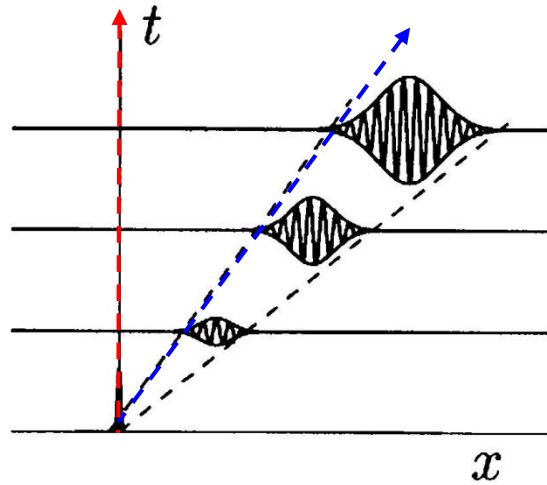


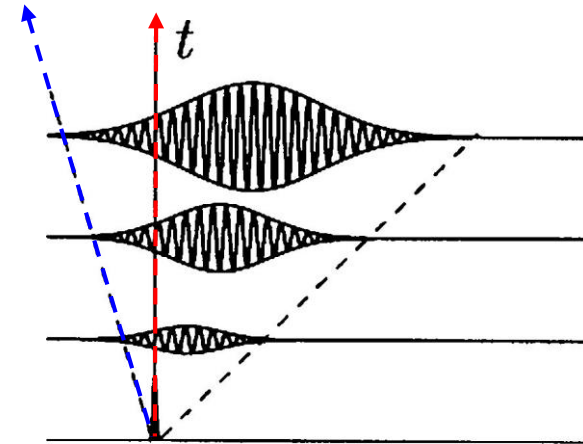
Most flows are unstable...

1. Intro-definitions
2. Rayleigh-Taylor
3. Rayleigh Plateau (destabilization through surface tension)
4. Rayleigh-Benard (convection)
5. Benard-Marangoni
6. Taylor Couette-Centrifugal instability
7. Kelvin-Helmholtz
8. Inflection point theorem Rayleigh
9. Orr sommerfeld, transient growth
10. Spatial growth
11. Spatio-temporal growth
12. Global stability analysis and resolvent

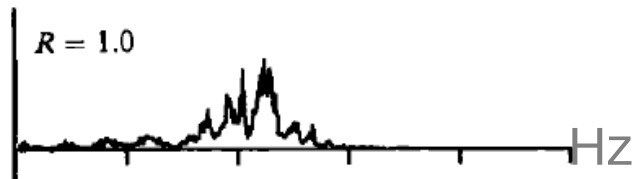
👉 Convective instability



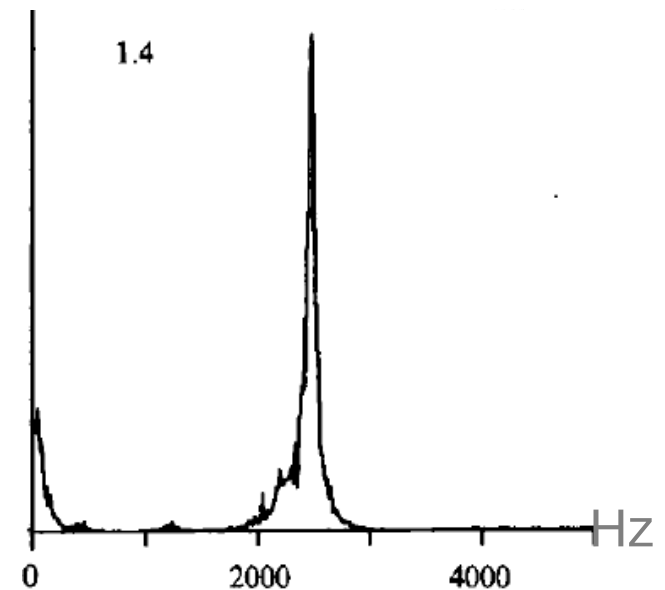
👉 Absolute instability



Spectrum



👉 amplifier

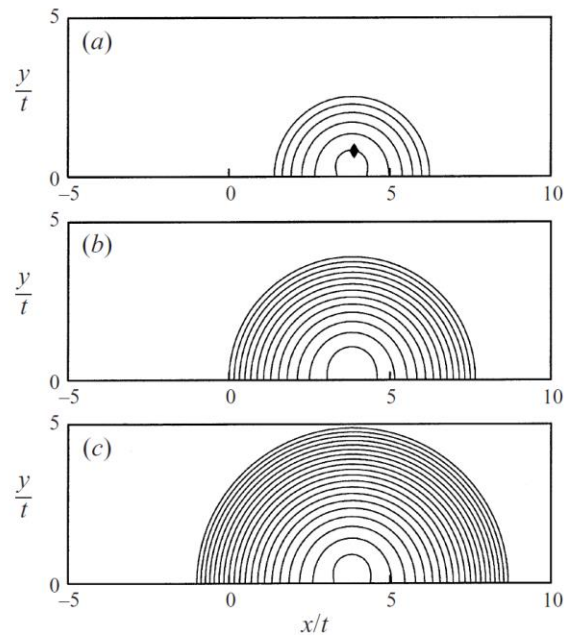


👉 oscillator

Mixing layer experiments by Strikowsky and Niccum (1991)

A/C analysis in Rayleigh-Benard Poiseuille Carrière and Monkewitz (1999)

P. Carrière and P. A. Monkewitz



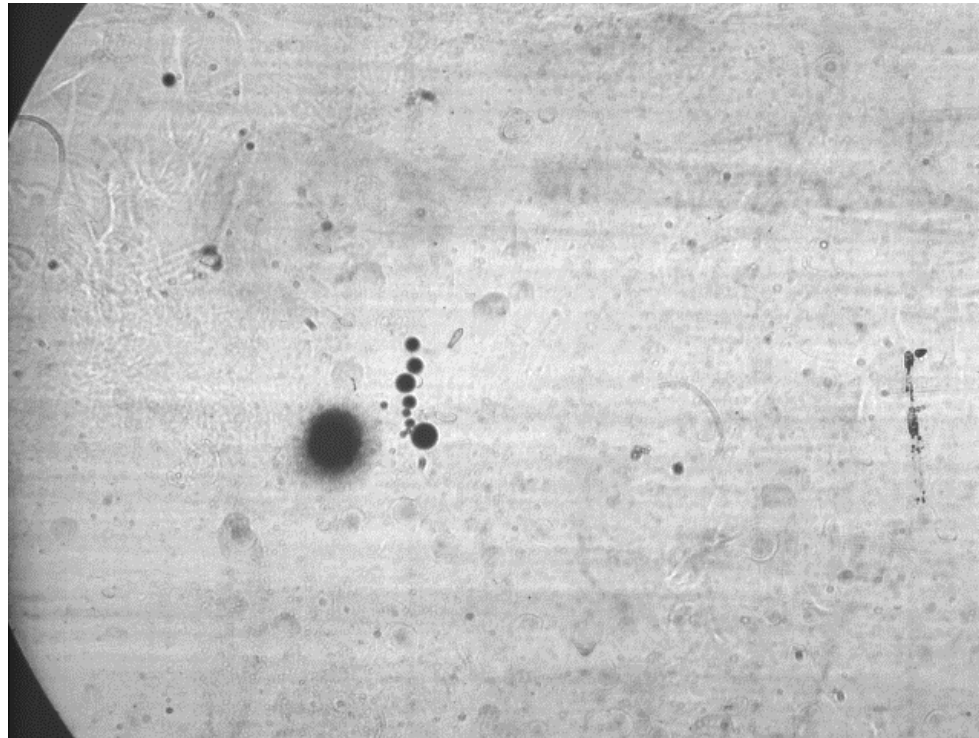
Ra=1760

Ra=1828

Ra=1890

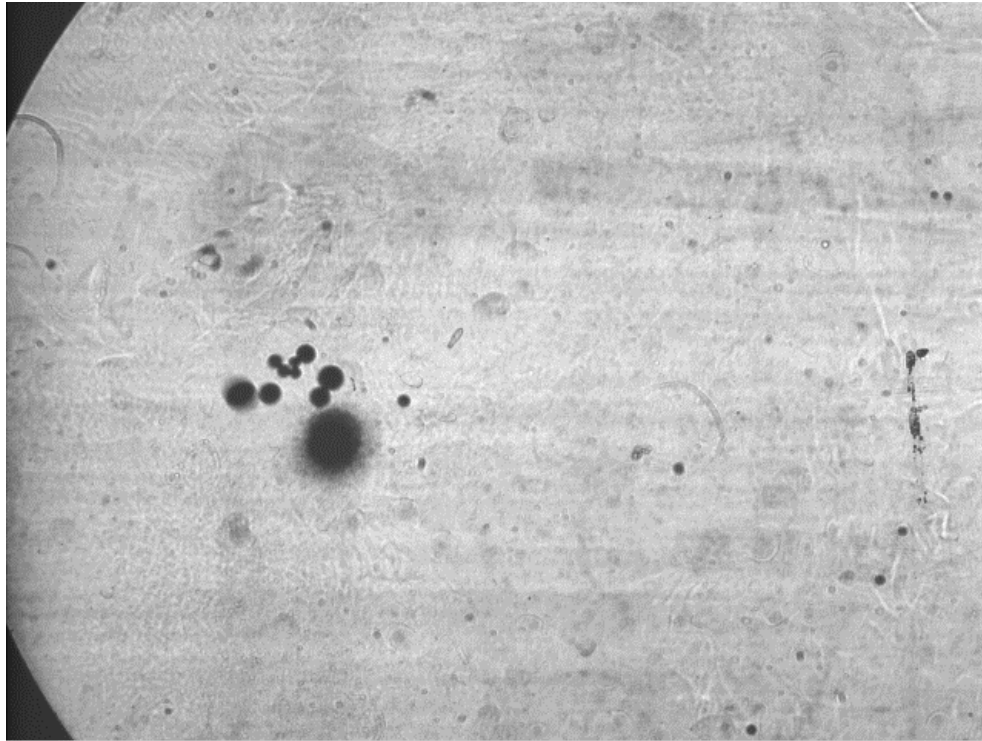
Re=0.63

Impulse in Rayleigh-Benard Poiseuille Grandjean and Monkewitz (2009)



$Ra > 1707$, $Re = 0.03$

Impulse in Rayleigh-Benard Poiseuille Grandjean and Monkewitz (2009)



$Ra > 1707$, $Re = 0.003$

Rayleigh-Plateau

$$\omega = Uk \pm \sqrt{\frac{\gamma k^2}{\rho} \left(k^2 - \frac{1}{R_0^2} \right) \frac{I'_0(kR)}{I_0(kR)}}$$

- **Unstable** if there exists one ω , $\text{Im}(\omega) > 0$ at $k < 1/R_0$
- **Neutral** if for all ω , $\text{Im}(\omega) = 0$ at $k > 1/R_0$
- **Stable (or damped)** if for all ω , $\text{Im}(\omega) < 0$:

The flow considered is not damped, we have neglected dissipation by neglecting viscosity

Surface tension is **destabilizing** as a consequence of the **radial** curvature
Surface tension is **stabilizing** as a consequence of the **axial** curvature

