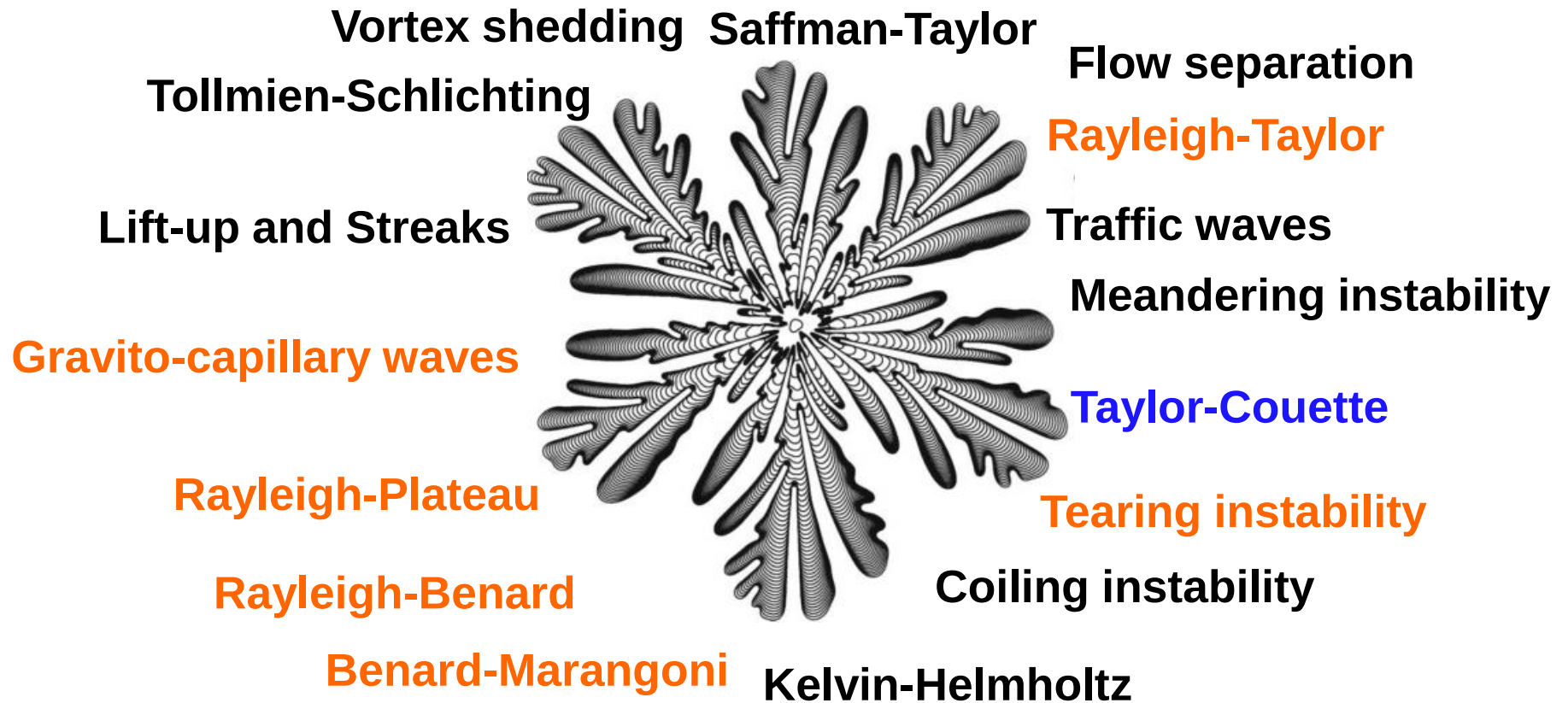
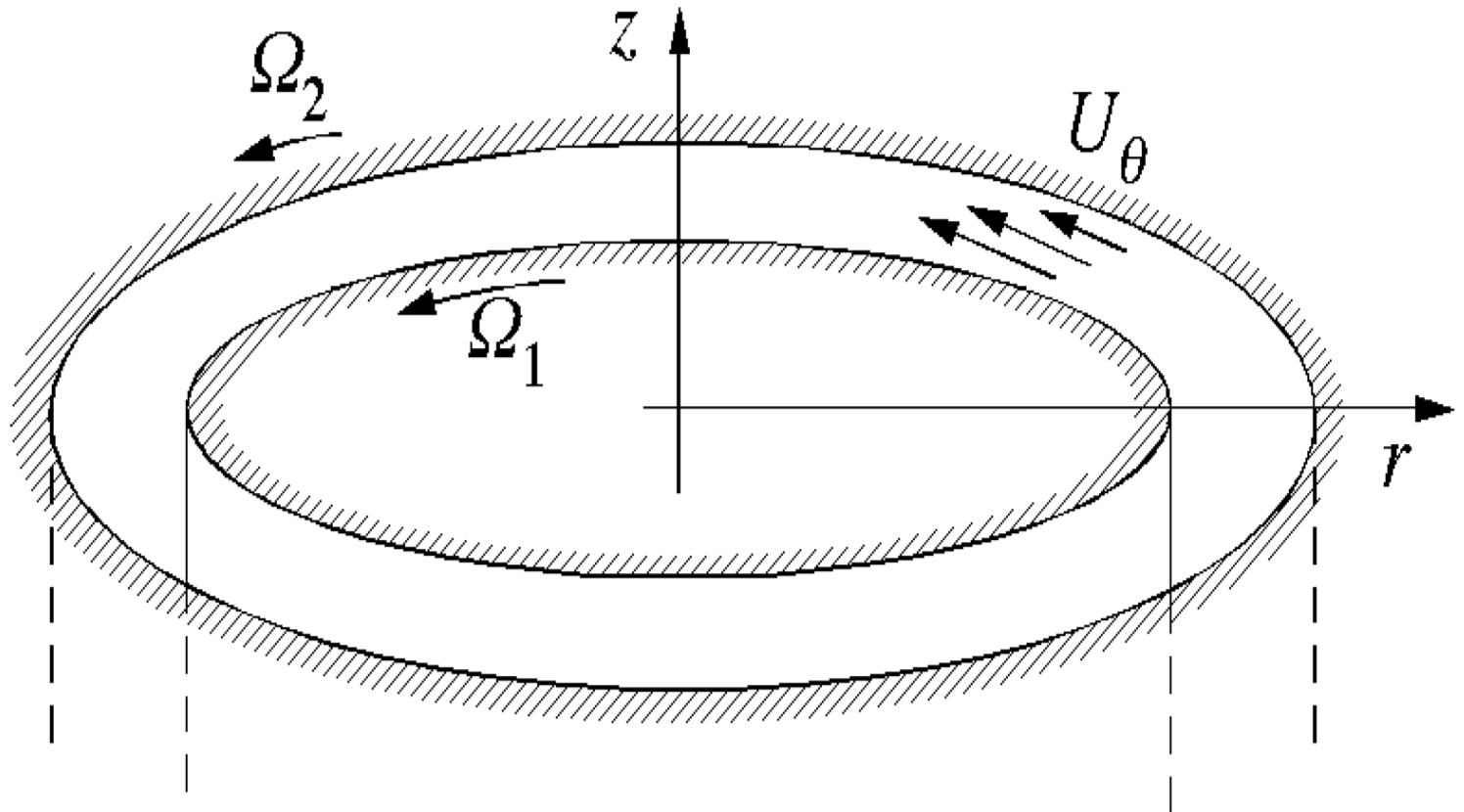


Most flows are unstable...



Taylor Couette

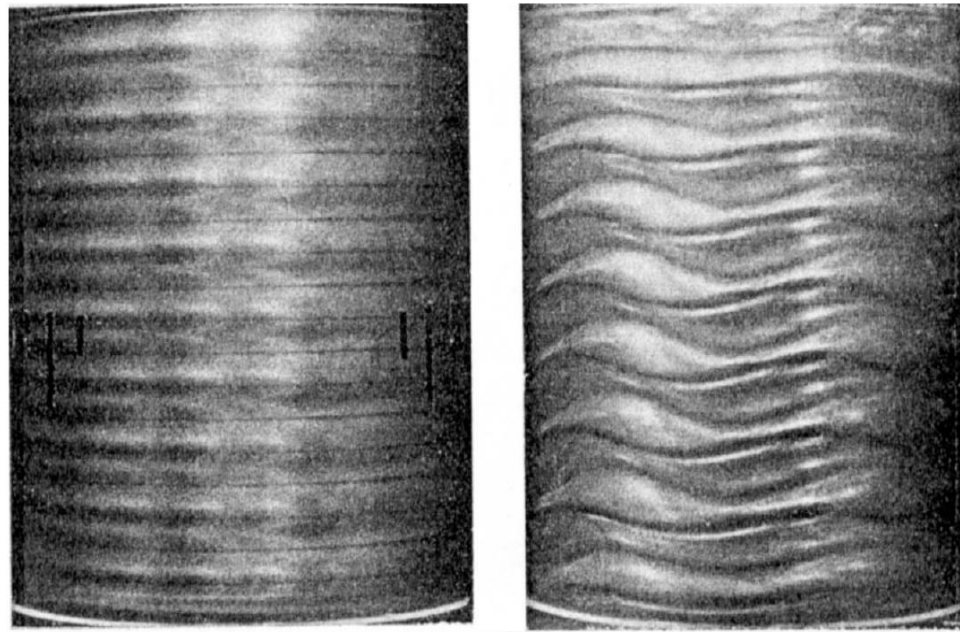


Taylor Couette

Movie by Garcia, Chomaz, Huerre, LadHyX, France

Taylor Couette

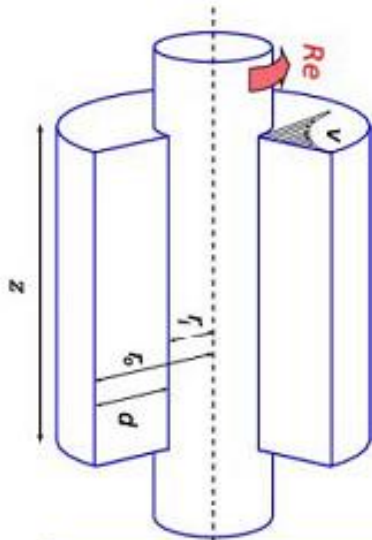
Movie by Garcia, Chomaz, Huerre, LadHyX, France



– *Rouleaux annulaires de Taylor. (a) $Ta/Ta_c = 1.1$; (b) $Ta/Ta_c = 6.0$, rouleaux ondulants apparus suite à une instabilité secondaire ($\lambda = 2\pi R/4$). (Fenstermacher, Swinney & Gollub 1979).*

Taylor Couette

Cascade of instabilities



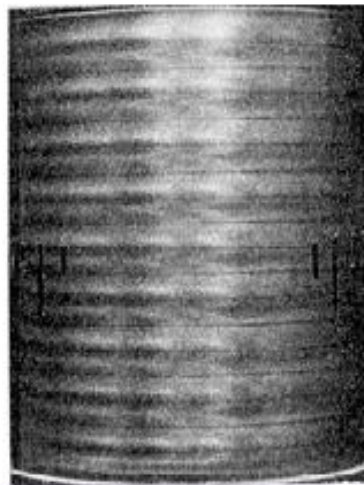
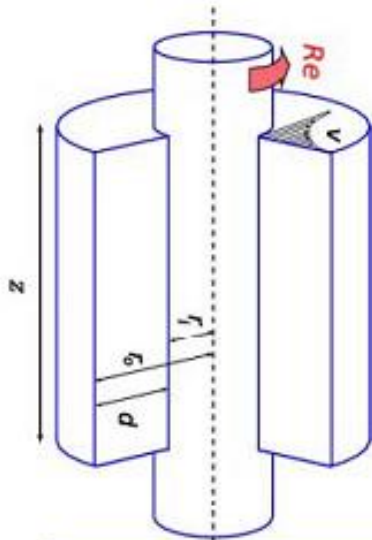
$$d/dt=0; d/d\theta=0; d/dz=0;$$

