefficients yields  $l_1 = 20$ ,  $l_2 = 204$ . This result can be verified using MATLAB's `place` command.

(d) The transfer function of the resulting compensator is,

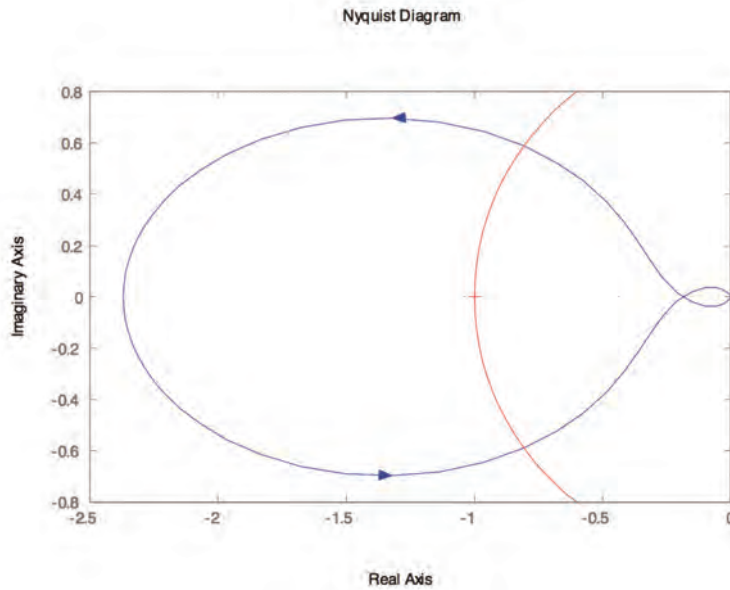
$$\begin{aligned}D_c(s) &= \frac{U(s)}{Y(s)} = -\mathbf{K}(s\mathbf{I} - \mathbf{A} + \mathbf{BK} + \mathbf{LC})^{-1}\mathbf{L}, \\ &= -[ 3 \quad 1 ] \begin{bmatrix} s + 20 & -1 \\ 212 & s + 4 \end{bmatrix}^{-1} \begin{bmatrix} 20 \\ 204 \end{bmatrix} = \frac{-264s - 692}{s^2 + 24s + 292}.\end{aligned}$$

This result can be verified using MATLAB's `ss2tf` command.

(e) The loop gain is given by:

$$-D_c(s)G(s) = \frac{264s + 692}{s^2 + 24s + 292} \frac{4}{s^2 - 4}$$

Note that if we consider a positive feedback and put the negative sign in the controller  $D_c(s)$  then the closed-loop poles are the zeros of  $1 - D_c(s)G(s)$  and the Nyquist plot should be drawn for  $-D_c(s)G(s)$ . We can see the Nyquist plot for this system below from which we notice that the system has both a positive and negative gain margin (in dB). The gain can be increased by 5.46 times ( $GM=1/0.183=5.46$ ), or decreased by 0.41 times ( $GM=1/2.37=0.41$ ) before the system becomes unstable. From the plot, we can also see that the phase margin is about  $55^\circ$ .



Nyquist plot for Problem 7.48.

## 51. Solution

- (a) The ODE of the system is  $\ddot{y} + \dot{y} = 10u$ . Define the state variables  $x_1 = y$  and  $x_2 = \dot{x}_1 = \dot{y}$ , we can give the state equations as

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 10 \end{bmatrix} u$$

$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \mathbf{x}$$

- (b) The desired characteristic equation is:

$$\alpha_c(s) = s^2 + 2\zeta\omega_n s + \omega_n^2 = s^2 + 3s + 9$$

The closed-loop characteristic equation is:

$$\det(s\mathbf{I} - \mathbf{A} + \mathbf{BK}) = 10K_1 + s + 10K_2s + s^2$$

Equating coefficients and solving gives  $K_1 = 0.9$  and  $K_2 = 0.2$ .