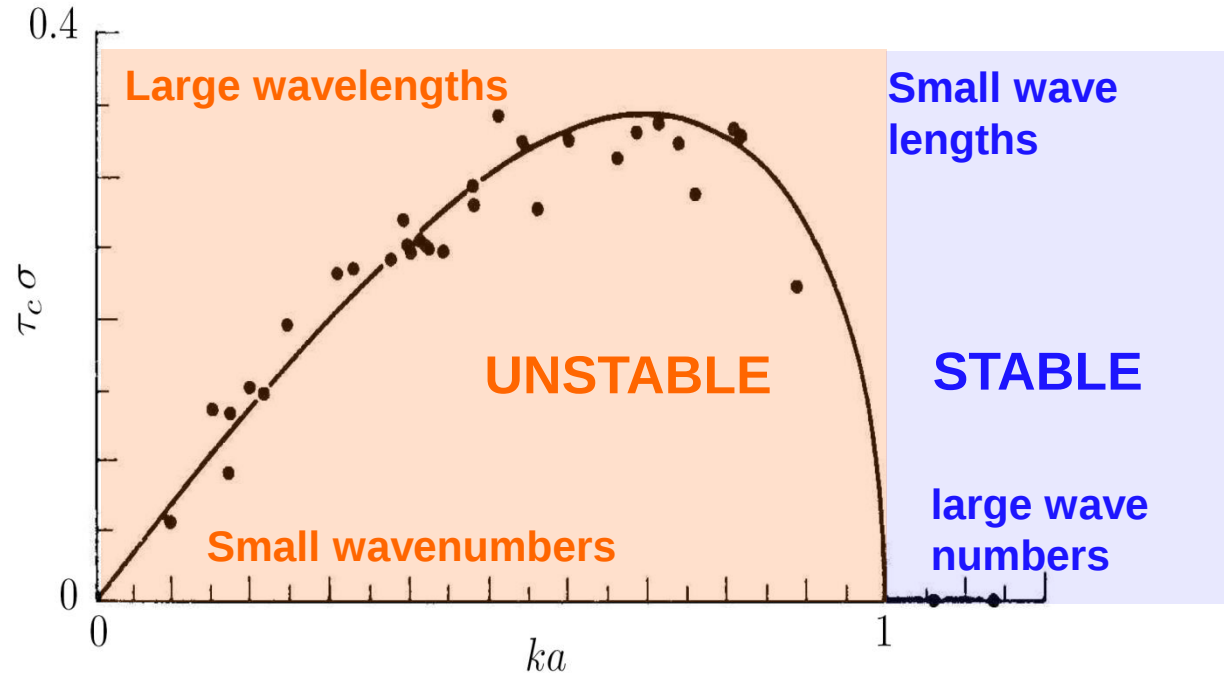


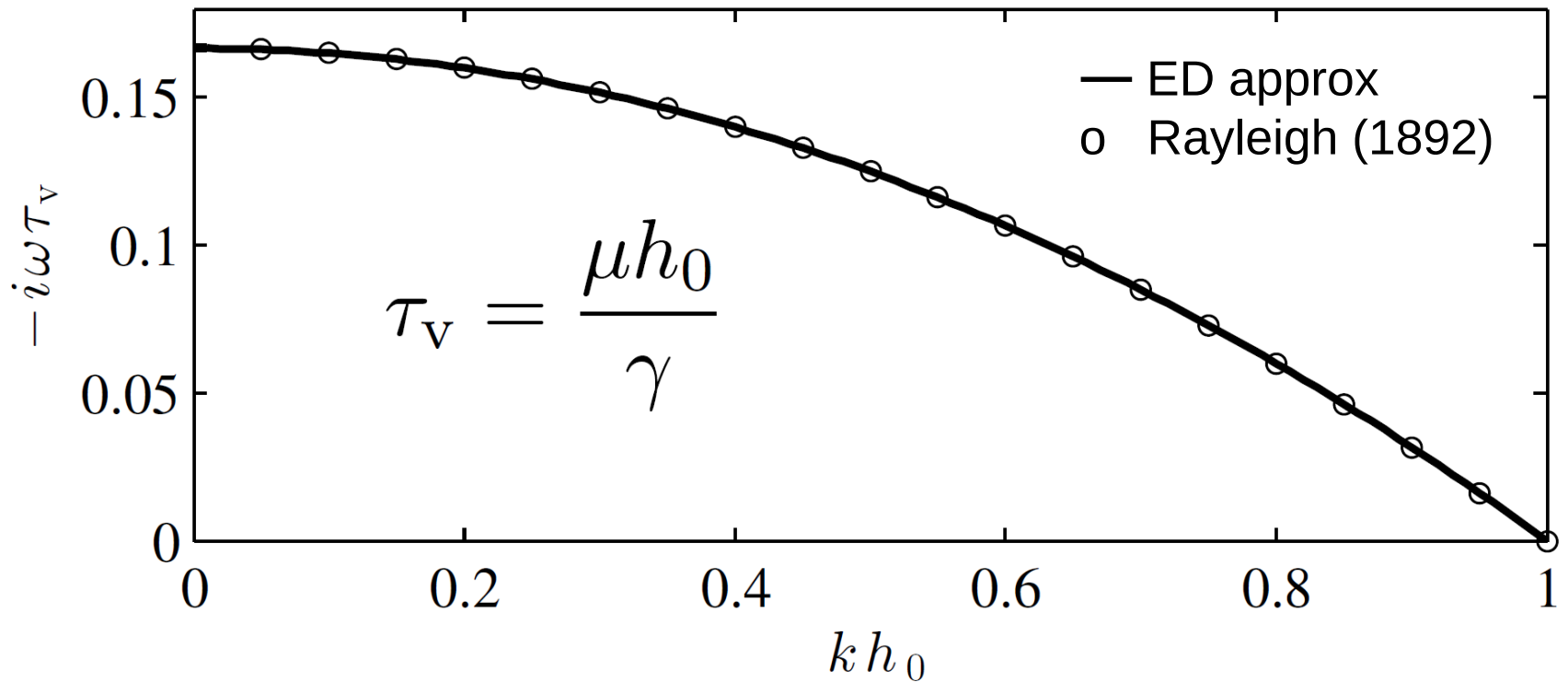
FIG. 2.10 – Taux de croissance $\tau_c \sigma$, avec $\tau_c = \sqrt{\rho a^3 / \gamma}$, de l'instabilité d'un filet fluide non visqueux, et points expérimentaux. D'après (Drazin & Reid 2004).

Oh $\gg$ 1 – Viscosity dominated
A very similar calculation yields
(Rayleigh)

$$\omega = kU_0 + i \frac{\gamma}{2\mu R_0} \frac{(1 - (kR_0)^2)}{(kR_0)^2 (I_0(kR_0)^2 / I_1(kR_0)^2) - (1 + (kR_0)^2)}$$

$$Oh = \frac{\mu}{\sqrt{\rho\gamma R_0}}$$

Eggers and Dupont (1994) equations Oh>>



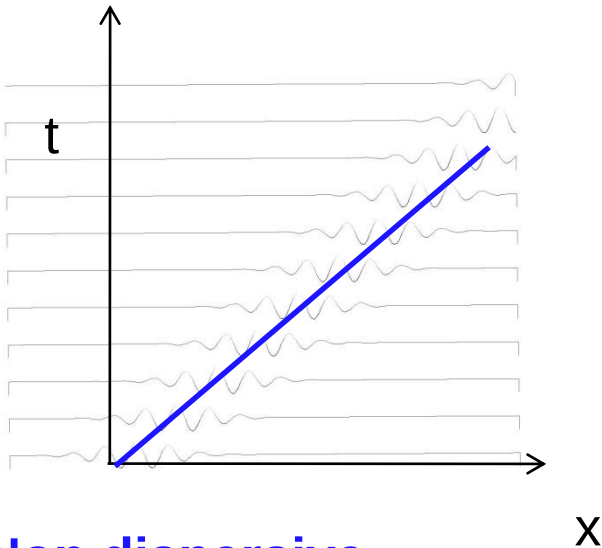
$$\omega = u_0 k + \frac{1}{6\tau_v} i \left(1 - (kh_0)^2 \right)$$

Nondimensionalize

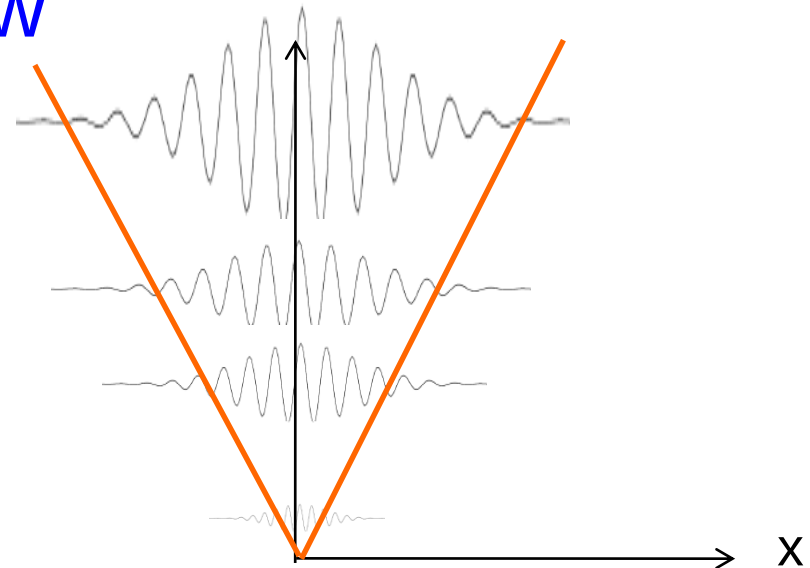
$$\omega = \text{Ca } k + \frac{i}{6}(1 - k^2)$$

$\text{Ca} = \mu u_0 / \gamma$ is the capillary number (a reduced velocity)

The instability waves simultaneously travel and grow



**Non dispersive
propagation at the
velocity of the interface**



**Propagating symmetric
growing wavepacket**

Find k_0

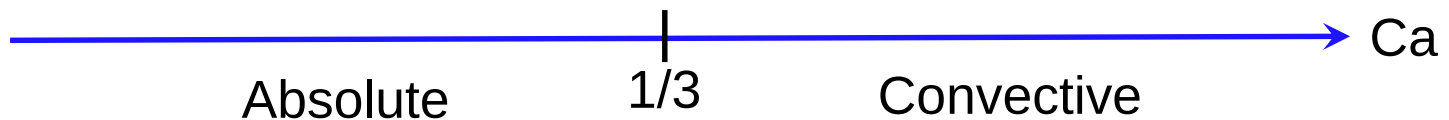
$$\omega = \text{Ca } k + \frac{i}{6}(1 - k^2)$$

$$\frac{d\omega}{dk}(k_0) = 0 \quad \Rightarrow \quad k_0 = -3i\text{Ca}$$

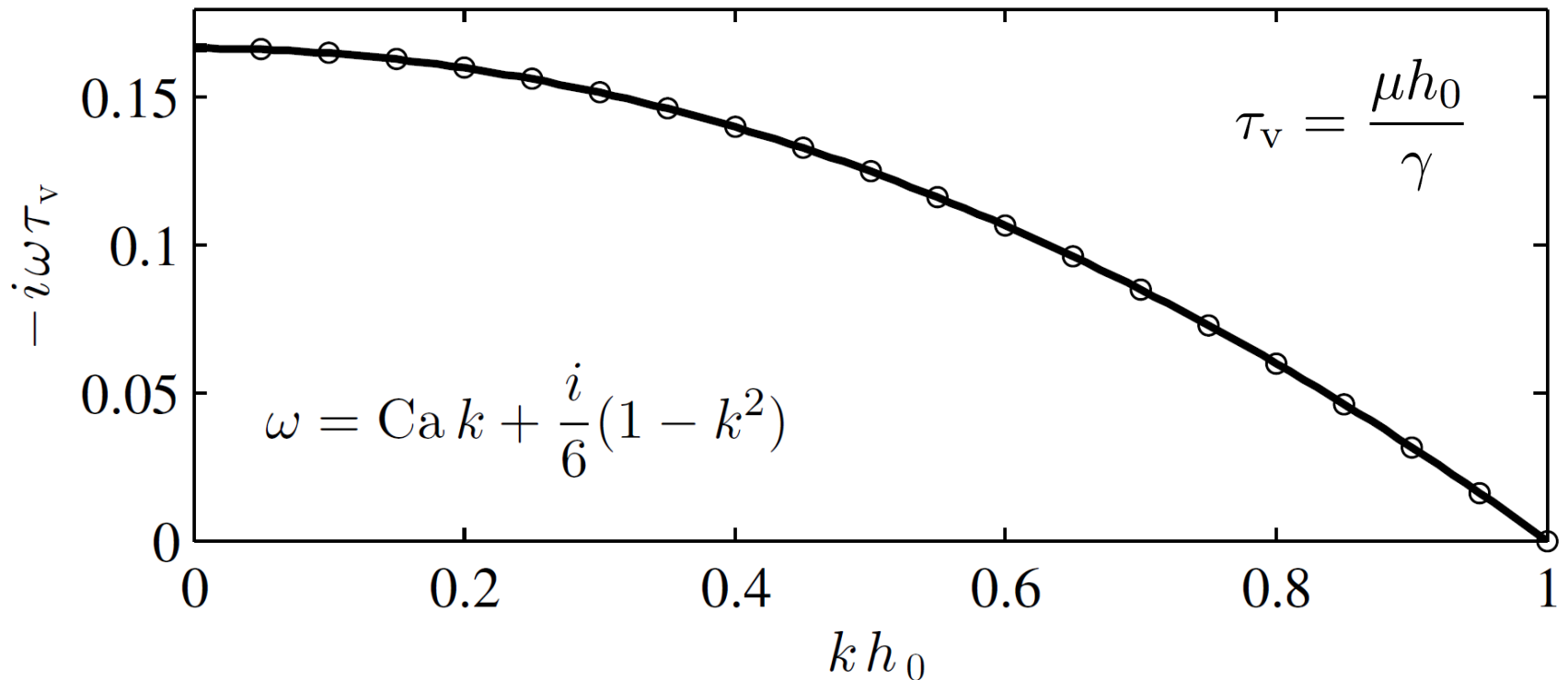
Evaluate ω_0

$$\omega_0 = \omega(k_0) = -\frac{3}{2}i\text{Ca}^2 + \frac{i}{6}$$

$$\text{Im}(\omega_0) = (1 - 9\text{Ca}^2)/6$$



Consider the governing equation of the viscous jet and impose boundary conditions