Ω

Taylor Couette

Cascade of instabilities

Taylor vortices



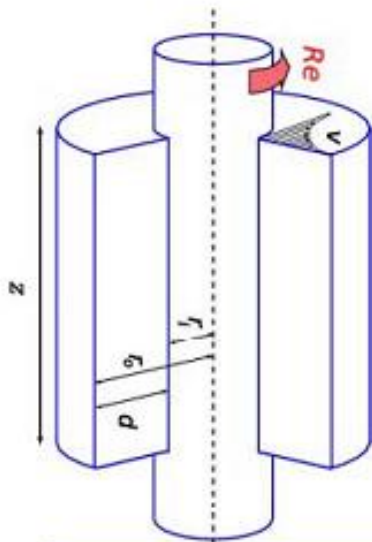
$d/dt=0; d/d\theta=0; d/dz=0;$

$d/dt=0; d/d\theta=0; d/d\mathbf{x}=0;$

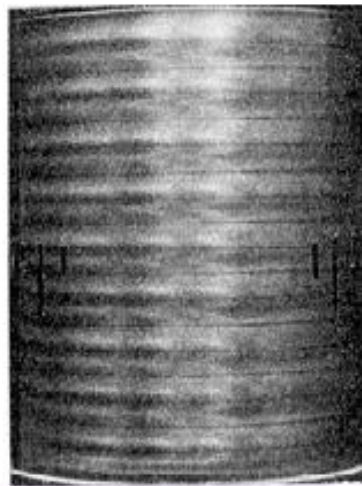
Ω

Taylor Couette

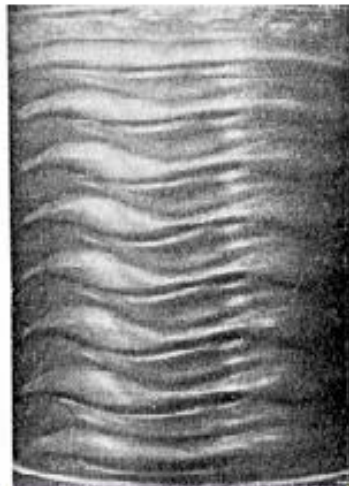
Cascade of instabilities



Taylor vortices



Ondulated vortices



$d/dt=0; d/d\theta=0; d/dz=0;$

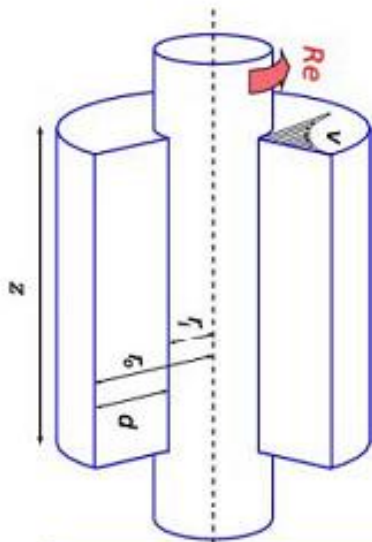
$d/dt=0; d/d\theta=0; d/d\phi=0;$

$d/dt=0; d/d\phi=0; d/d\psi=0;$

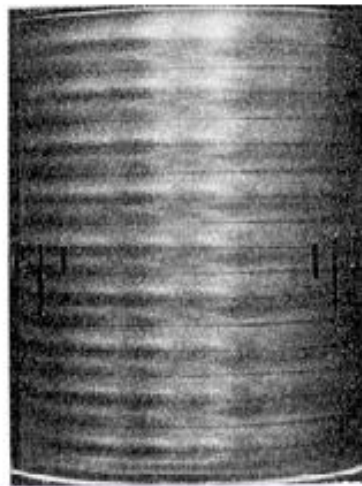
Ω

Taylor Couette

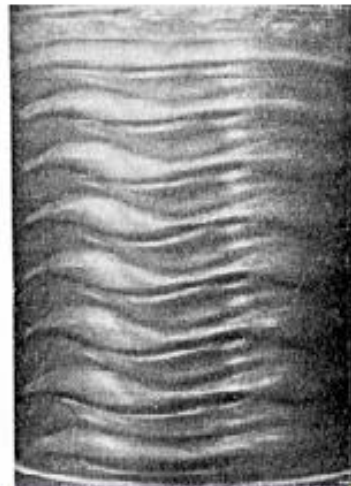
Cascade of instabilities



Taylor vortices



Ondulated vortices



Wavy vortices



..turbulence

$d/dt=0; d/d\theta=0; d/dz=0;$

$d/dt=0; d/d\theta=0; d/d\phi=0;$

$d/dt=0; d/d\phi=0; d/d\psi=0;$

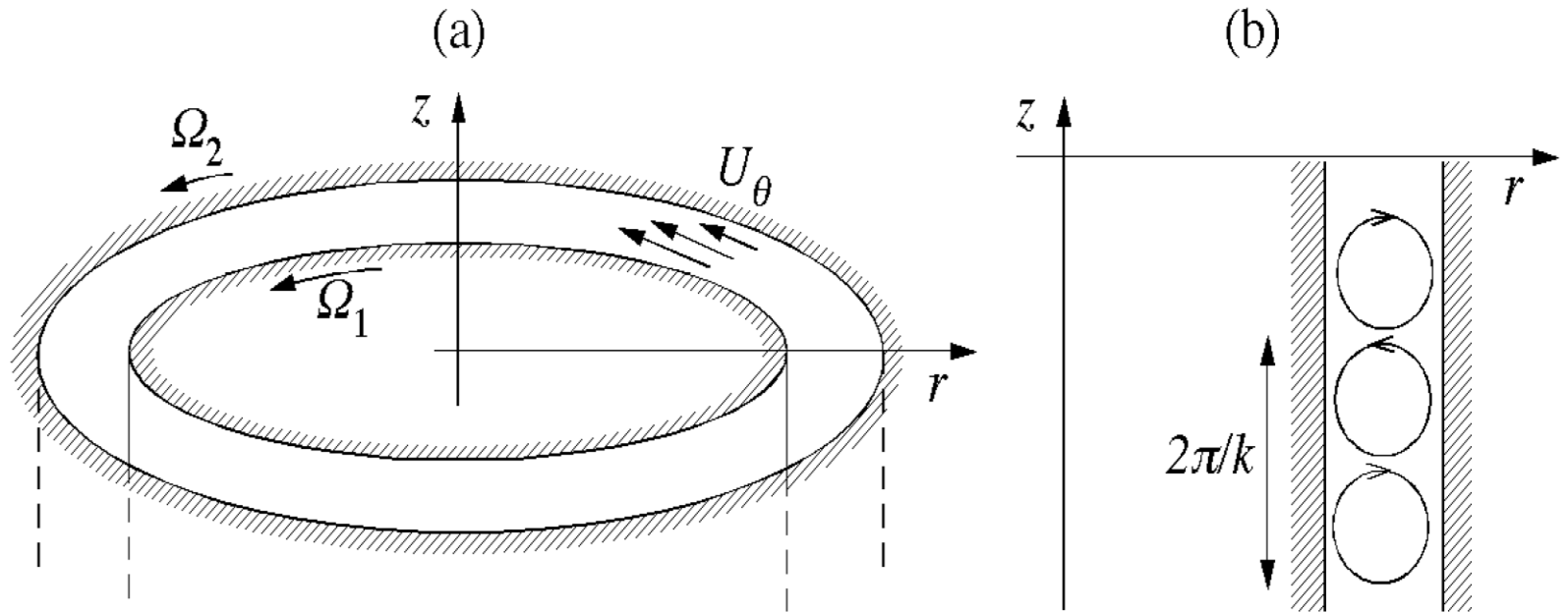
$d/d\psi=0; d/d\chi=0; d/d\eta=0;$

Ω

Instability analysis:

1. Physical mechanism
2. Equations and boundary conditions
3. Base state
4. Linearized equations
5. Normal mode expansion
6. Dispersion relation
7. Analysis of the dispersion relation

Taylor Couette



Navier-Stokes

$$\begin{aligned}
 \underbrace{\rho \left(\frac{Du_r}{Dt} - \frac{u_\theta^2}{r} \right)} &= \underbrace{-\frac{\partial p}{\partial r}} + \underbrace{\mu \left(\nabla^2 u_r - \frac{2}{r^2} \frac{\partial u_\theta}{\partial \theta} - \frac{u_r}{r^2} \right)} \\
 \rho \left(\frac{Du_\theta}{Dt} - \frac{u_r u_\theta}{r} \right) &= -\frac{1}{r} \frac{\partial p}{\partial \theta} + \mu \left(\nabla^2 u_\theta + \frac{2}{r^2} \frac{\partial u_r}{\partial \theta} - \frac{u_\theta}{r^2} \right) \\
 \rho \frac{Du_z}{Dt} &= -\frac{\partial p}{\partial z} + \mu \nabla^2 u_z
 \end{aligned}$$

$$\frac{D}{Dt} = \left(\frac{\partial}{\partial t} + u_r \frac{\partial}{\partial r} + \frac{u_\theta}{r} \frac{\partial}{\partial \theta} + u_z \frac{\partial}{\partial z} \right) \quad \text{convective derivative}$$

$$\nabla^2 = \left(\frac{1}{r} \frac{\partial}{\partial r} + \frac{\partial^2}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2} + \frac{\partial^2}{\partial z^2} \right) \quad \text{Laplacian}$$

Navier-Stokes

inertia

pressure stress

viscous stress

$$\begin{aligned}\rho \left(\frac{Du_r}{Dt} - \frac{u_\theta^2}{r} \right) &= -\frac{\partial p}{\partial r} + \mu \left(\nabla^2 u_r - \frac{2}{r^2} \frac{\partial u_\theta}{\partial \theta} - \frac{u_r}{r^2} \right) \\ \rho \left(\frac{Du_\theta}{Dt} - \frac{u_r u_\theta}{r} \right) &= -\frac{1}{r} \frac{\partial p}{\partial \theta} + \mu \left(\nabla^2 u_\theta + \frac{2}{r^2} \frac{\partial u_r}{\partial \theta} - \frac{u_\theta}{r^2} \right) \\ \rho \frac{Du_z}{Dt} &= -\frac{\partial p}{\partial z} + \mu \nabla^2 u_z\end{aligned}$$

$$\frac{D}{Dt} = \left(\frac{\partial}{\partial t} + u_r \frac{\partial}{\partial r} + \frac{u_\theta}{r} \frac{\partial}{\partial \theta} + u_z \frac{\partial}{\partial z} \right) \quad \text{convective derivative}$$

$$\nabla^2 = \left(\frac{1}{r} \frac{\partial}{\partial r} + \frac{\partial^2}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2} + \frac{\partial^2}{\partial z^2} \right) \quad \text{Laplacian}$$

Rayleigh criterion

Rayleigh criterion: A necessary and sufficient condition for stability of the $\{O, U_\theta(r), 0\}$ basic flow ($r \in]R_1, R_2[$) to inviscid axisymmetric perturbations, is that $\partial (rU_\theta)^2 / \partial r > 0$ everywhere in $[R_1, R_2]$.