(c) The desired characteristic equation is:

$$\alpha_e(s) = s^2 + 2\zeta\omega_n s + \omega_n^2 = s^2 + 15s + 225$$

The closed-loop characteristic equation is:

$$\det(s\mathbf{I} - \mathbf{A} + \mathbf{LC}) = (s + l_1)(s + 1) + l_2 = s^2 + (l_1 + 1)s + (l_1 + l_2)$$

Equating coefficients and solving gives  $l_1 = 14$  and  $l_2 = 211$ .

(d) The transfer function of the controller is:

$$D_c(s) = -\mathbf{K}(s\mathbf{I} - \mathbf{A} + \mathbf{BK} + \mathbf{LC})^{-1}\mathbf{L} = \frac{-(54.8s + 202.5)}{s^2 + 17s + 262}$$

### 58. Solution:

(a) Using feedback of the form,  $u = -\mathbf{K}\mathbf{x} + \bar{N}r$ , we have,

$$\det(s\mathbf{I} - \mathbf{A} + \mathbf{BK}) = (s + 2 + k_1)(s + 3 + k_2) + k_1(1 - k_2) = s^2 + 6s + 18,$$

when  $\mathbf{K} = [5 \ -4]$ . This result can be verified using the MATLAB place command.

(b) We can find the desired value for  $\bar{N}$  by setting the DC gain from  $r$  to  $y$  equal to unity. The closed-loop system equations are,

$$\begin{aligned}\dot{\mathbf{x}} &= \mathbf{A}\mathbf{x} + \mathbf{B}(-\mathbf{K}\mathbf{x} + \bar{N}r) = (\mathbf{A} - \mathbf{BK})\mathbf{x} + \mathbf{B}\bar{N}r, \\ y &= \mathbf{C}\mathbf{x}.\end{aligned}$$

Therefore, the transfer function is,

$$T(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A} + \mathbf{BK})^{-1}\mathbf{B}\bar{N},$$

and the DC gain is simply,

$$T(0) = \mathbf{C}(-\mathbf{A} + \mathbf{BK})^{-1}\mathbf{B}\bar{N} = \frac{5}{9}\bar{N} = 1.$$

Hence, we choose  $\bar{N} = \frac{9}{5}$ .

(c) Change  $\mathbf{A}$  to  $(\mathbf{A} + \delta\mathbf{A})$ , and let the value of  $\bar{N}$  that keeps the tracking error at zero be  $N'$ . Then letting  $T'(s)$  be the transfer function associated with the perturbed system,

$$\begin{aligned}N'^{-1} &= T'(0) = -\mathbf{C}(\mathbf{A} + \delta\mathbf{A} - \mathbf{BK})^{-1}\mathbf{B}, \\ &= -\mathbf{C}[(\mathbf{A} - \mathbf{BK})(\mathbf{I} - (\mathbf{A} - \mathbf{BK})^{-1}\delta\mathbf{A})]^{-1}\mathbf{B}, \\ &= -\mathbf{C}(\mathbf{I} - (\mathbf{A} - \mathbf{BK})^{-1}\delta\mathbf{A})^{-1}(\mathbf{A} - \mathbf{BK})^{-1}\mathbf{B}.\end{aligned}$$

For  $\delta\mathbf{A}$  small,

$$(\mathbf{I} - (\mathbf{A} - \mathbf{BK})^{-1}\delta\mathbf{A})^{-1} = \mathbf{I} + (\mathbf{A} - \mathbf{BK})^{-1}\delta\mathbf{A}.$$

Hence,

$$N'^{-1} = \underbrace{-\mathbf{C}(\mathbf{A} - \mathbf{BK})^{-1}\mathbf{B}}_{\bar{N}^{-1}} - \mathbf{C}(\mathbf{A} - \mathbf{BK})^{-1}\delta\mathbf{A}(\mathbf{A} - \mathbf{BK})^{-1}\mathbf{B}.$$

And for arbitrary  $\delta\mathbf{A}$  we arrive at,

$$N'^{-1} \neq \bar{N}^{-1}.$$

Therefore, small changes in the plant matrix  $\mathbf{A}$  prevent the steady-state error from reaching zero. The control system is not robust with respect to changes in  $\mathbf{A}$ .