$$\frac{\partial \eta}{\partial t} = -Ca \frac{\partial \eta}{\partial z} + \frac{1}{6} \eta(z, t) + \frac{1}{6} \frac{\partial^2 \eta}{\partial z^2}$$

$$\eta(0, t) = \eta(l, t) = 0$$

Find eigenmodes $\eta(x, t) = \bar{\eta}(z)e^{\lambda t}$

$$\frac{\partial \eta}{\partial t} = -\text{Ca} \frac{\partial \eta}{\partial z} + \frac{1}{6} \eta(z, t) + \frac{1}{6} \frac{\partial^2 \eta}{\partial z^2}$$

$$\eta(0, t) = \eta(l, t) = 0$$

$$\left(\lambda - \frac{1}{6} \right) \bar{\eta}(z) = -\text{Ca} \frac{d\bar{\eta}}{dz} + \frac{1}{6} \frac{d^2 \bar{\eta}}{dz^2}$$
$$\bar{\eta}(0) = \bar{\eta}(l) = 0$$

Fundamental solutions $\bar{\eta}(z) = \hat{\eta} e^{\alpha z}$

$$\alpha^2 - 6\text{Ca}\alpha + 1 - 6\lambda = 0$$

$$\alpha_{1,2} = 3\text{Ca} \pm \frac{1}{2} \sqrt{36\text{Ca}^2 + 24\lambda - 4}$$

Fundamental solutions $\bar{\eta}(z) = \hat{\eta}e^{\alpha z}$

$$\alpha^2 - 6\text{Ca}\alpha + 1 - 6\lambda = 0$$

$$\alpha_{1,2} = 3\text{Ca} \pm \frac{1}{2}\sqrt{36\text{Ca}^2 + 24\lambda - 4}$$

Impose boundary conditions

$$\begin{bmatrix} 1 & 1 \\ e^{\alpha_1 l} & e^{\alpha_2 l} \end{bmatrix} \begin{bmatrix} \hat{h}_1 \\ \hat{h}_2 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

$$\Rightarrow e^{\alpha_2 l} - e^{\alpha_1 l} = 0$$

$$\Rightarrow \alpha_1 - \alpha_2 = i\frac{2\pi m}{l} \quad (\text{m integer})$$

$$\sqrt{36\text{Ca}^2 + 24\lambda_m - 4} = i\frac{2\pi m}{l}$$

Find eigenmodes $\eta(x, t) = \bar{\eta}(z)e^{\lambda t}$

$$\lambda_m = -\frac{3}{2}\text{Ca}^2 + \frac{1}{6} - \frac{m^2\pi^2}{6l^2} = \underbrace{-i\omega_0}_{>0 \text{ if } \text{Im}(\omega_0) > 0} - \frac{m^2\pi^2}{6l^2}$$

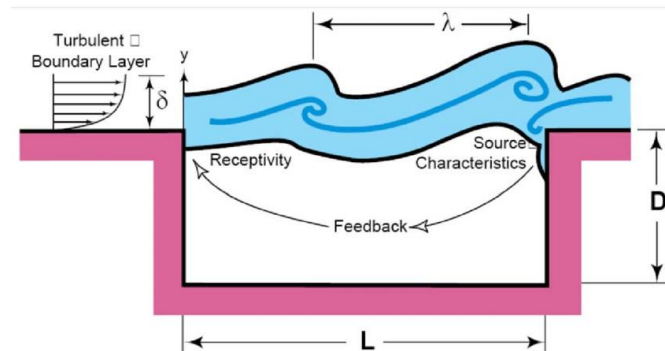
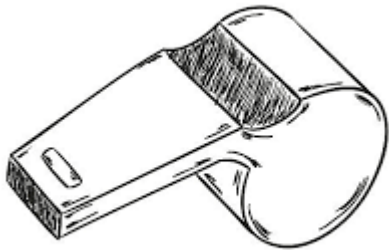
>0 if $\text{Im}(\omega_0) > 0$

absolute frequency $\omega(k_0)$

The flow including boundary conditions is
globally unstable only if it is locally
absolutely unstable

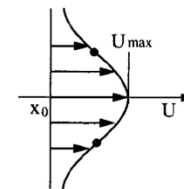
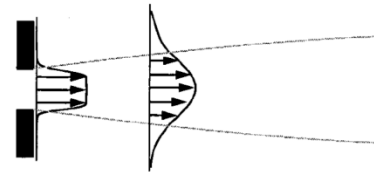
Global instability is needed when the streamwise invariance is broken

-by inlet/outlet boundary conditions



-by the nonparallelism of the flow

2D jet



Global instability

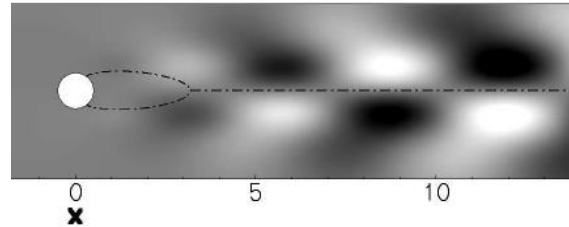
The normal mode expansion $u(y)\exp(i(kx-\omega t))$
is replaced by
a global mode expansion $u(x,y)\exp(\sigma t)$

The dispersion relation $\omega(k)$
is replaced by
a set of global eigenvalues σ

Most often, a necessary condition for global inst. $\sigma_r > 0$
is not temporal instability $\exists k \ \omega_i(k) > 0$

But rather that of absolute instability $\omega_{0i}(k_0) > 0$

Global linear instability of flows



*Stability of viscous flow past
a circular cylinder*
A. Zebib



1971

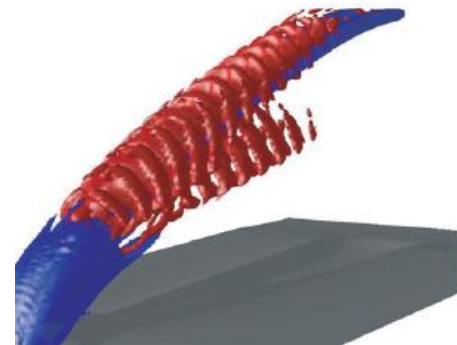
2 D

2009

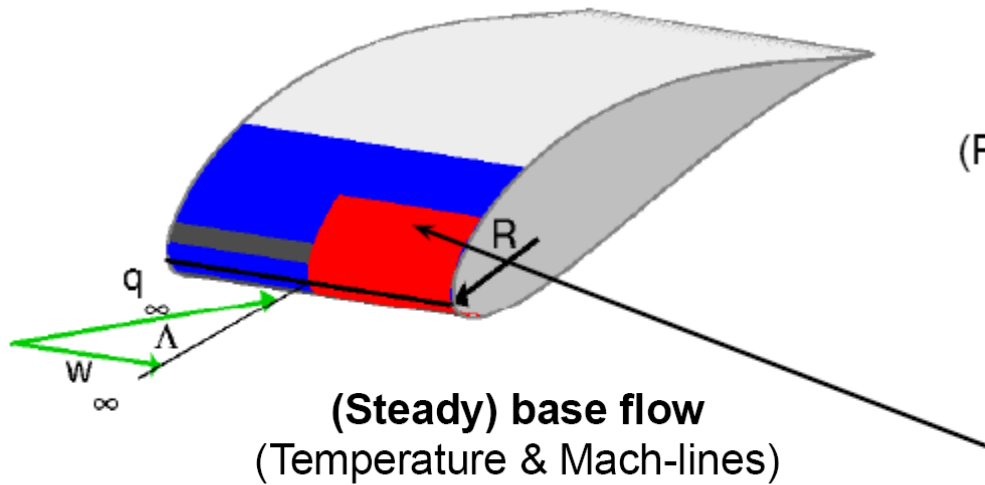
3 D

*Accurate solution of the Orr–
Sommerfeld stability equation*
S. A. Orszag

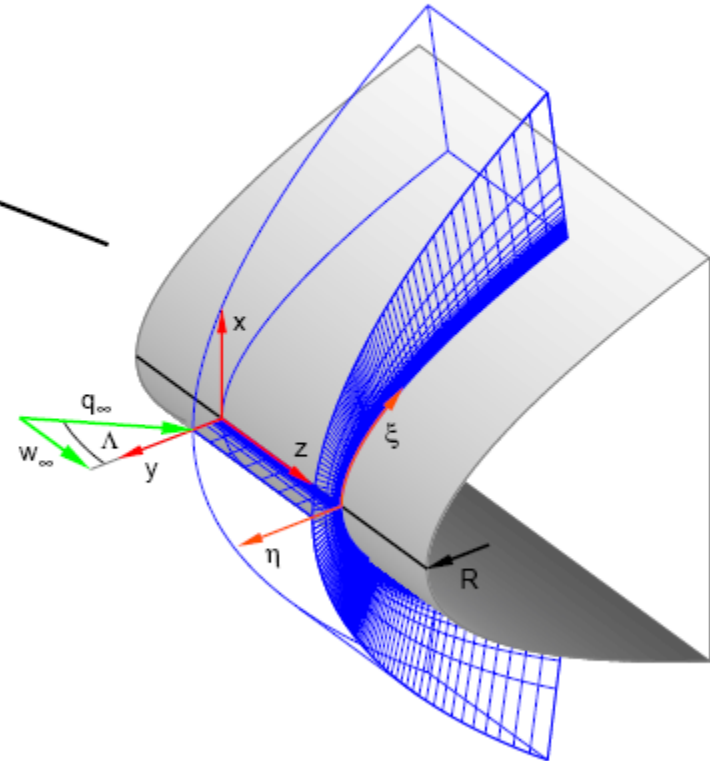
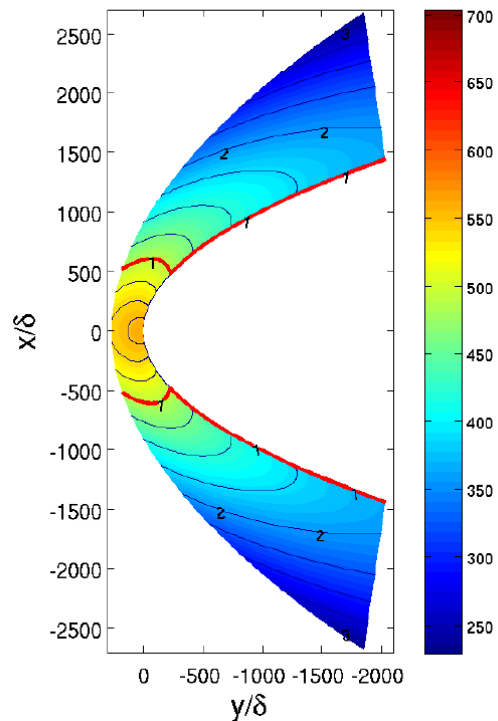
*Global stability of a jet
in crossflow*
S. Bagheri