Inviscid flow

steady flow

axisymmetric flow

parallel flow

unidirectional flow

Rayleigh criterion

Rayleigh criterion: A necessary and sufficient condition for stability of the $\{O, U_\theta(r), 0\}$ basic flow ($r \in]R_1, R_2[$) to inviscid axisymmetric perturbations, is that $\partial (rU_\theta)^2 / \partial r > 0$ everywhere in $[R_1, R_2]$.

Inviscid flow	$\mu = 0$
steady flow	$\partial / \partial t = 0$
axisymmetric flow	$\partial / \partial \theta = 0$
parallel flow	$\partial / \partial z = 0$
unidirectional flow	$\{0, U_\theta(r), 0\}$

$$\rho U_\theta^2 / r = \partial P / \partial r$$







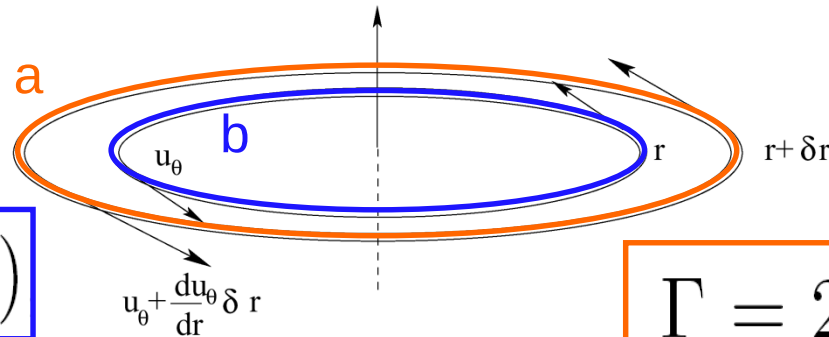
Aspirateurs à force centrifuge, Dyson



Mechanism for Rayleigh Criterion

Let us move an annulus of flow at r_b to another location $r_a > r_b$

Velocity profile before moving annulus



$$\Gamma = 2\pi r_b U_\theta(r_b)$$

$$\Gamma = 2\pi r_a u_\theta(r_a)$$

Kelvin's theorem:

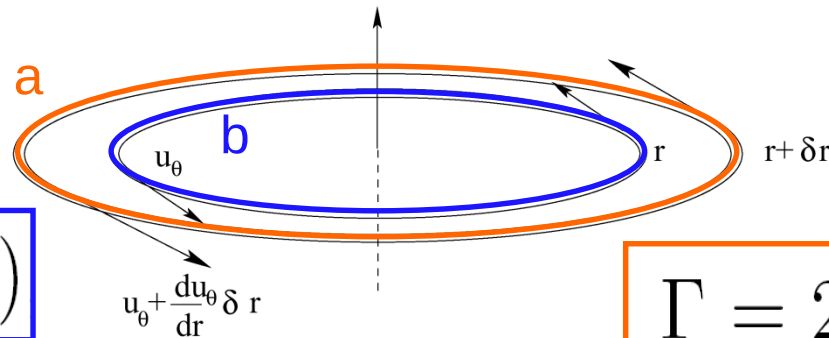
$$u_\theta(r_a) = (r_b/r_a) U_\theta(r_b)$$

Velocity after moving annulus

Mechanism for Rayleigh Criterion

Let us move an annulus of flow at r_b to another location r_a

Velocity profile before moving annulus



$$\Gamma = 2\pi r_b U_\theta(r_b)$$

$$\Gamma = 2\pi r_a u_\theta(r_a)$$

Kelvin's theorem:

$$u_\theta(r_a) = (r_b/r_a) U_\theta(r_b)$$

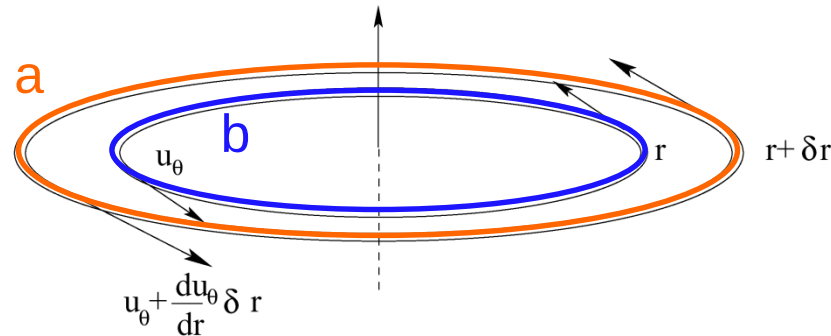
Velocity after moving annulus

Centrifugal force/ Pressure balance

$$\rho U_\theta^2 / r = \partial P / \partial r$$

Centrifugal instabilities

Mechanism and Rayleigh criterion 1916



if $\overbrace{\rho u_{\theta}^2(r_a)}^{\text{new velocity}} / r_a > (dP/dr)(r_a) = \rho U_{\theta}^2(r_a) / r_a$

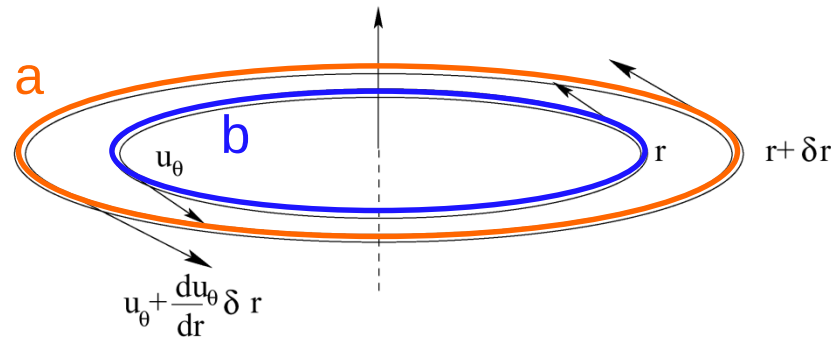
centrifugal force $>$ pressure gradient

$\Rightarrow$ The annulus further escapes towards high r

$\Rightarrow$ UNSTABLE

Centrifugal instabilities

Mechanism and Rayleigh criterion 1916



if $\overbrace{\rho u_{\theta}^2(r_a)/r_a}^{\text{new velocity}} < (dP/dr)(r_a) = \rho U_{\theta}^2(r_a)/r_a$

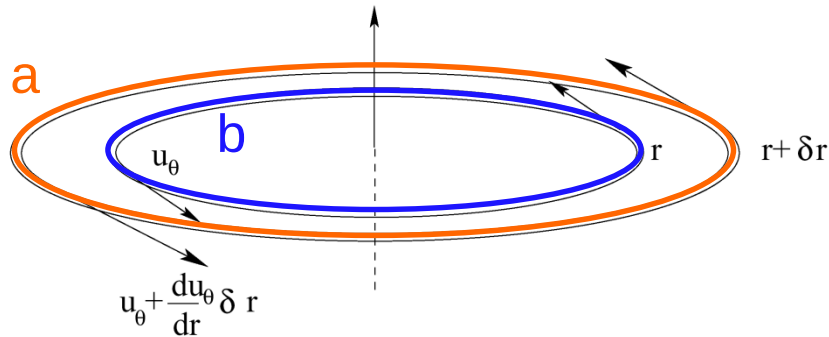
centrifugal force < pressure gradient

⇒ The annulus is brought back to its initial position

⇒ STABLE

Centrifugal instabilities

Mechanism and Rayleigh criterion 1916



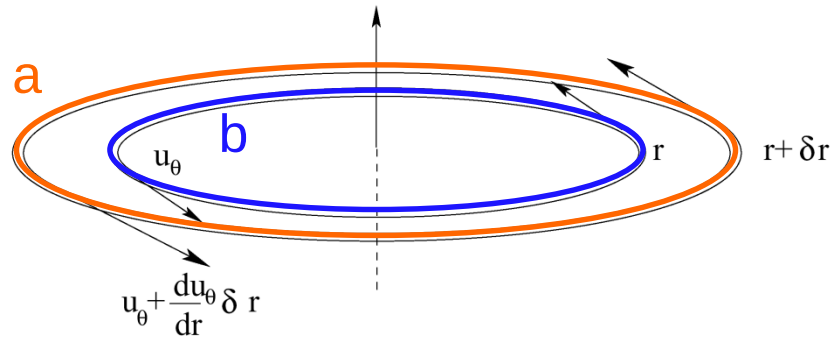
Condition for stability

$$U_{\theta}^2(r_b)r_b^2 < U_{\theta}^2(r_a)r_a^2$$

$$d(rU_{\theta})^2 / dr > 0$$

Centrifugal instabilities

Mechanism and Rayleigh criterion 1916



Condition for stability

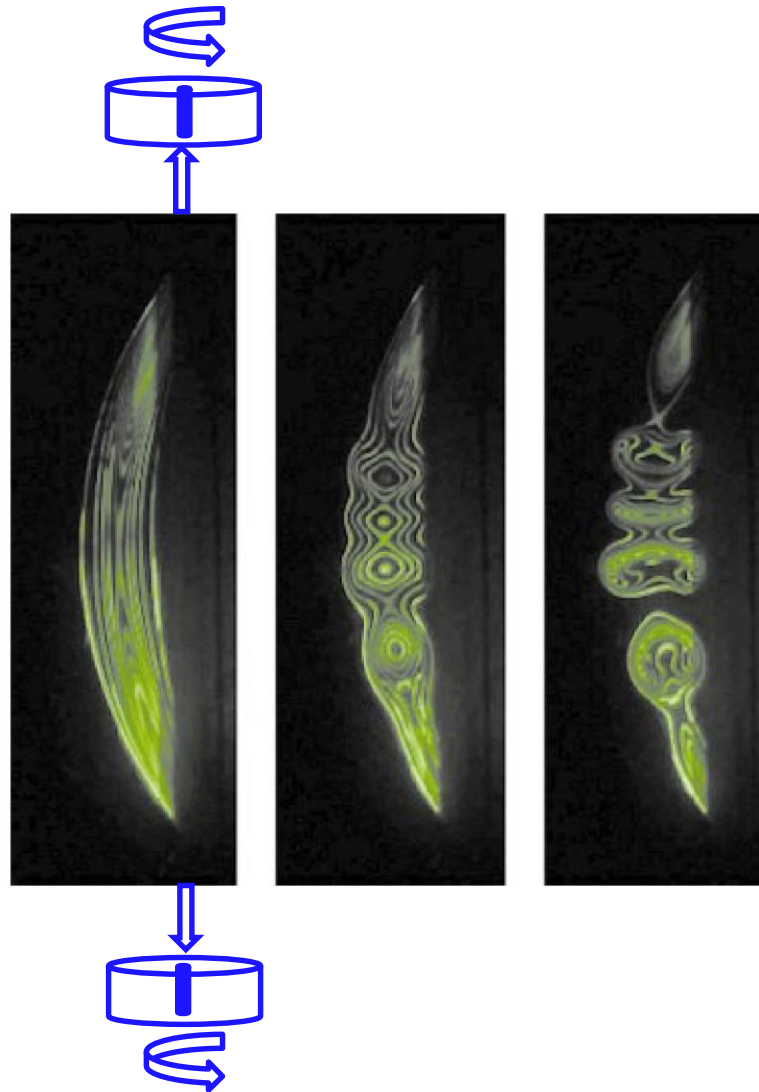
$$U_{\theta}^2(r_b)r_b^2 < U_{\theta}^2(r_a)r_a^2$$