(d) Augmenting the system equations with an integrator state,  $x_I$ , the state equation become,

$$\begin{aligned} \begin{bmatrix} \dot{\mathbf{x}} \\ \dot{x}_I \end{bmatrix} &= \begin{bmatrix} \mathbf{A} & \mathbf{0} \\ -\mathbf{C} & 0 \end{bmatrix} \begin{bmatrix} \mathbf{x} \\ x_I \end{bmatrix} + \begin{bmatrix} \mathbf{B} \\ 0 \end{bmatrix} u + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r, \\ y &= [\mathbf{C} \ 0] \begin{bmatrix} \mathbf{x} \\ x_I \end{bmatrix}. \end{aligned}$$

or with  $\mathbf{z} = [\mathbf{x} \ x_I]^T$ ,

$$\begin{aligned} \dot{\mathbf{z}} &= \mathbf{A}_a \mathbf{z} + \mathbf{B}_a u + \mathbf{B}_r r, \\ y &= \mathbf{C}_a \mathbf{z}. \end{aligned}$$

Using feedback of the form  $u = -\mathbf{K}\mathbf{x} - k_I x_I = -\mathbf{K}_a \mathbf{z}$ , we have,

$$\det(s\mathbf{I} - \mathbf{A}_a + \mathbf{B}_a \mathbf{K}_a) = 0 \text{ for } s = -3, -2 \pm j\sqrt{3},$$

when  $\mathbf{K}_a = [0.3 \ 1.7 \ -2.1]$ . This result can be verified using the MATLAB place command.

(e) We can show that the closed-loop DC gain from  $r$  to  $y$  is independent of  $\mathbf{A}$ ,

$$\begin{aligned} y_\infty &= T(0)r_\infty = [\mathbf{C} \ 0] \begin{bmatrix} -\mathbf{A} + \mathbf{BK} & \mathbf{B}k_I \\ \mathbf{C} & 0 \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ 1 \end{bmatrix} r_\infty \\ &= [\mathbf{C} \ 0] \begin{bmatrix} * & (\mathbf{A} - \mathbf{BK})^{-1} \mathbf{B}k_I [\mathbf{C}(\mathbf{A} - \mathbf{BK})^{-1} \mathbf{B}k_I]^{-1} \\ * & * \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} r_\infty \\ &= [\mathbf{C}(\mathbf{A} - \mathbf{BK})^{-1} \mathbf{B}k_I] [\mathbf{C}(\mathbf{A} - \mathbf{BK})^{-1} \mathbf{B}k_I]^{-1} r_\infty = r_\infty \text{ independent of } \mathbf{A}, \mathbf{B}, \mathbf{C}. \end{aligned}$$

**Remark:** In the second line of the above equations, the following matrix inversion lemma is used :

$$\begin{aligned} \begin{bmatrix} \mathbf{A} & \mathbf{U} \\ \mathbf{V} & \mathbf{C} \end{bmatrix}^{-1} &= \begin{bmatrix} \mathbf{I} & \mathbf{A}^{-1}\mathbf{U} \\ 0 & \mathbf{I} \end{bmatrix}^{-1} \begin{bmatrix} \mathbf{A} & 0 \\ 0 & \mathbf{C} - \mathbf{V}\mathbf{A}^{-1}\mathbf{U} \end{bmatrix}^{-1} \begin{bmatrix} \mathbf{I} & 0 \\ \mathbf{V}\mathbf{A}^{-1} & \mathbf{I} \end{bmatrix}^{-1} \\ &= \begin{bmatrix} \mathbf{I} & -\mathbf{A}^{-1}\mathbf{U} \\ 0 & \mathbf{I} \end{bmatrix} \begin{bmatrix} \mathbf{A}^{-1} & 0 \\ 0 & (\mathbf{C} - \mathbf{V}\mathbf{A}^{-1}\mathbf{U})^{-1} \end{bmatrix} \begin{bmatrix} \mathbf{I} & 0 \\ -\mathbf{V}\mathbf{A}^{-1} & \mathbf{I} \end{bmatrix} \\ &= \begin{bmatrix} \mathbf{A}^{-1} + \mathbf{A}^{-1}\mathbf{U}(\mathbf{C} - \mathbf{V}\mathbf{A}^{-1}\mathbf{U})^{-1}\mathbf{V}\mathbf{A}^{-1} & -\mathbf{A}^{-1}\mathbf{U}(\mathbf{C} - \mathbf{V}\mathbf{A}^{-1}\mathbf{U})^{-1} \\ -(\mathbf{C} - \mathbf{V}\mathbf{A}^{-1}\mathbf{U})^{-1}\mathbf{V}\mathbf{A}^{-1} & (\mathbf{C} - \mathbf{V}\mathbf{A}^{-1}\mathbf{U})^{-1} \end{bmatrix} \end{aligned}$$