Global linear instability of flows in complex geometry

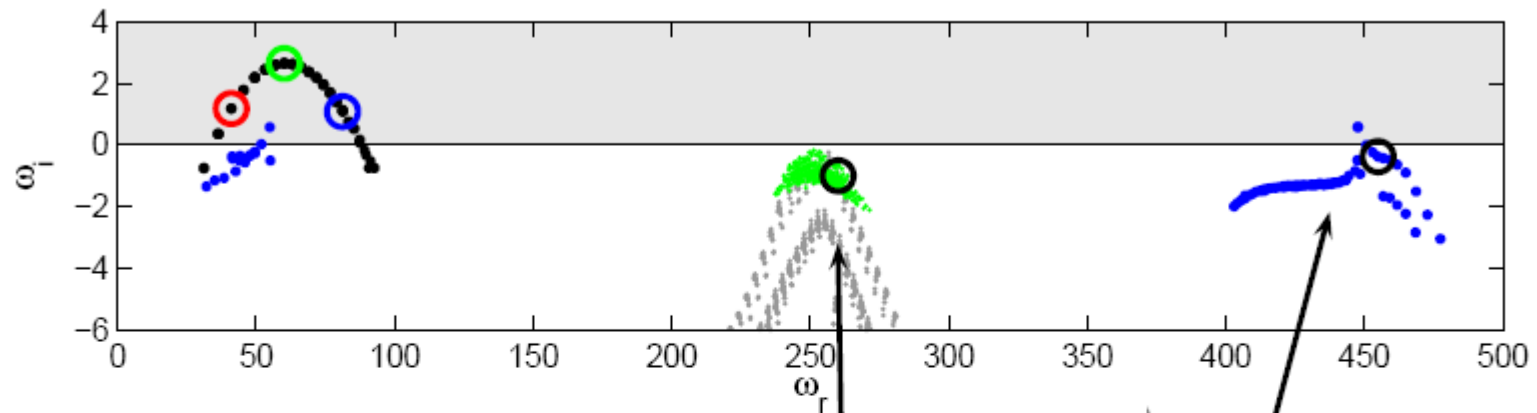


Leading-edge region
(Parabolic body of infinite span)

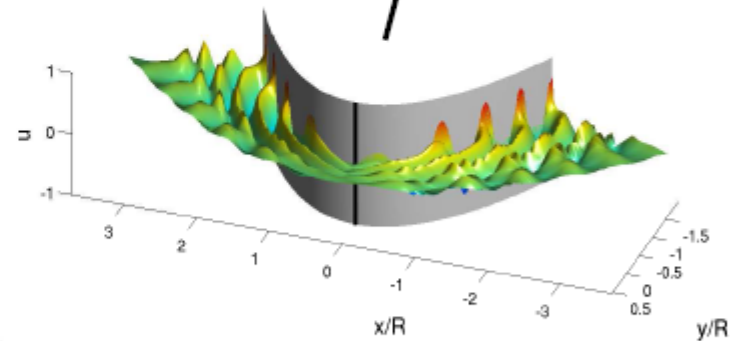
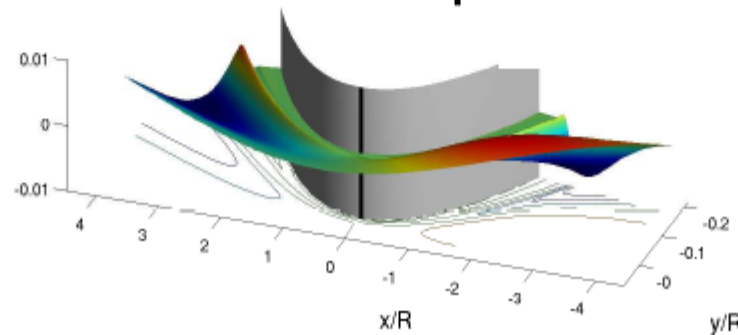
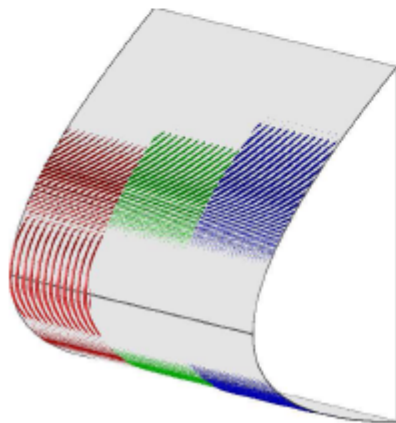


Mack and Schmid 2010

Global spectrum: $\phi'(x, y, z, t) = \tilde{\phi}(x, y)e^{i(\beta z - \omega t)}$ with $\omega = \omega_r + i\omega_i$

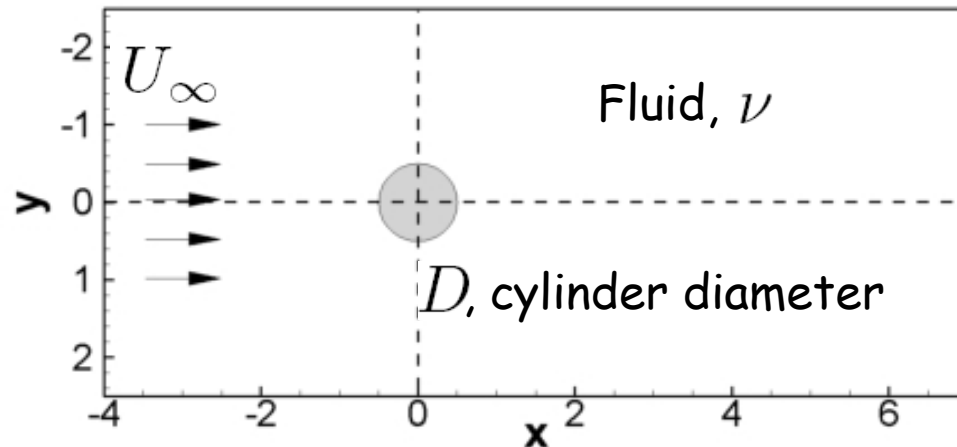


Associated global modes $\tilde{v}(x, y, z)$



Mack and Schmid 2010

Wake behind cylinder flow



$$Re = \frac{U_\infty D}{\nu}$$

Incompressible 2D flow described by

Velocity (u, p) Pressure

Navier-Stokes equations

$$\partial_t u + \nabla u \cdot u = -\nabla p + Re^{-1} \nabla^2 u,$$

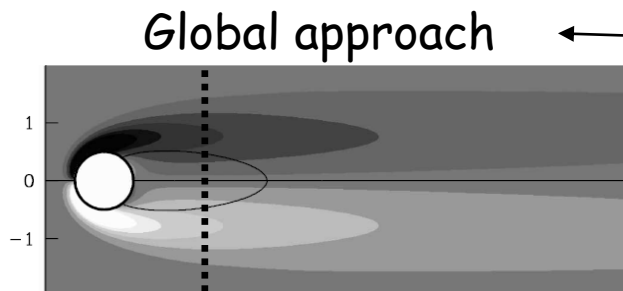
$$\nabla \cdot u = 0$$

Stability analysis

Perturbation expansion

$$(u, p) = (U, P) + (u', p') \quad \text{Perturbations}$$

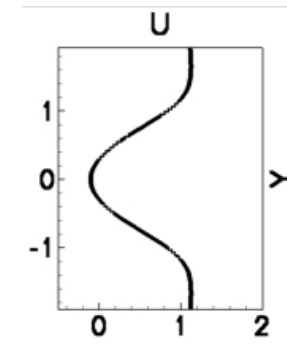
Stationary base flow



$$\leftarrow U(x, y) \quad U(y) \longrightarrow$$

Local approach

Huerre & Monkewitz (1990)



Base flow equations

$$\begin{aligned} \nabla U \cdot U &= -\nabla P + Re^{-1} \nabla^2 U, \\ \nabla \cdot U &= 0 \end{aligned}$$

Zebib, Jackson (1987)

Linearized perturbation equations

$$\begin{aligned} \partial_t u' + \nabla U \cdot u' + \nabla u' \cdot U + \cancel{\nabla u' \cdot u'} &= -\nabla p' + Re^{-1} \nabla^2 u', \\ \nabla \cdot u' &= 0 \end{aligned}$$

Analyse de stabilité globale

$$\begin{array}{c}
 U(x, y) \\
 \downarrow \\
 (\mathbf{u}', p')(x, y, t) = (\hat{\mathbf{u}}, \hat{p})(x, y) \exp[\sigma t]
 \end{array}$$

Global mode(complex) $\xrightarrow{\quad}$ $\hat{\mathbf{u}}$ $\xleftarrow{\text{Growth-rate}}$ $\sigma = \lambda + i\omega$

ω $\xleftarrow{\text{frequency}}$ $St = \frac{\omega}{2\pi}$

Global stability equations

$$\begin{aligned}
 \sigma \hat{\mathbf{u}} + \nabla \hat{\mathbf{u}} \cdot \mathbf{U} + \nabla \mathbf{U} \cdot \hat{\mathbf{u}} &= -\nabla \hat{p} + Re^{-1} \nabla^2 \hat{\mathbf{u}}, \\
 \nabla \cdot \hat{\mathbf{u}} &= 0,
 \end{aligned}$$

Global stability analysis solvers

For a given value of Re , numerically solve

- non linear equations,

$$\nabla U \cdot U = -\nabla P + Re^{-1} \nabla^2 U,$$

(Newton method)

- Eigenvalue problem

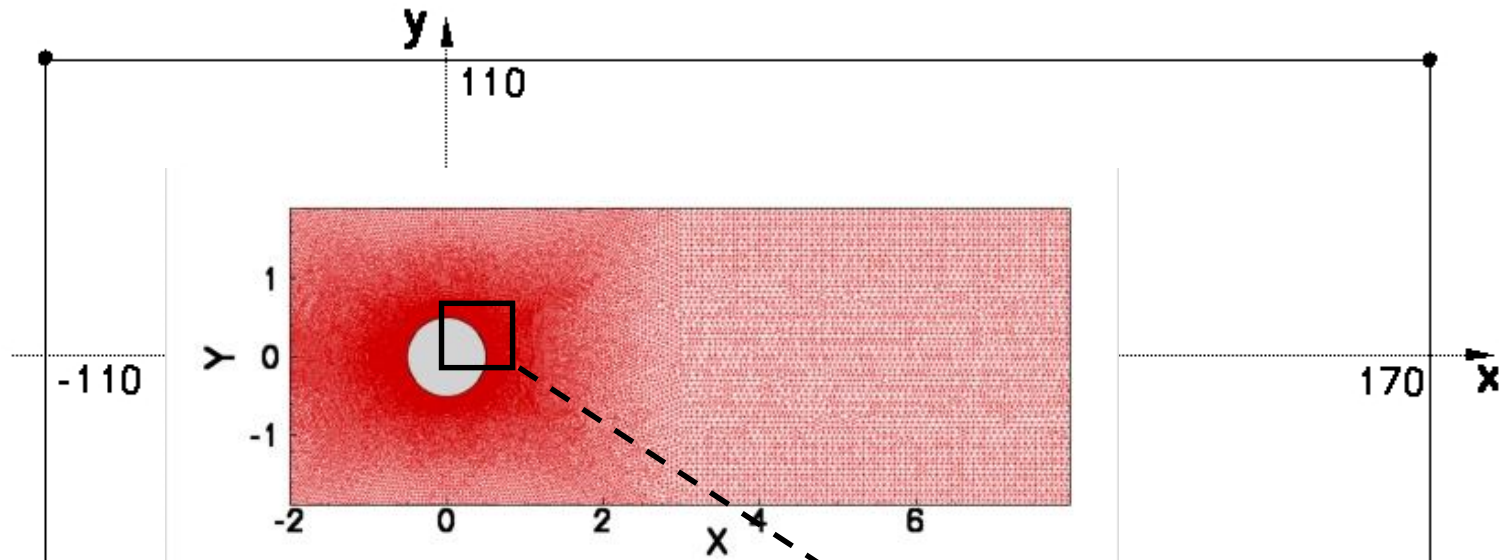
$$\sigma \hat{u} + \nabla \hat{u} \cdot U + \nabla U \cdot \hat{u} = -\nabla \hat{p} + Re^{-1} \nabla^2 \hat{u},$$

(Krylov-Arnoldi method)

Spatial discretization = finite element methods

(FreeFem++ freeware)