$$d(rU_{\theta})^2/dr > 0$$

$$U_{\theta}\zeta > 0$$

axial vorticity
 $\zeta = 1/r \, d(ru_{\theta})/dr$

Centrifugal instability of a vortex



Bottausci & Petitjean 2002

Anticyclone/Cyclone

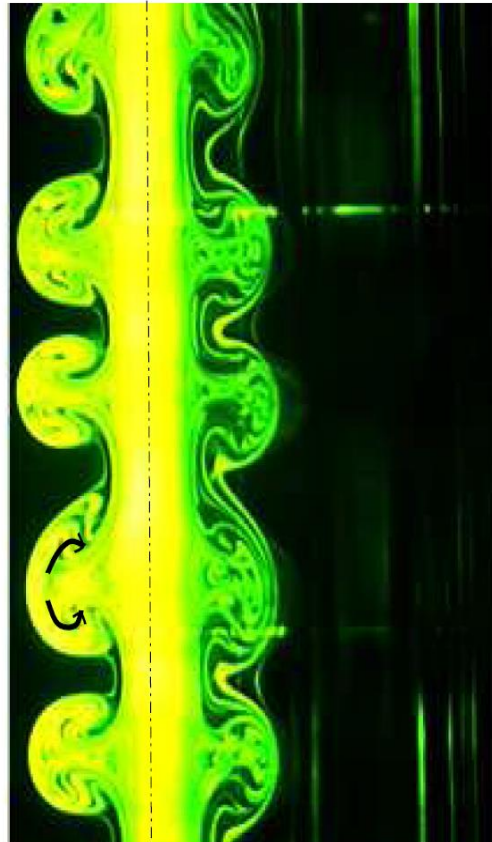
Condition for instability in presence of background rotation Ω

$$(\Omega + u_\theta/r) (2\Omega + \zeta) > 0$$

axial vorticity
 $\zeta = 1/r \, d(r u_\theta)/dr$

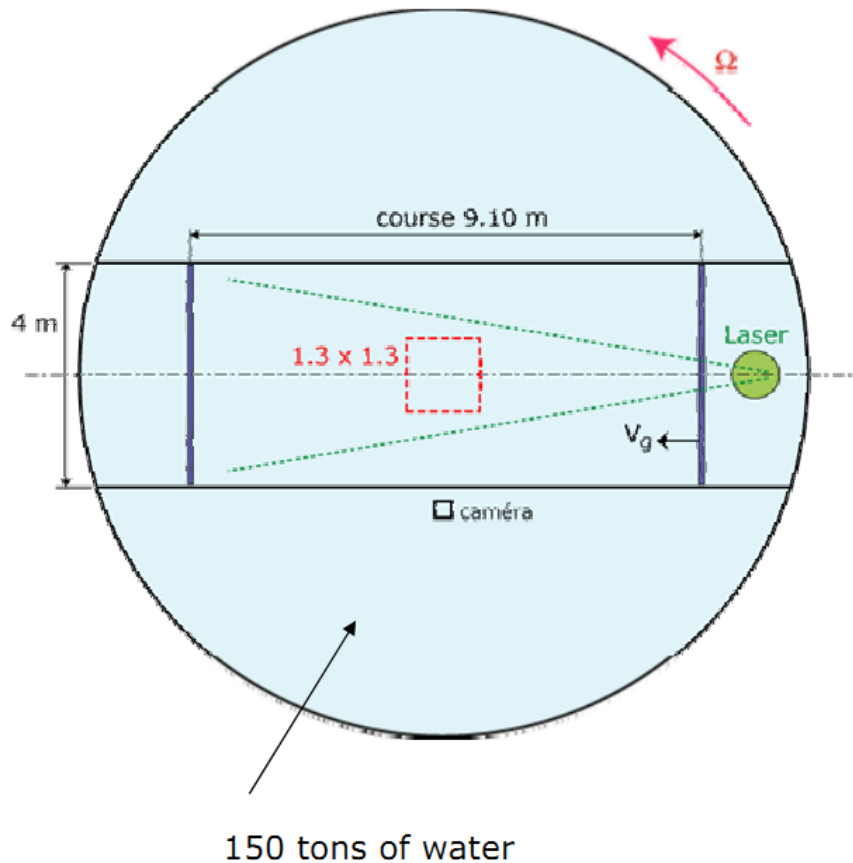
Anticyclone/Cyclone

Anticyclone Cyclone



Fontane et al. (2002)

Experimental setup: 'Coriolis' Rotating Platform (LEGI, Grenoble)



9 m x 4 m x 1 m channel

Grid (of square mesh $M = 15$ cm),
translated at $V_g = 0.3$ m s⁻¹

mounted on the **13 m** diameter
'Coriolis' rotating platform

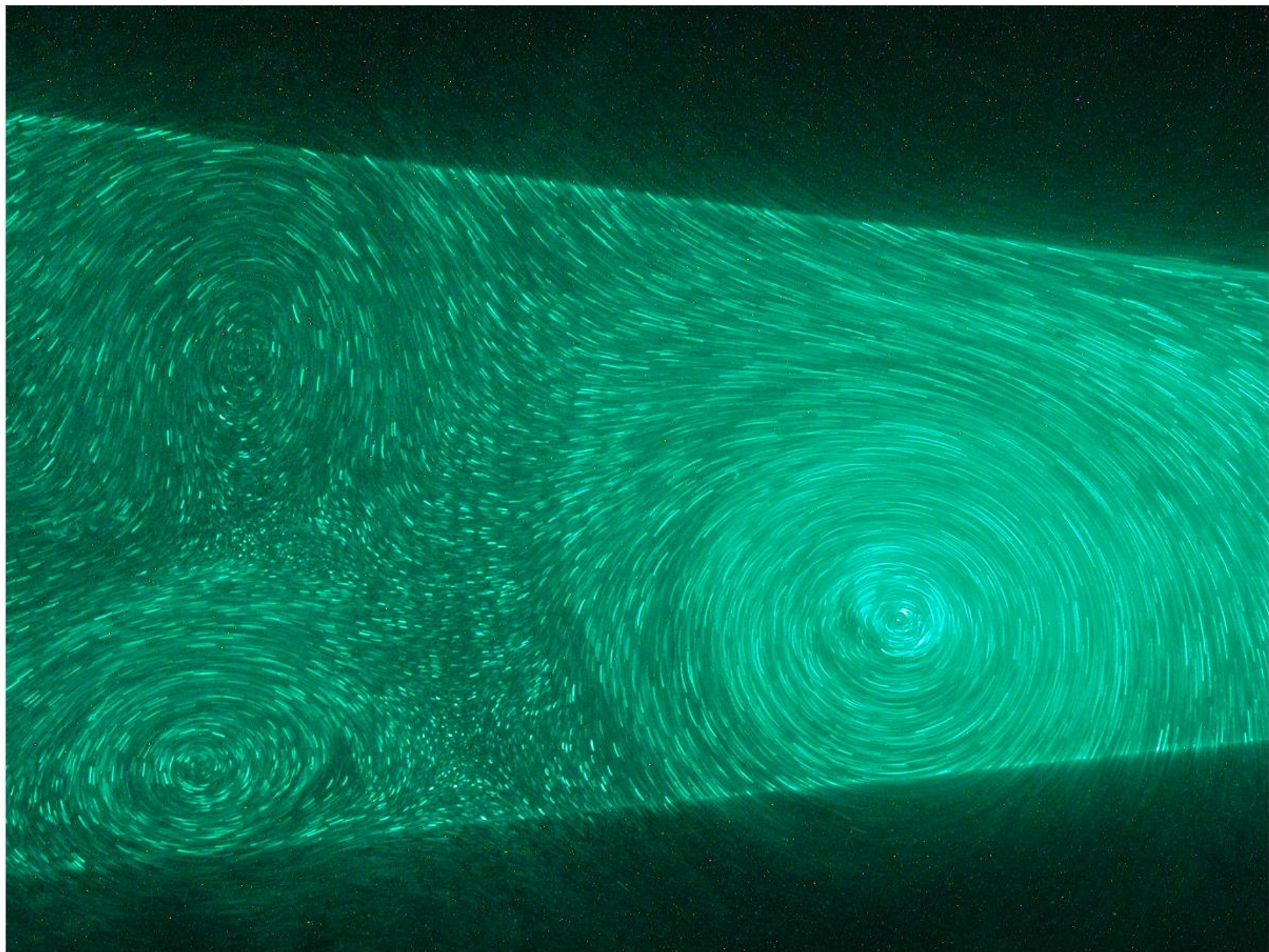
Rotation periods: $T = 30, 60, 120$ s
1 decay ~ 1 hour $\sim 10^4$ M/V_g

PIV measurements in horizontal
and vertical planes
2000x2000 HR camera

Experimental setup: 'Coriolis' Rotating Platform (LEGI, Grenoble)