## 60. Solution

1. The state-space model is given by:

$$\mathbf{A} = \begin{bmatrix} 0 & 0 \\ 1 & 0 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} 1 \\ 0 \end{bmatrix}, \quad \mathbf{C} = [1 \quad \alpha], \quad D = 0$$

The observability matrix is:

$$\mathcal{O} = \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \end{bmatrix} = \begin{bmatrix} 1 & \alpha \\ \alpha & 0 \end{bmatrix} \Rightarrow \det(\mathcal{O}) = -\alpha^2 \neq 0, \quad \text{iff } \alpha \neq 0$$

2. We should find  $\mathbf{P} = \mathbf{P}^T > 0$  from the following equation:

$$\mathbf{A}^T \mathbf{P} + \mathbf{P} \mathbf{A} - \mathbf{P} \mathbf{B} \mathbf{R}^{-1} \mathbf{B}^T \mathbf{P} + \mathbf{Q} = 0$$

$$\begin{aligned} \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} P_{11} & P_{12} \\ P_{12} & P_{22} \end{bmatrix} + \begin{bmatrix} P_{11} & P_{12} \\ P_{12} & P_{22} \end{bmatrix} \begin{bmatrix} 0 & 0 \\ 1 & 0 \end{bmatrix} - \begin{bmatrix} P_{11} & P_{12} \\ P_{12} & P_{22} \end{bmatrix} \begin{bmatrix} 1 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} P_{11} & P_{12} \\ P_{12} & P_{22} \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} &= 0 \\ \Rightarrow \begin{bmatrix} P_{12} & P_{22} \\ 0 & 0 \end{bmatrix} + \begin{bmatrix} P_{12} & 0 \\ P_{22} & 0 \end{bmatrix} - \begin{bmatrix} P_{11} & 0 \\ P_{12} & 0 \end{bmatrix} \begin{bmatrix} P_{11} & P_{12} \\ P_{12} & P_{22} \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} &= 0 \\ \Rightarrow \begin{bmatrix} 2P_{12} & P_{22} \\ P_{22} & 0 \end{bmatrix} - \begin{bmatrix} P_{11}^2 & P_{11}P_{12} \\ P_{12}P_{11} & P_{12}^2 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} &= 0 \end{aligned}$$

$$-P_{12}^2 + 1 = 0 \quad \Rightarrow \quad P_{12} = 1$$

$$2P_{12} - P_{11}^2 + 1 = 0 \quad \Rightarrow \quad P_{11} = \sqrt{3}$$

$$P_{22} - P_{11}P_{12} = 0 \quad \Rightarrow \quad P_{22} = \sqrt{3}$$

Then  $\mathbf{K} = \mathbf{R}^{-1} \mathbf{B}^T \mathbf{P} = [\sqrt{3} \quad 1]$ .