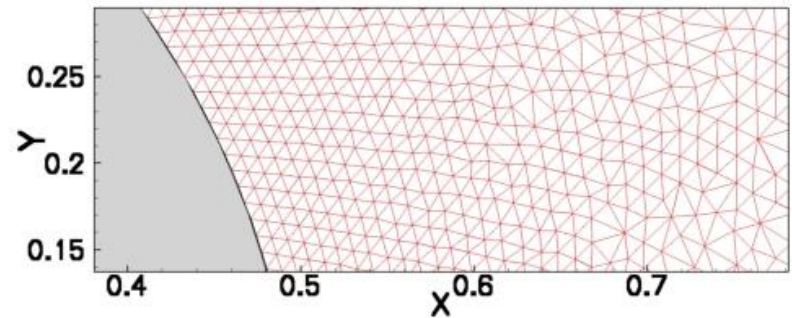
Computational model and mesh



Triangular mesh

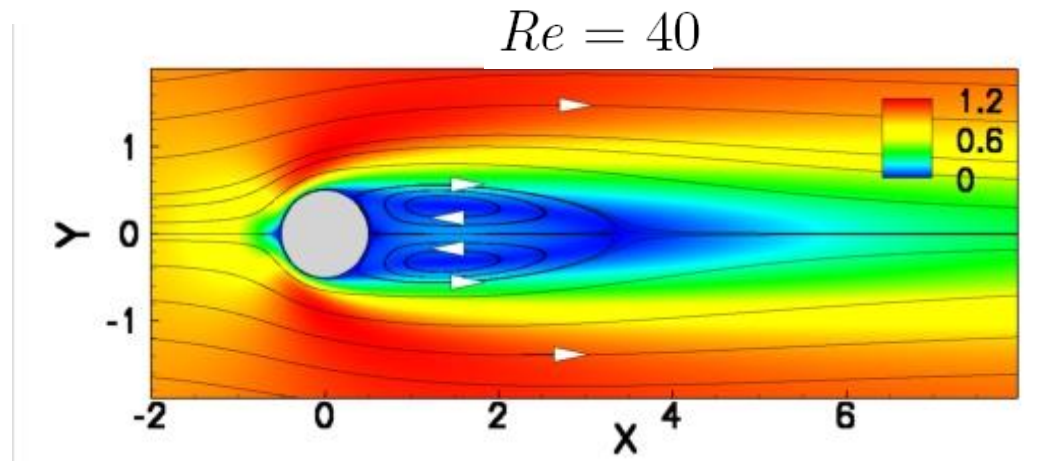
Number of triangles $\sim 375\,000$

Number of nodes $\sim 190\,000$



Taylor-Hood finite elements (P2,P2,P1) $\rightarrow$ number of degrees of freedom $\sim 1.6 \cdot 10^6$

Base flow



Drag coefficient

$$C_D = \frac{F_D}{0.5 D U_\infty^2}$$

C_D	$Re = 20$	$Re = 40$
Dennis & Chang (1970)	2.05	1.52
Fornberg (1980)	2.00	1.50
Ye et al. (1999)	2.03	1.52
Giannetti & Luchini (2007)	2.05	1.54
Marquet et al. (2008)	2.00	1.50

Valeur propre et mode global dominants

$$\Re\{\hat{u}\} \quad (\hat{u}, \hat{p}) \exp[\sigma t] \quad \sigma = \lambda + i\omega$$

Leading eigenvalue:

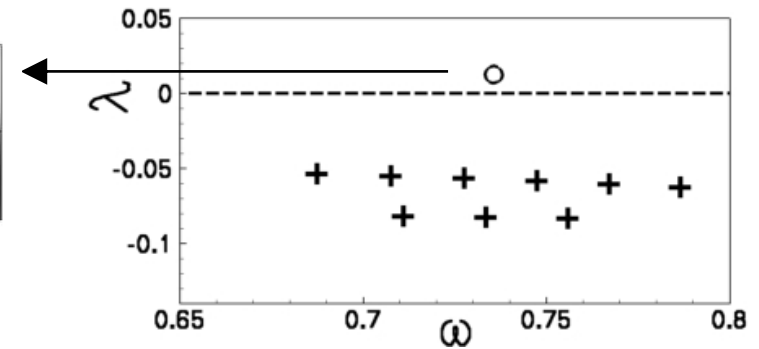
λ_{max}

if $\lambda_{max} < 0 \rightarrow$ Stable flow

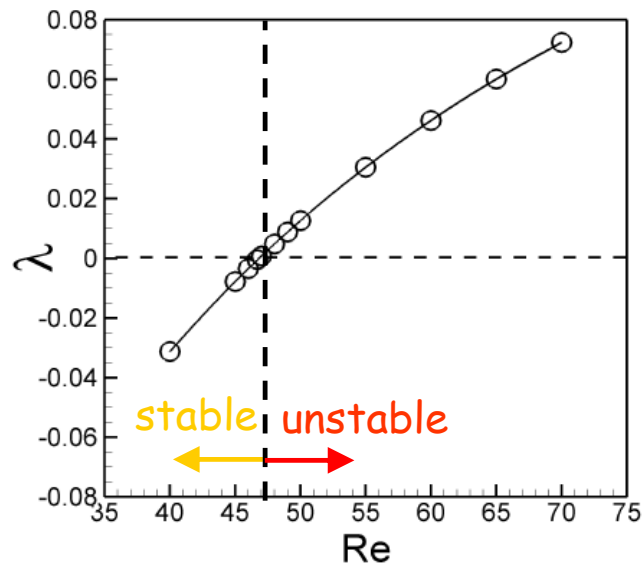
if $\lambda_{max} > 0 \rightarrow$ Unstable flow

Leading global mode

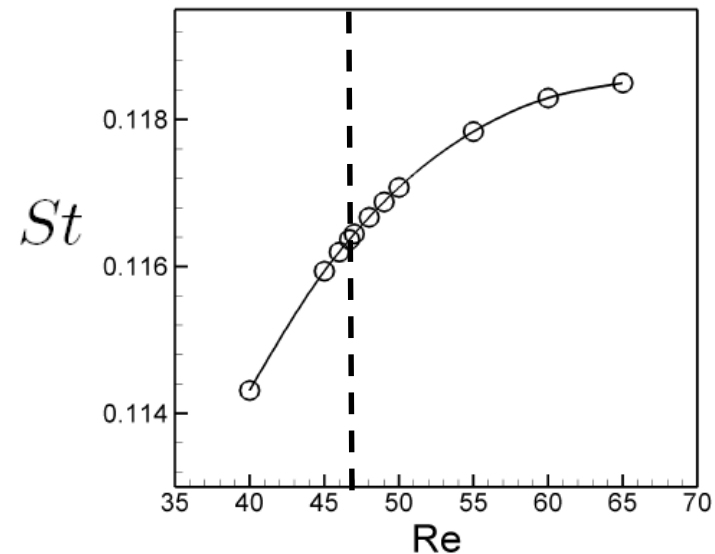
Spectrum at $Re = 50$



Evolution as a function of the Reynolds number



$$Re_c = 46.7(\pm 0.1)$$



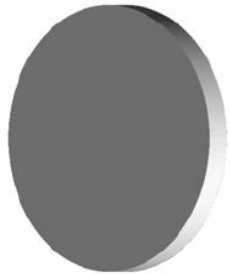
$$St_c = 0.116$$

To keep in mind

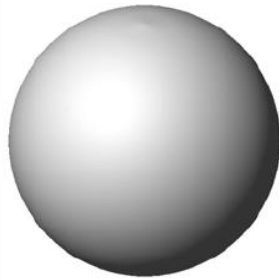
Global stability analysis=
Nonlinear problem (base flow) +
Eigenvalue problem