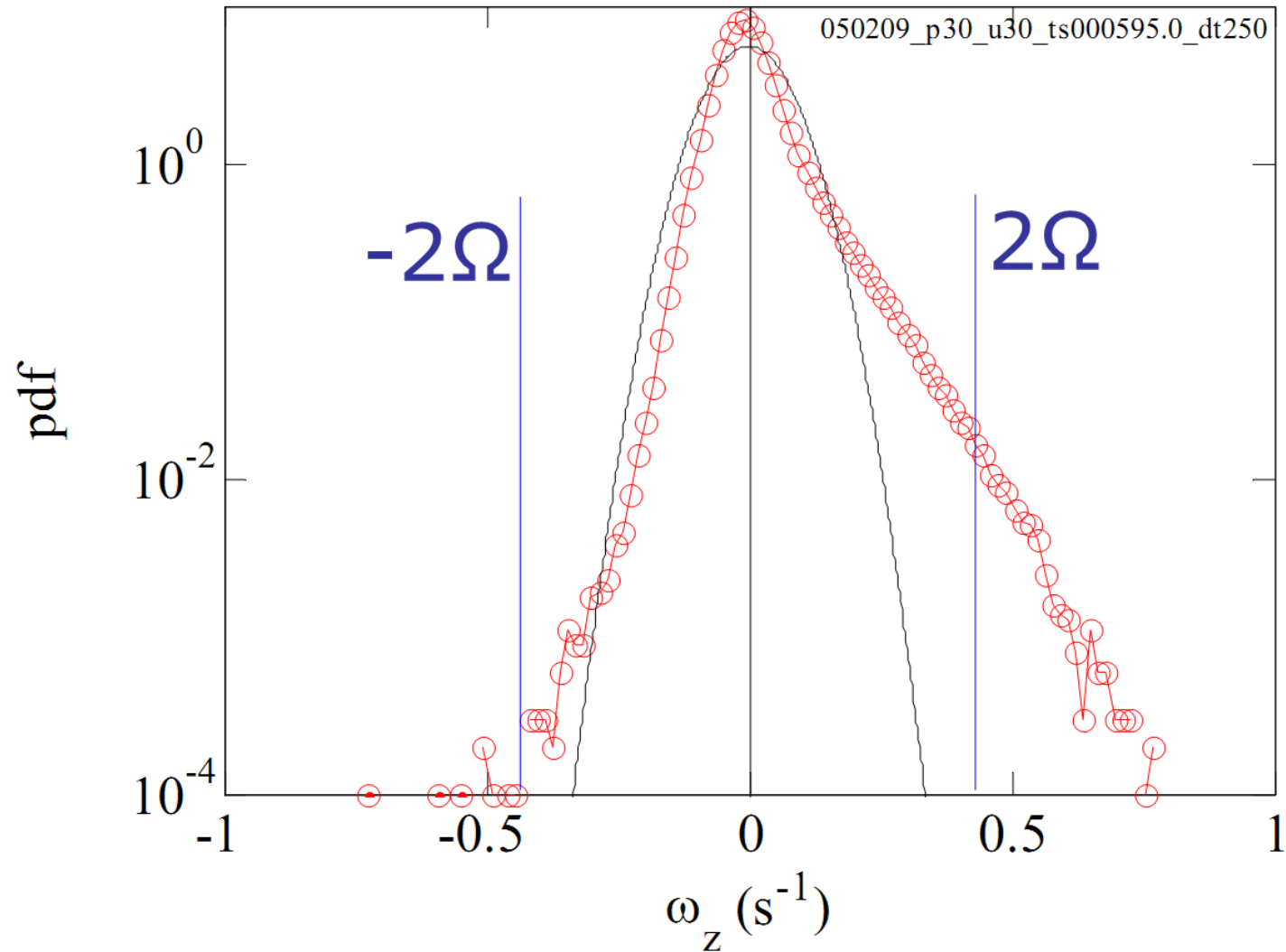


H. Didelle, S. Viboud



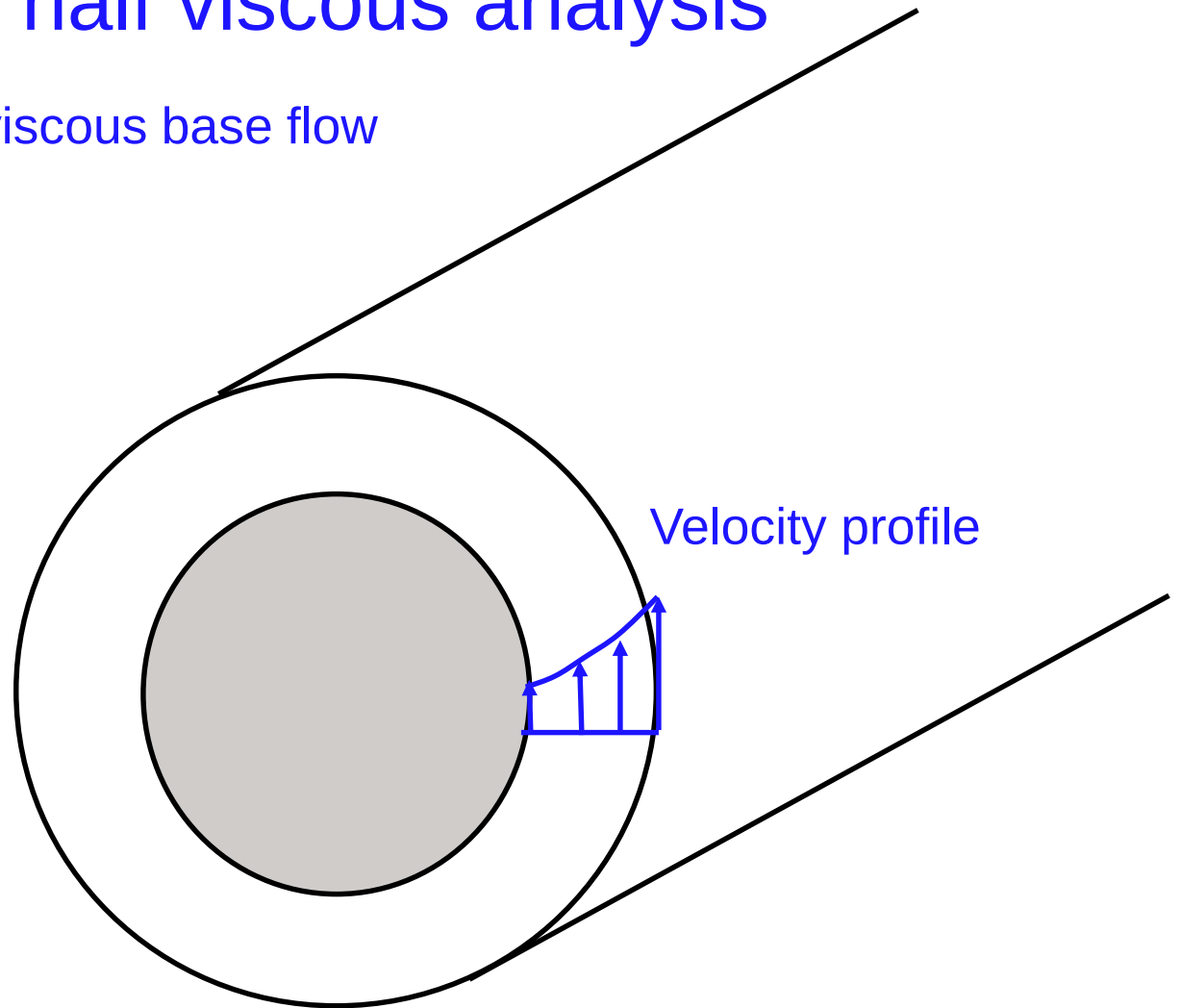
Anticyclones

Cyclones



The half viscous analysis

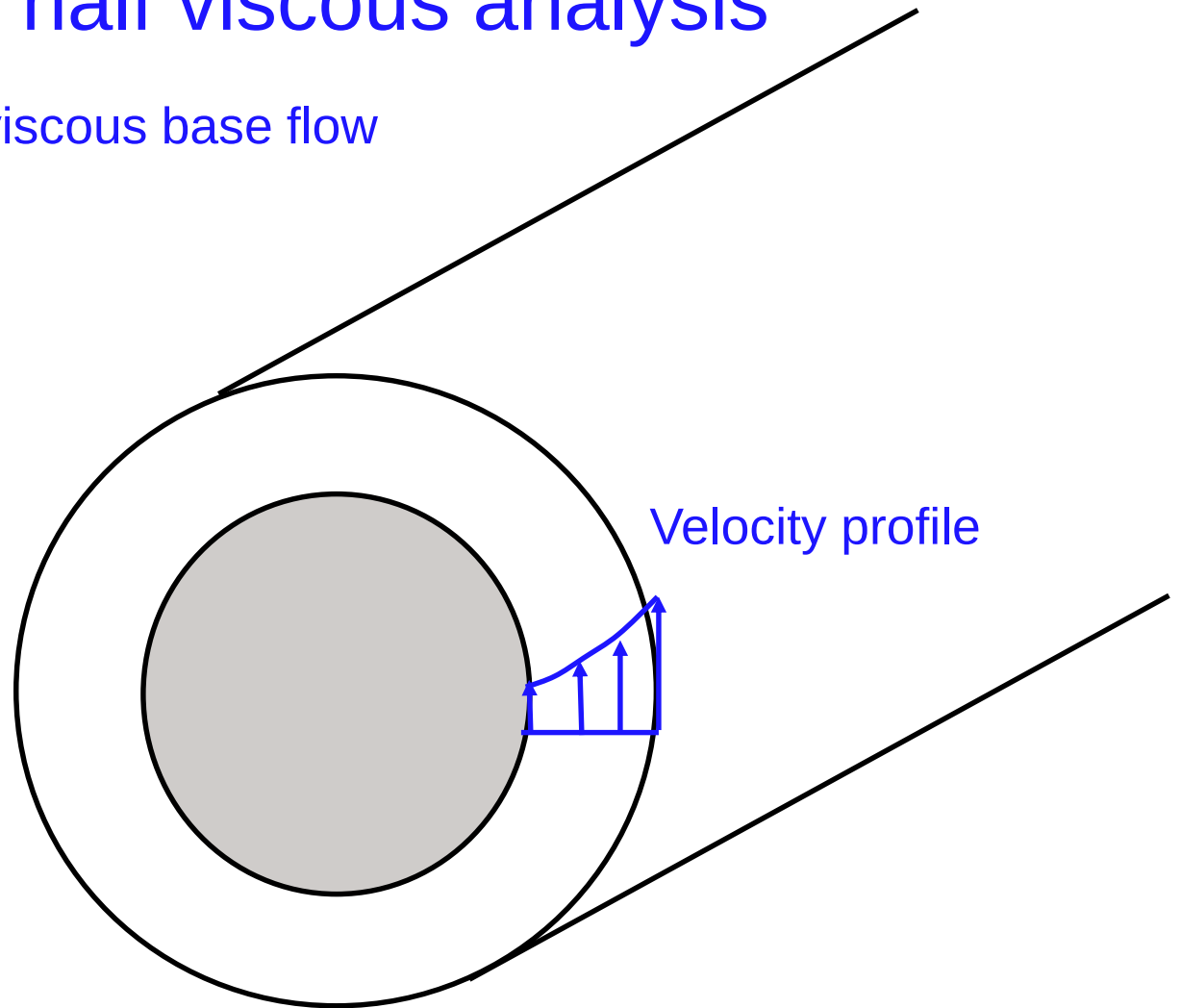
Inviscid analysis of a viscous base flow



The base flow velocity profile is only selected by viscosity!

The half viscous analysis

Inviscid analysis of a viscous base flow



if stable, viscosity is expected to be further stabilizing

if unstable, we expect a critical reynolds number

Base flow

$$\rho \frac{U_\theta^2}{r} = \frac{\partial P}{\partial r}$$
$$\frac{\partial^2 U_\theta}{\partial r^2} + \frac{1}{r} \frac{\partial U_\theta}{\partial r} - \frac{U_\theta}{r^2} = 0$$

Boundary conditions

$$U_\theta(R_1) = R_1 \Omega_1$$

$$U_\theta(R_2) = R_2 \Omega_2$$

The half viscous analysis

Inviscid analysis of a viscous base flow?

viscous time?

convective time?