3. The characteristic polynomial is:  $\alpha_o(s) = (s+3)^2 = s^2 + 6s + 9$ . We have:

$$\det(s\mathbf{I} - \mathbf{A} + \mathbf{L}\mathbf{C}) = \alpha_o(s)$$

Therefore:

$$\begin{aligned} \det \left( \begin{bmatrix} s & 0 \\ 0 & s \end{bmatrix} - \begin{bmatrix} 0 & 0 \\ 1 & 0 \end{bmatrix} + \begin{bmatrix} l_1 & l_1 \\ l_2 & l_2 \end{bmatrix} \right) &= s^2 + 6s + 9 \\ \det \left( \begin{bmatrix} s+l_1 & l_1 \\ l_2-1 & s+l_2 \end{bmatrix} \right) &= (s+l_1)(s+l_2) + l_1(1-l_2) = s^2 + 6s + 9 \end{aligned}$$

which leads to  $\mathbf{L}^T = [9 \quad -3]$ .

## 61. Solution

a) The controllable canonical representation is given by:

$$A = \begin{bmatrix} 0 & 0 \\ 1 & 0 \end{bmatrix} \quad B = \begin{bmatrix} 1 \\ 0 \end{bmatrix} \quad C = [ 1 \quad 10 ]$$

The augmented system with integrator is defined as:

$$\bar{A} = \begin{bmatrix} A & 0 \\ -C & 0 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 0 \\ 1 & 0 & 0 \\ -1 & -10 & 0 \end{bmatrix} \quad \bar{B} = \begin{bmatrix} B \\ 0 \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix}$$

We should solve  $\det(sI - \bar{A} + \bar{B}K) = (s + 2)^3$  to find the state feedback controller  $K = [k_1 \quad k_2 \quad k_3]$ .

$$\det \left( \begin{bmatrix} s & 0 & 0 \\ -1 & s & 0 \\ 1 & 10 & s \end{bmatrix} + \begin{bmatrix} k_1 & k_2 & k_3 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \right) = \det \begin{bmatrix} s+k_1 & k_2 & k_3 \\ -1 & s & 0 \\ 1 & 10 & s \end{bmatrix} = (s+k_1)s^2 + k_2s - 10k_3 - k_3s$$

Therefore  $(s+k_1)s^2 + k_2s - 10k_3 - k_3s = s^3 + 6s^2 + 12s + 8$  that leads to  $k_1 = 6, k_2 = 11.2$  and  $k_3 = -0.8$ .  
If we use an observer canonical representation we will compute  $k_1 = 1.12, k_2 = 0.488$  and  $k_3 = -0.8$ .

b) The closed-loop state space equations are:

$$\begin{aligned} \dot{x}(t) &= Ax(t) + Bu(t) = Ax(t) - B[k_1 \quad k_2]x(t) - Bk_3x_I(t) + Br(t) \\ \dot{x}_I(t) &= r(t) - y(t) = r(t) - Cx(t) - w(t) \\ u(t) &= -[k_1 \quad k_2]x(t) - k_3x_I(t) + r(t) \\ y(t) &= Cx(t) + w(t) \end{aligned}$$

$$\begin{bmatrix} \dot{x}(t) \\ \dot{x}_I(t) \end{bmatrix} = \begin{bmatrix} A - B[k_1 \quad k_2] & -Bk_3 \\ -C & 0 \end{bmatrix} \begin{bmatrix} x(t) \\ x_I(t) \end{bmatrix} + \begin{bmatrix} B \\ 1 \end{bmatrix} r(t) + \begin{bmatrix} 0 \\ -1 \end{bmatrix} w(t)$$

$$A_{cl} = \bar{A} - \bar{B}K = \begin{bmatrix} -k_1 & -k_2 & -k_3 \\ 1 & 0 & 0 \\ -1 & -10 & 0 \end{bmatrix}, \quad B_{cl} = \begin{bmatrix} 1 \\ 0 \\ 1 \end{bmatrix}, \quad B'_{cl} = \begin{bmatrix} 0 \\ 0 \\ -1 \end{bmatrix}$$

- Between  $w(t)$  and  $y(t)$ :  $(A_{cl}, B'_{cl}, [C \quad 0], 1)$
- Between  $r(t)$  and  $u(t)$ :  $(A_{cl}, B_{cl}, -K, 1)$