Prototype flows

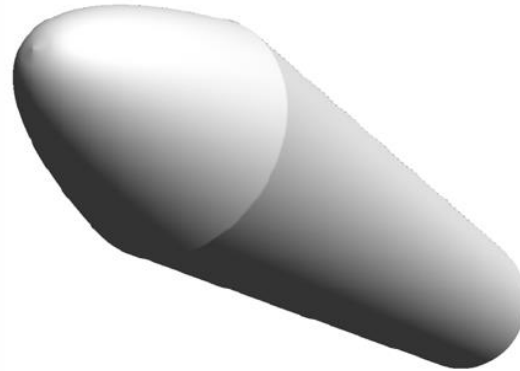
Axisymmetric wakes



Disk



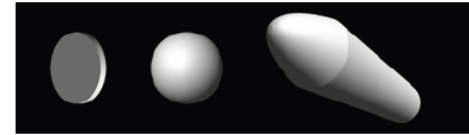
Sphere



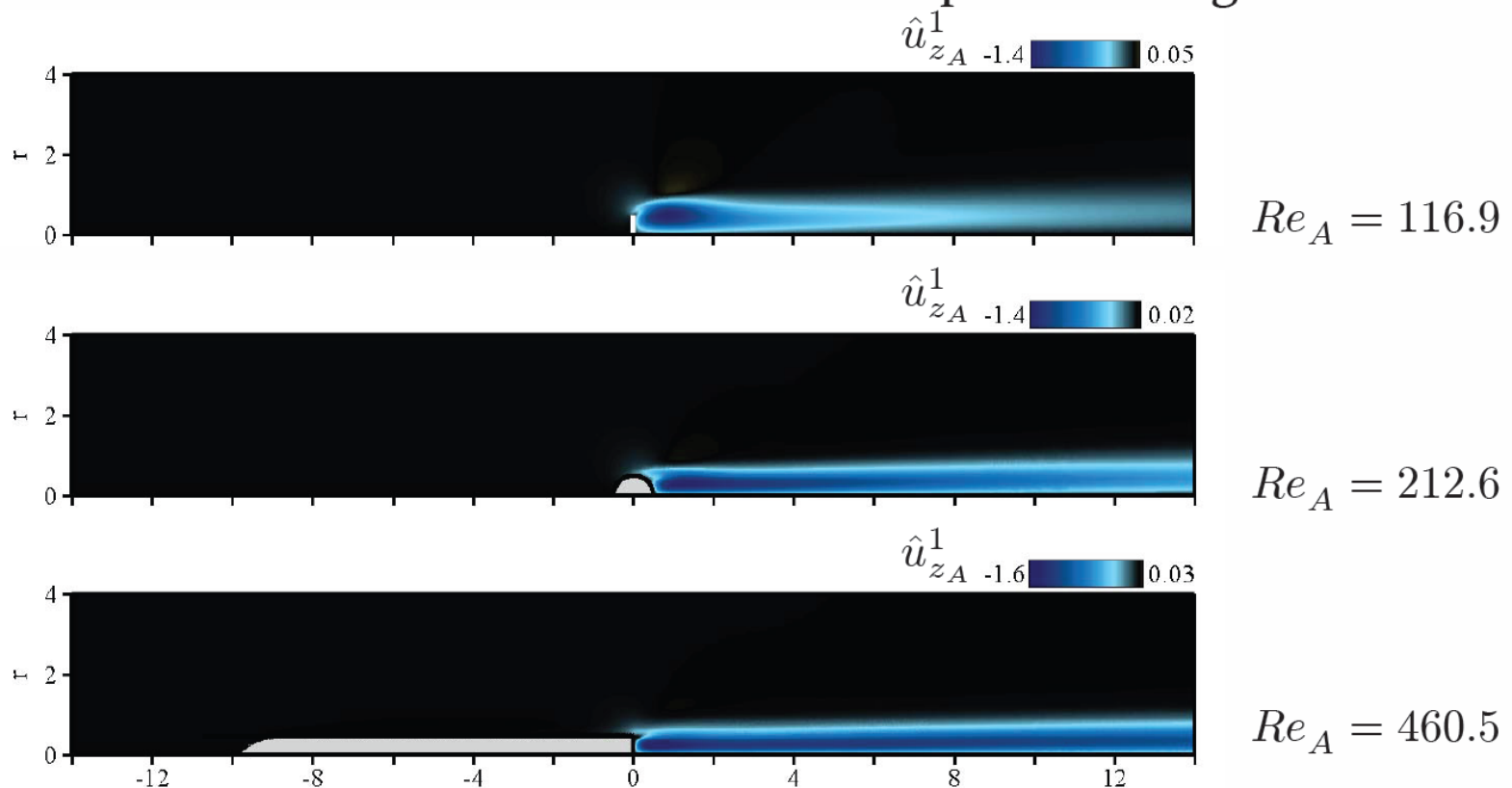
Blunt base

Meliga et al. 2008,2009,2010

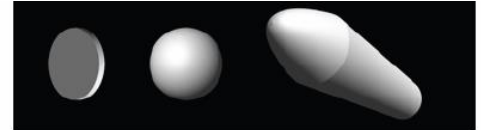
Leading global modes



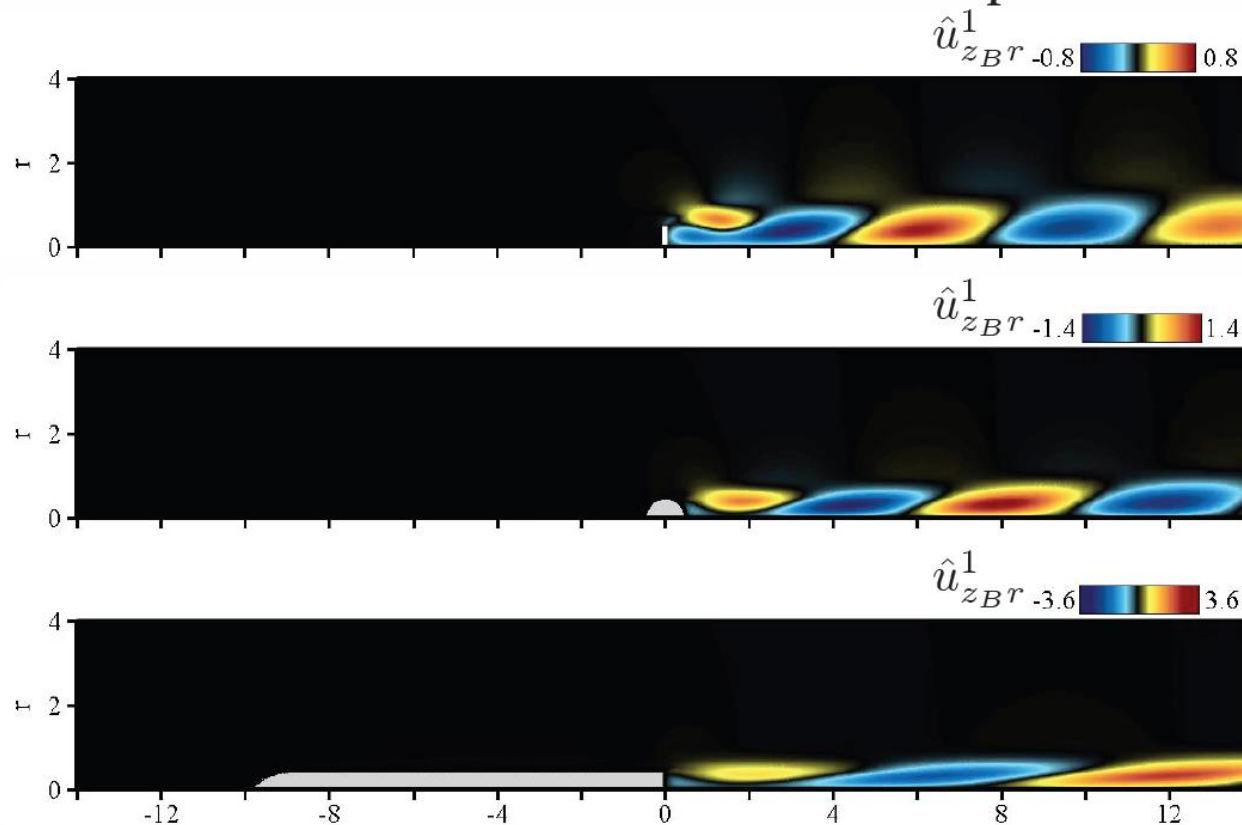
Same bifurcation sequence for all prototype flows,
in the **INCOMPRESSIBLE** & compressible regimes



Leading global modes



Same bifurcation sequence for all prototype flows,
in the **INCOMPRESSIBLE** & compressible regimes



$Re_B = 125.8$

$St = 0.12$

$Re_B = 280.7$

$St = 0.11$

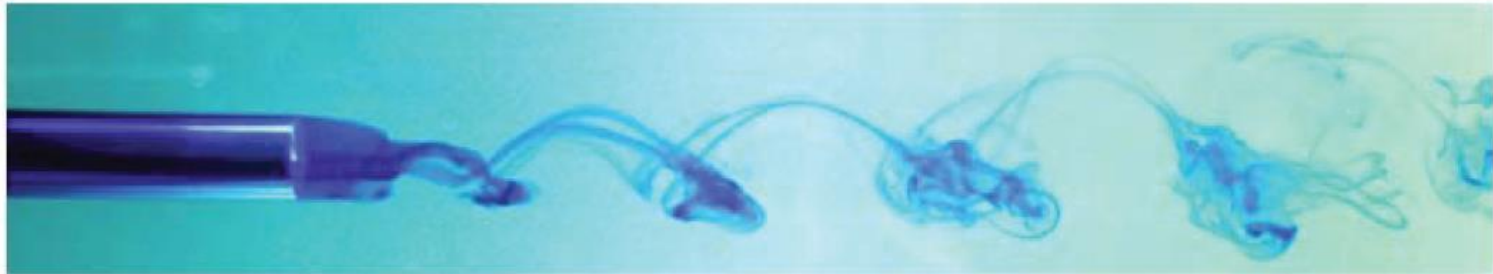
$Re_B = 909.1$

$St = 0.07$

How do these theoretical considerations compare with experiments



Sphere – Johnson & Patel (1999)

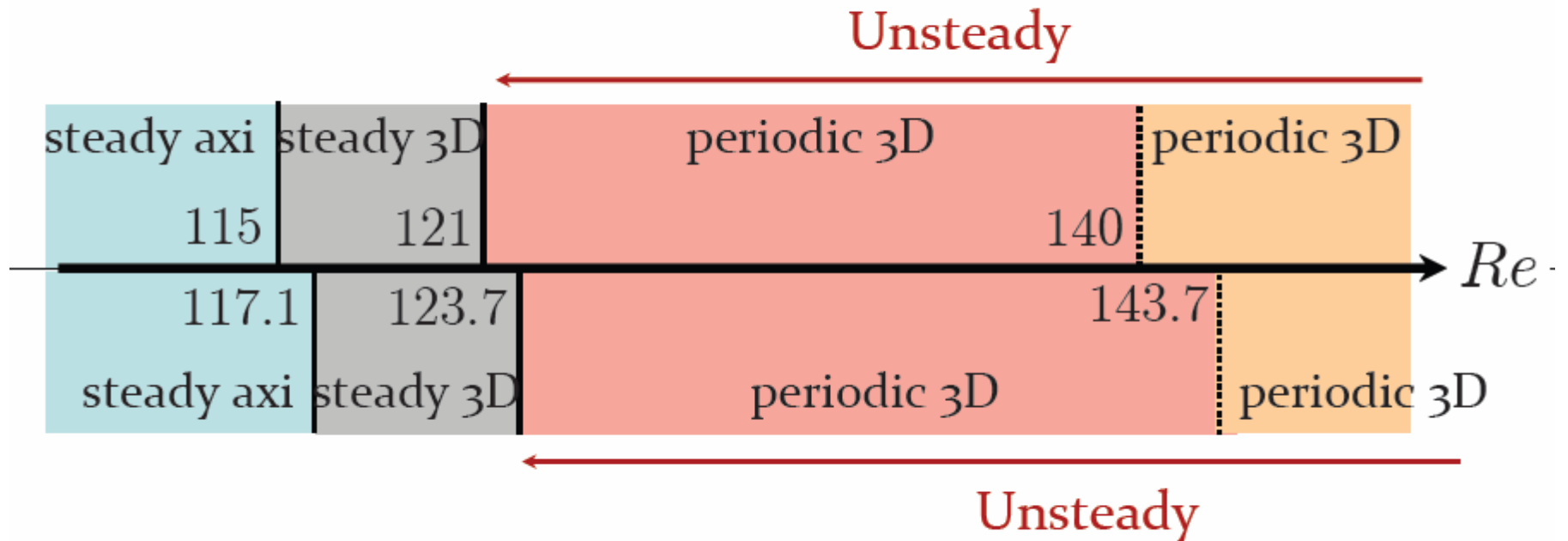


Blunt base – Siegel et al. (2008)

Remarkable predictions of successive symmetry breakings and thresholds

Direct numerical simulation of the 3D flow

Fabre et al. (2008)

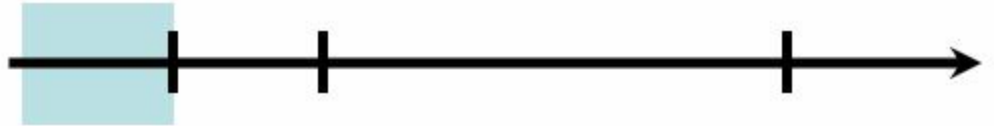


Weakly nonlinear global stability

Meliga et al. 2009

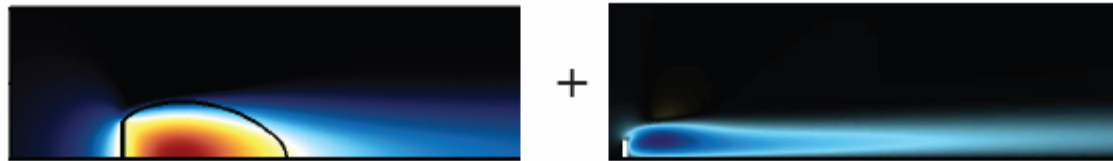
Stable solutions

- Steady axi



Stable solutions

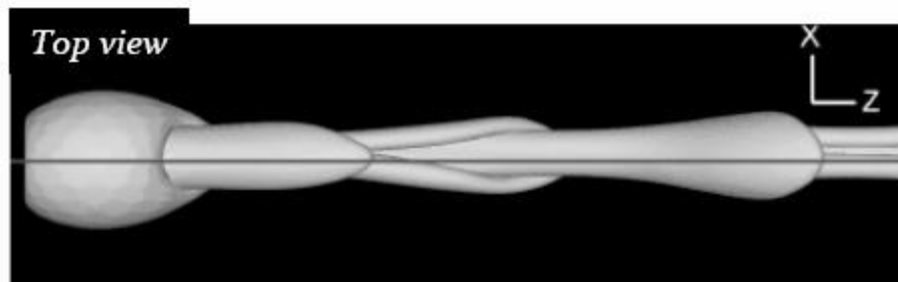
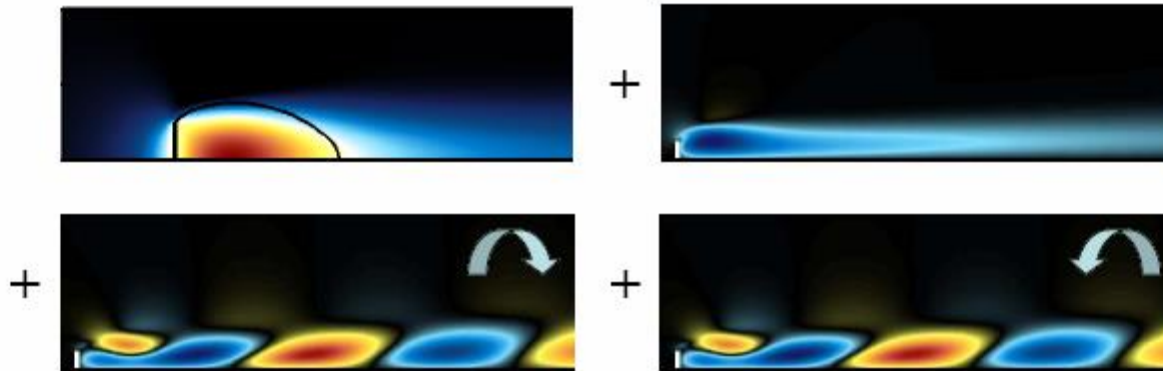
- Steady 3D + planar sym.



Numerical dye lines

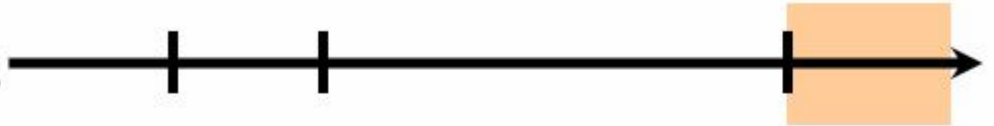
Stable solutions

- Periodic 3D, no sym.