The half viscous analysis

Inviscid analysis of a viscous base flow

$$\tau_\nu = (R_2 - R_1)^2 / \nu$$

viscous time

$$\tau_{U_2} = (R_2 - R_1) / (\Omega_2 R_2)$$

convective time

$$\tau_\nu / \tau_{U_2} = (R_2 - R_1) \Omega_2 R_2 / \nu = \text{Re}_2$$

Reynolds number

General solution

$$U_{\theta}(r) = Ar + B/r$$

Boundary conditions

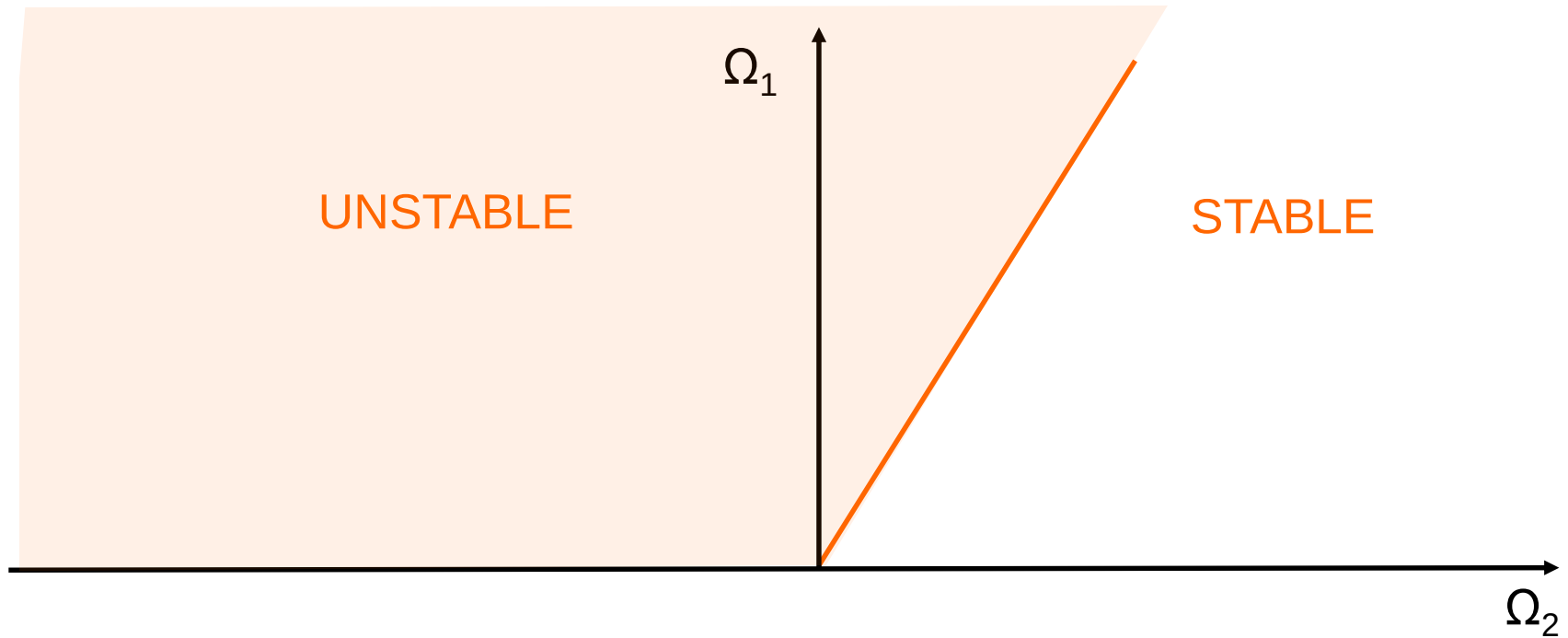
$$A = \frac{\Omega_2 R_2^2 - \Omega_1 R_1^2}{R_2^2 - R_1^2} \quad ; \quad B = \frac{(\Omega_1 - \Omega_2) R_1^2 R_2^2}{R_2^2 - R_1^2}$$

Rayleigh criterion

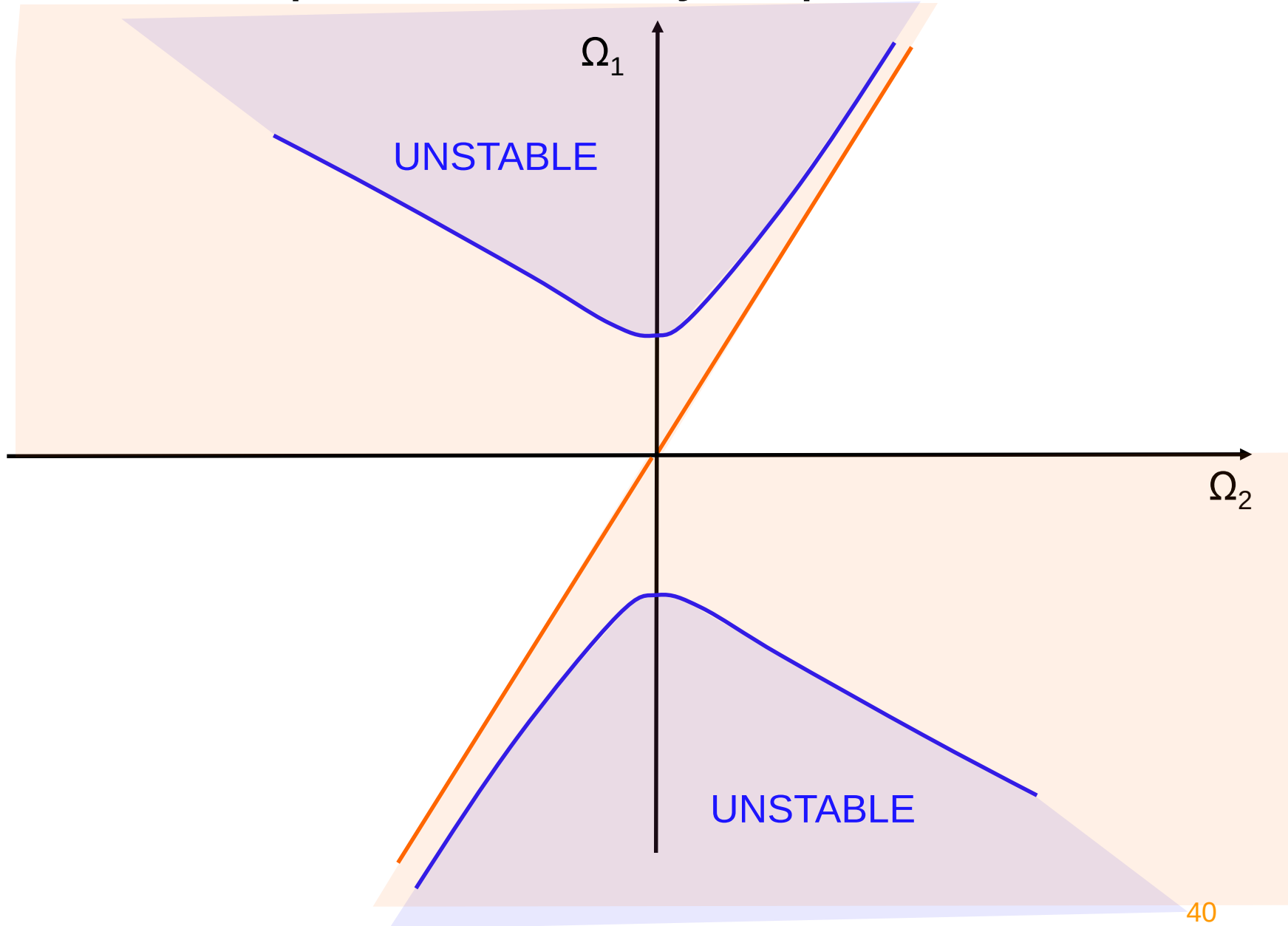
$$d(rU_\theta)^2/dr = 4Ar(Ar^2 + B) < 0$$

Unstable if $\Omega_2 r_2^2 < \Omega_1 r_1^2$.

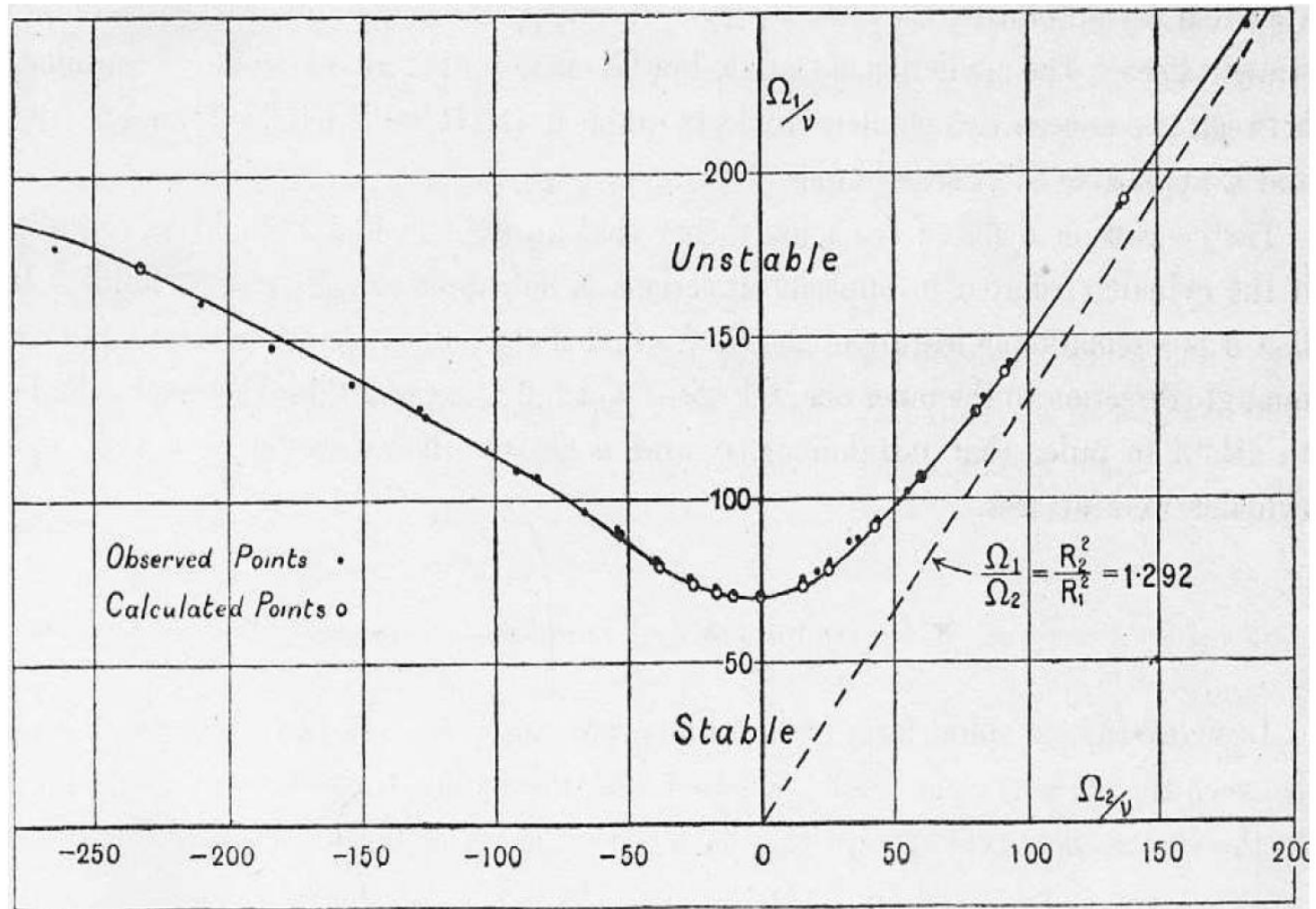
Comparison Theory/Experiment



Comparison Theory/Experiment

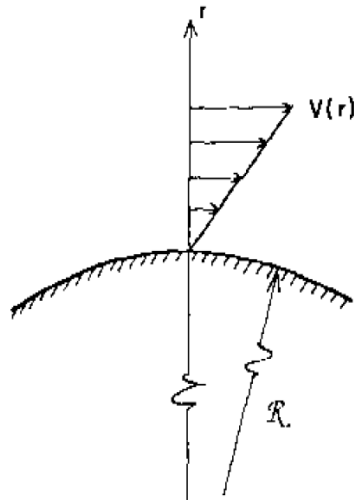


Comparison Theory/Experiment

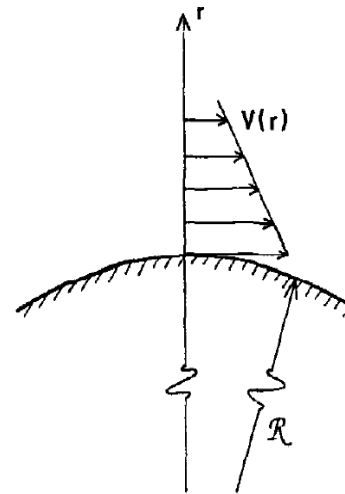


Comparison between observed and calculated speeds at which instability first appears in the case when $R_1 = 3.55$ cm, $R_2 = 4.035$ cm (from Taylor, 1923).

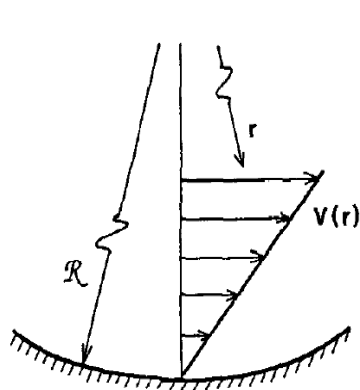
Wall bounded flows



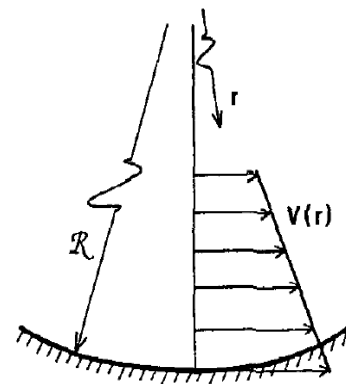
(a) stable



(b) unstable

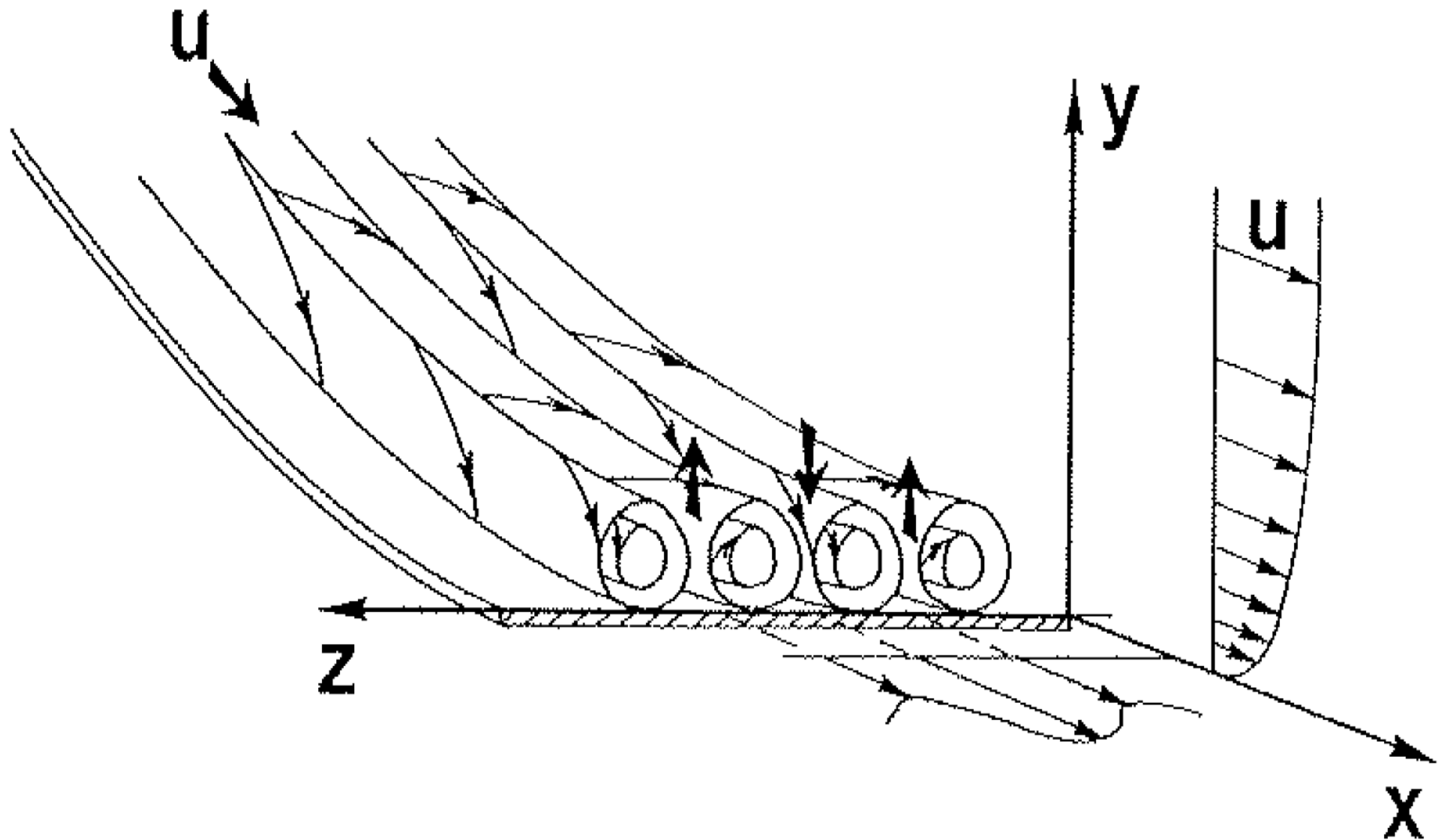


(c) unstable

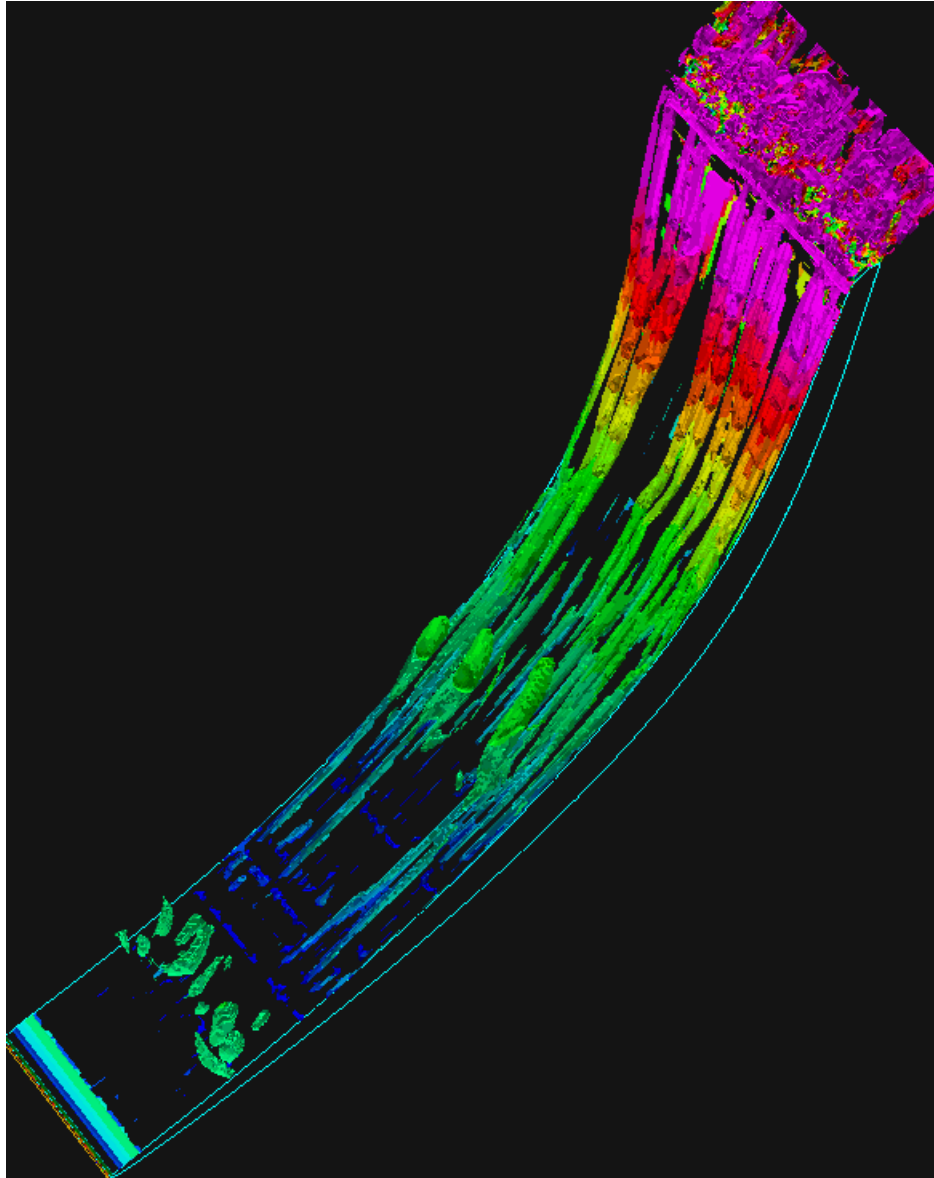


(d) stable

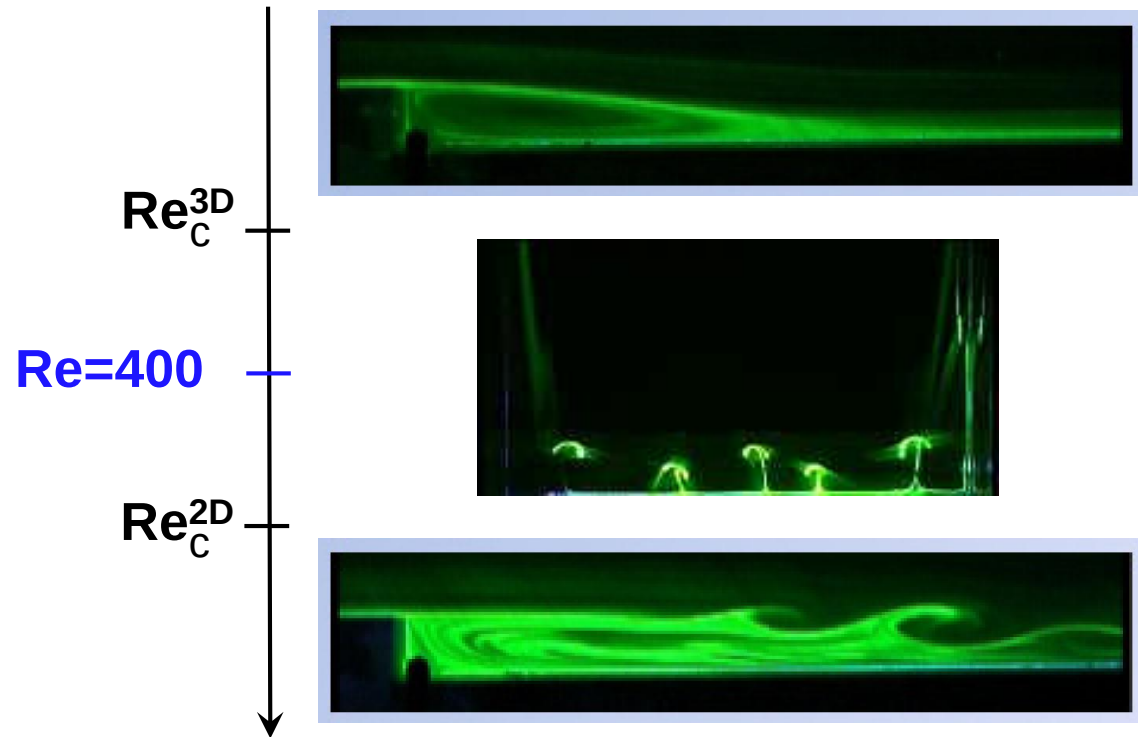
Gortler instability



Gortler instability



Backward facing step

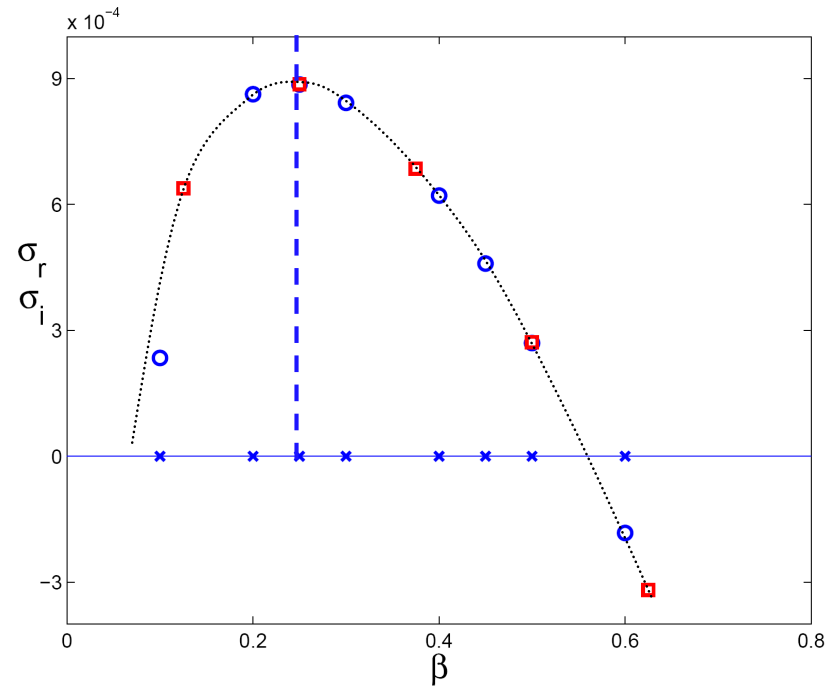


Beaudoin et al.
(Eur. J. Mech. 2004).

DNS by Kaiktsis et al. (JFM 93,96) and linear analysis by Barkley et al. (JFM 2002),

Global dispersion relation

A single stationary unstable global mode



$\beta=0.25$

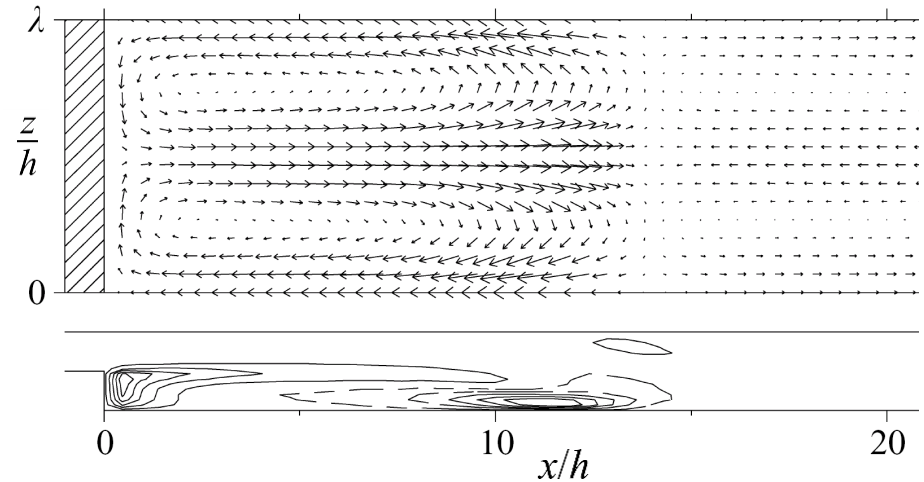
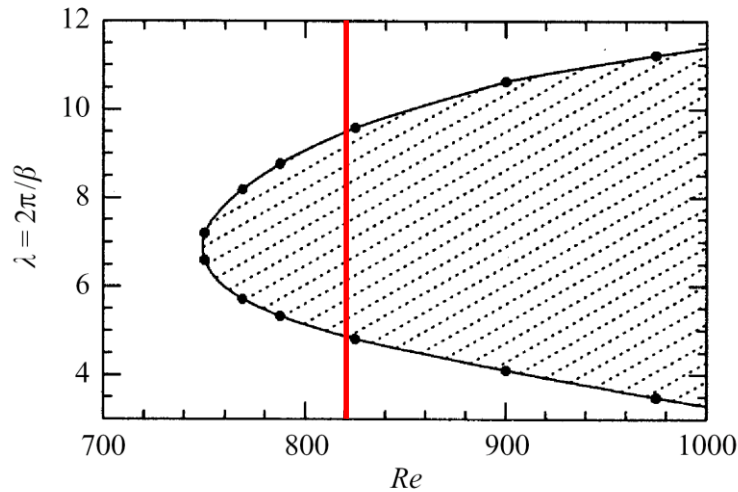


Confirms Barkley et al. (JFM 2002) on backward facing step

Global dispersion relation

A single stationary unstable global mode

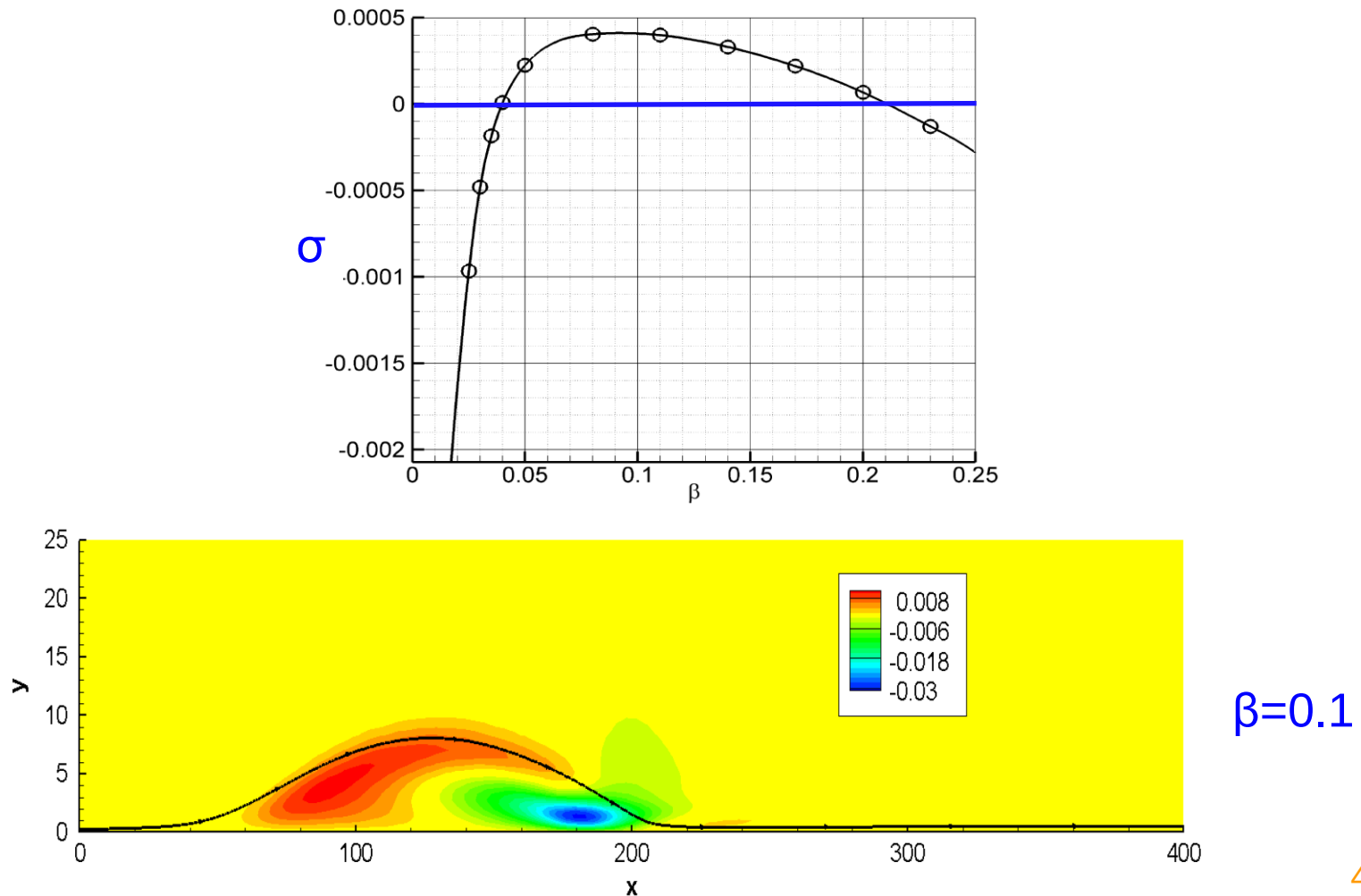
Confirms Barkley et al. (JFM 2002) on backward facing step



Global dispersion relation

A single stationary unstable global mode

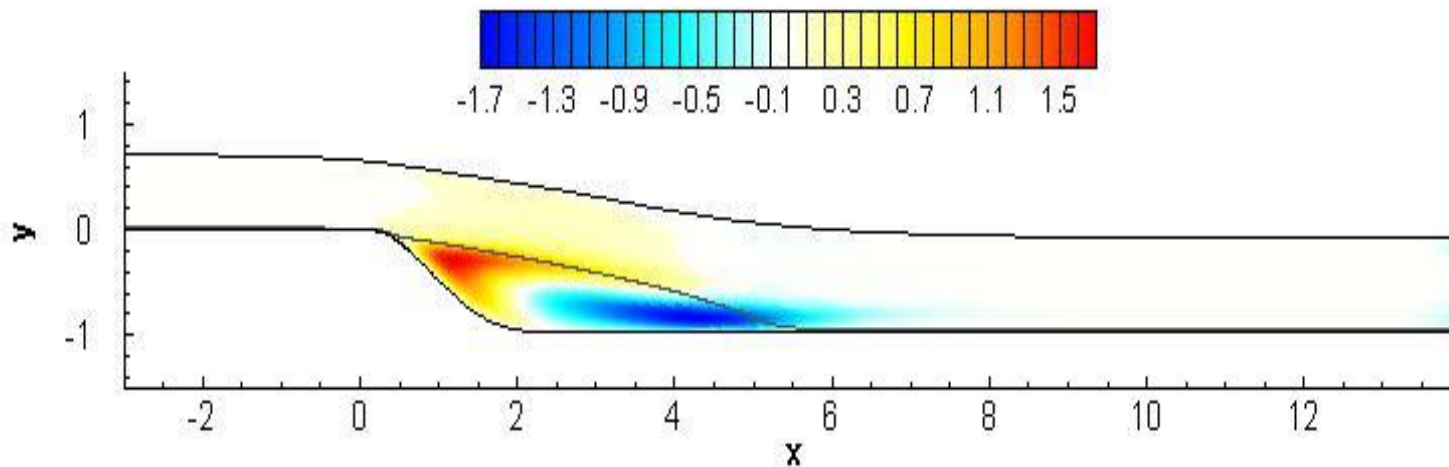
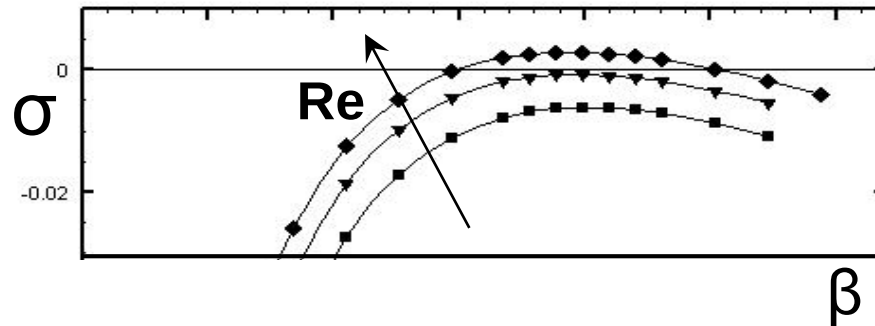
pressure induced separation (Alizard & Robinet 2007)



Global dispersion relation

A single stationary unstable global mode

rounded facing step (Marquet, Chomaz, Sipp & Jacquin 2007)



$\beta=2$