Stable solutions

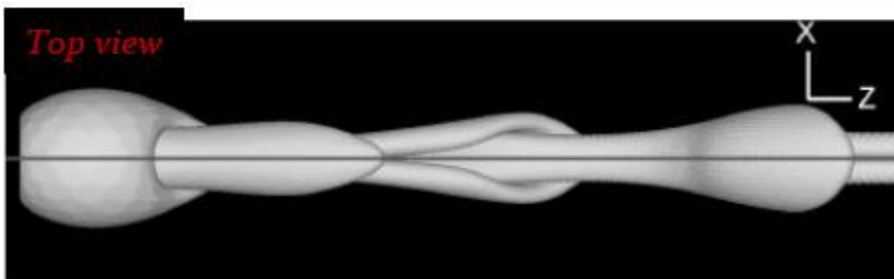
- Periodic 3D + planar sym.



+



+



What about globally stable flows but locally unstable flows?

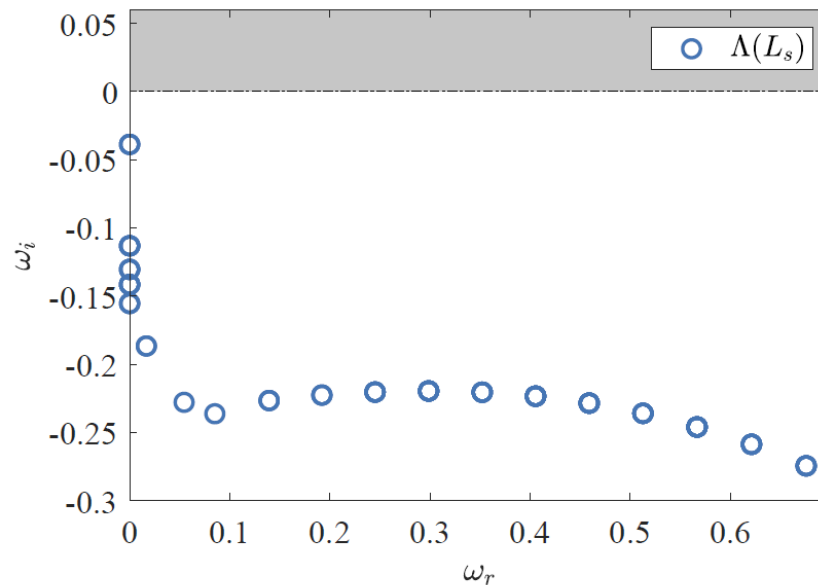
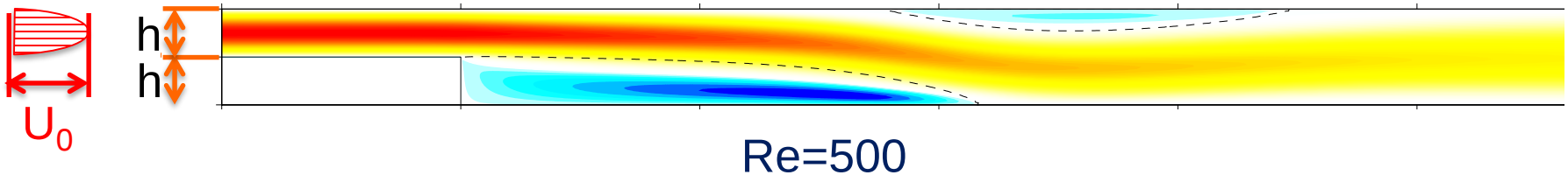
Amplifier flows

Linearly **globally stable**

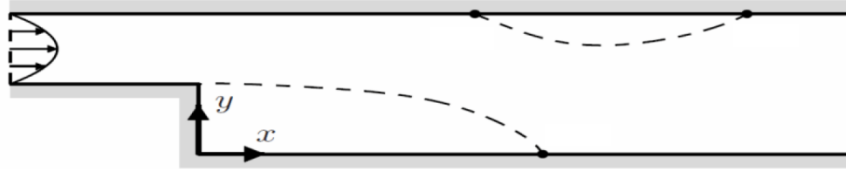
Amplification of external disturbances, irrespective of the eigenfrequency spectrum

Selective but broad band response

Examples: Jets, Separated boundary layers,...



Amplifier flows



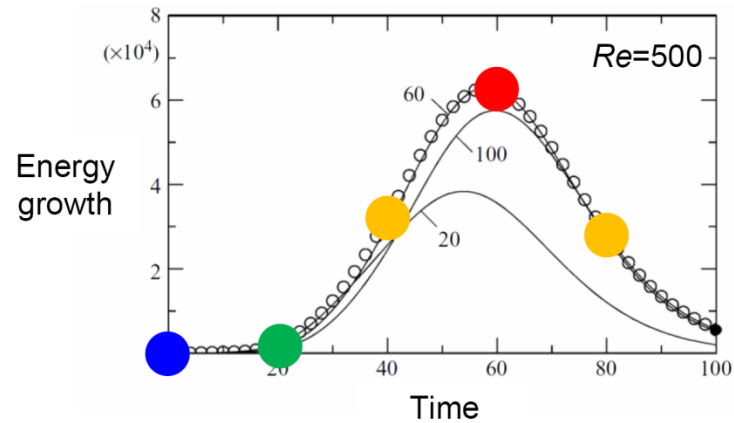
Response to initial disturbance

Response to sustained harmonic forcing

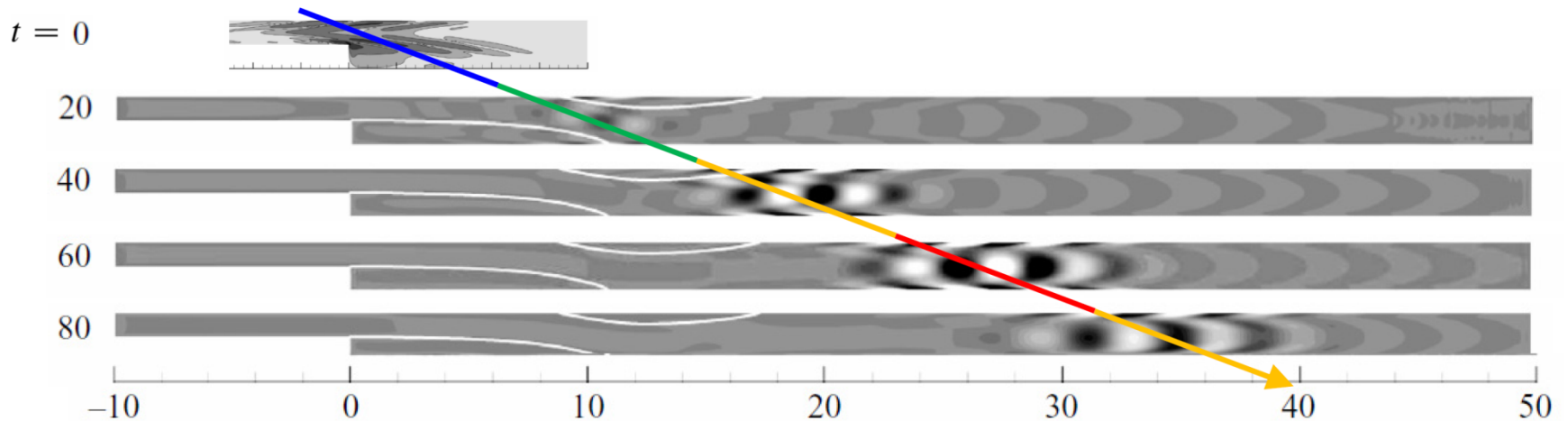
Response to stochastic forcing

Large transient growth

[Blackburn, Barkley & Sherwin, 2008]



Growth in space & time: optimal perturbations amplified while convected downstream

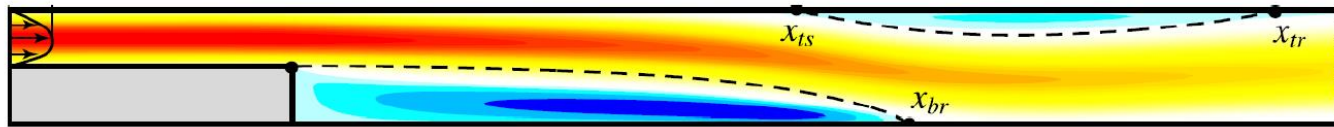


Backward facing step flow

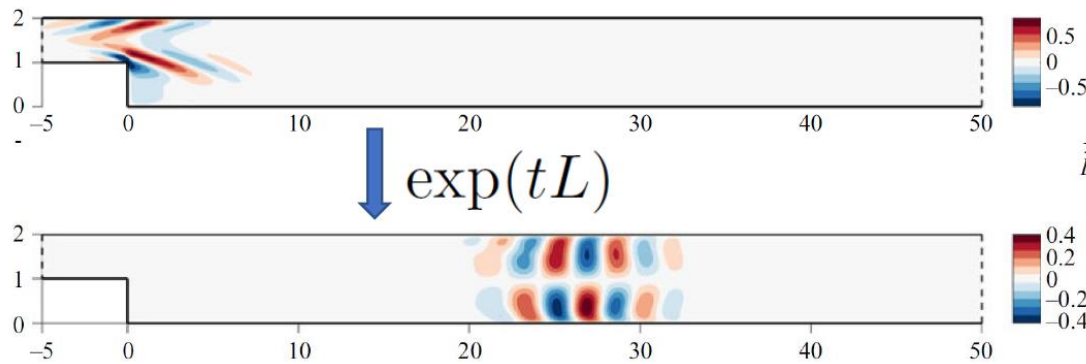
$$\frac{\partial \mathbf{u}}{\partial t} = L\mathbf{u} + \mathbf{f}, \quad \mathbf{u}(0) \neq \mathbf{0} \quad \text{and} \quad LL^\dagger \neq L^\dagger L$$

Transient growth:

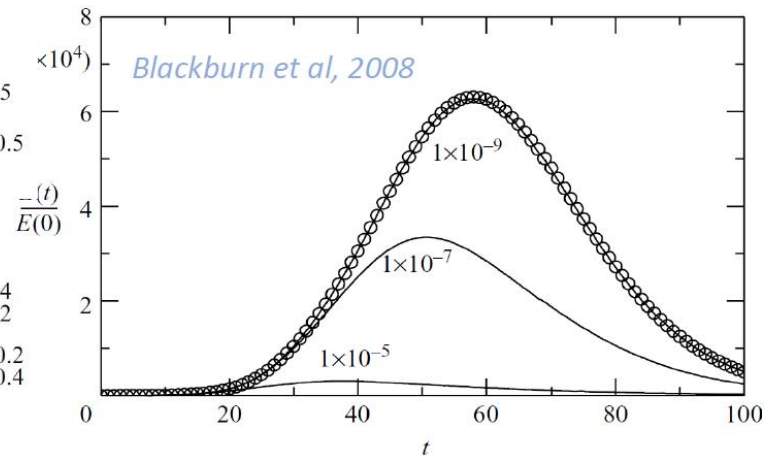
Re=500



initial condition



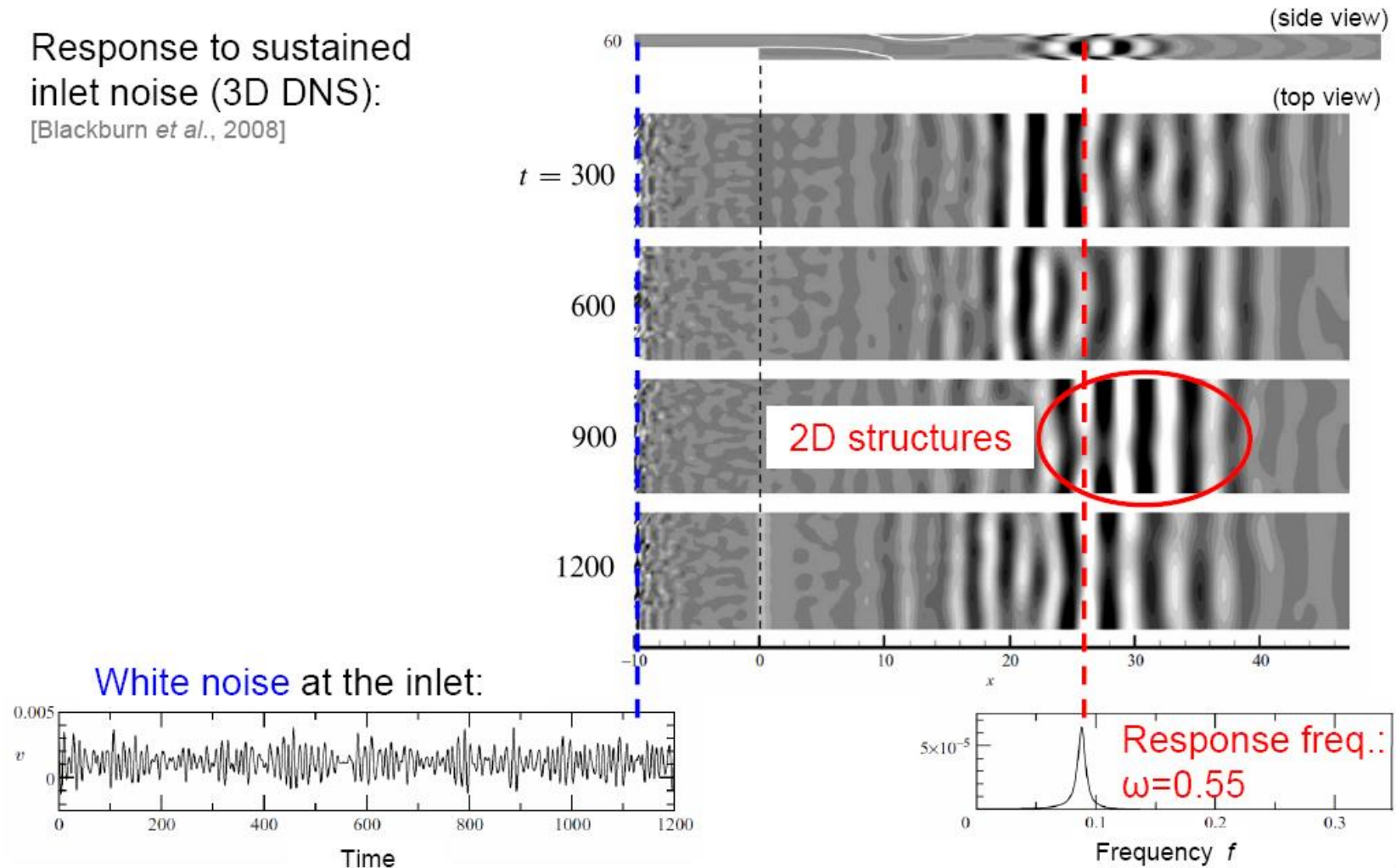
amplified solution at t=58



Response to white noise

Response to sustained
inlet noise (3D DNS):

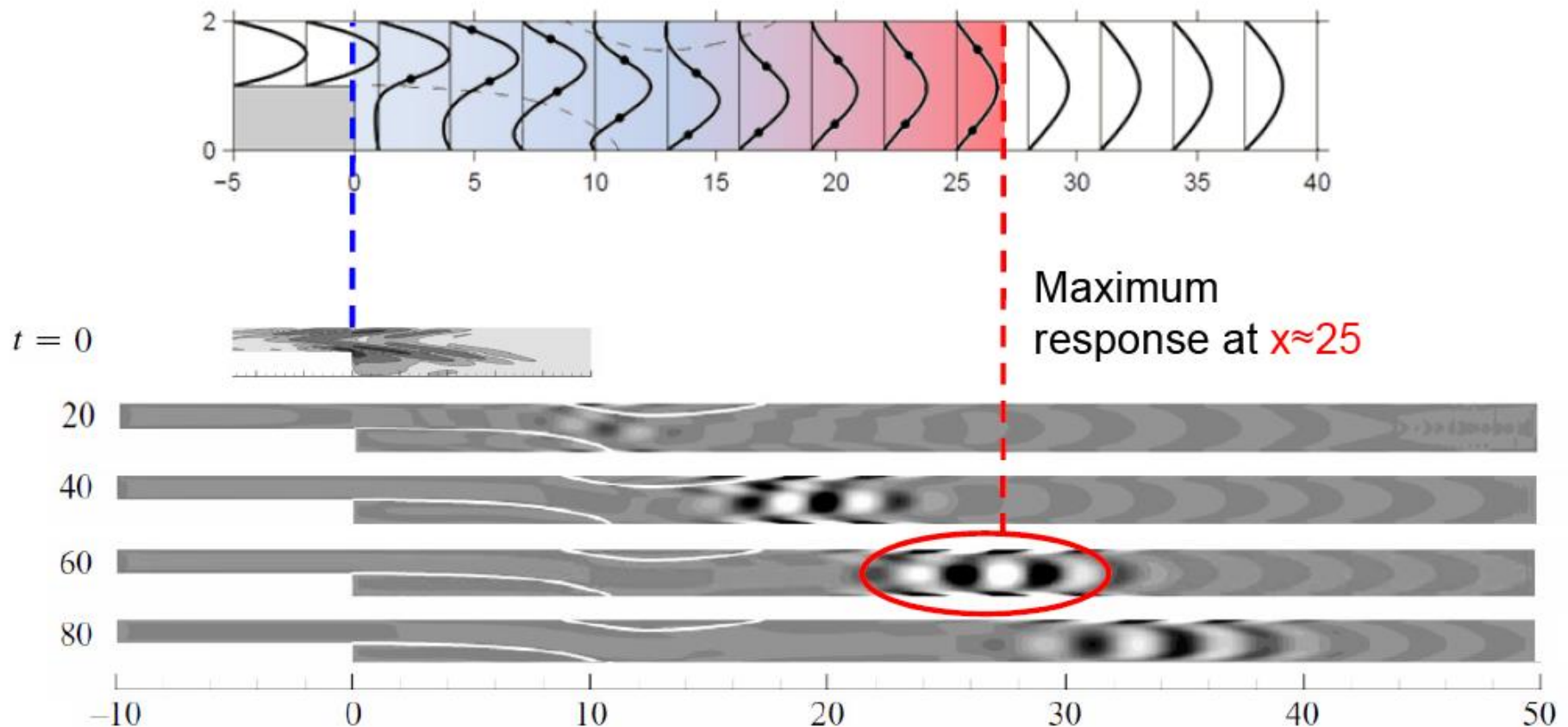
[Blackburn *et al.*, 2008]



⇒ Very selective amplifier

The global transient growth is due to the local spatial growth

Amplification over the
region of **local instability**



The global transient growth is due to the local spatial growth

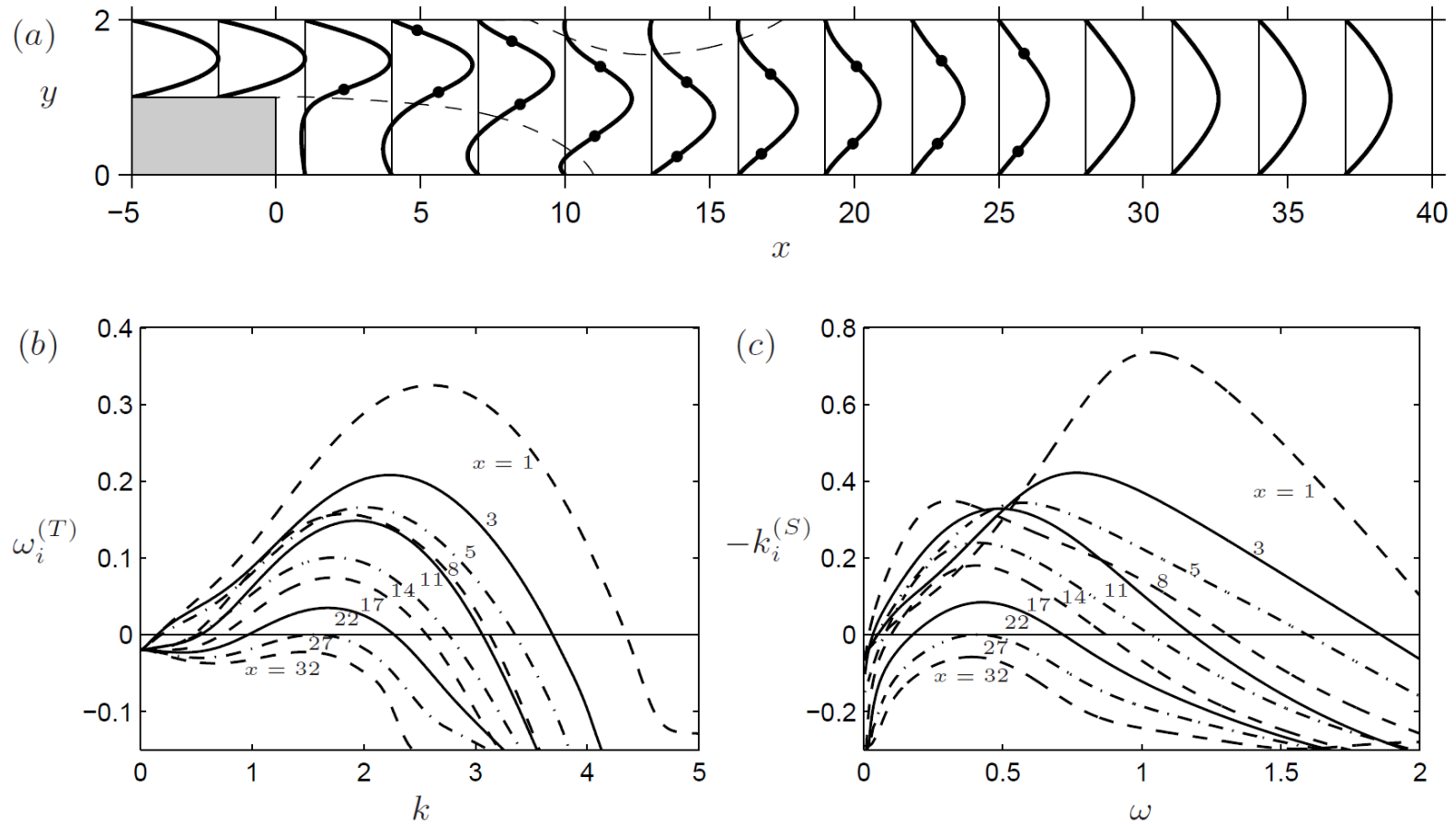
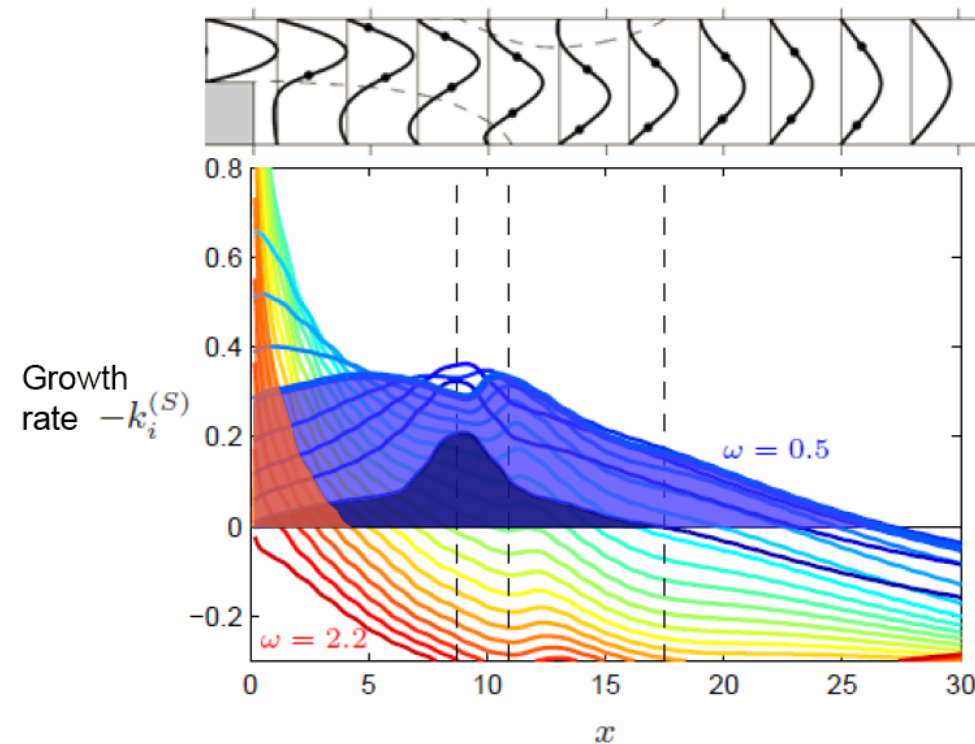


FIGURE 7. (a) Profiles of streamwise velocity for $\Gamma = 0.5$, $Re = 500$, with inflexion points shown as dots. (b) Temporal and (c) spatial growth rates obtained from local stability analysis.

Method 1: weakly non parallel spatial stability analysis

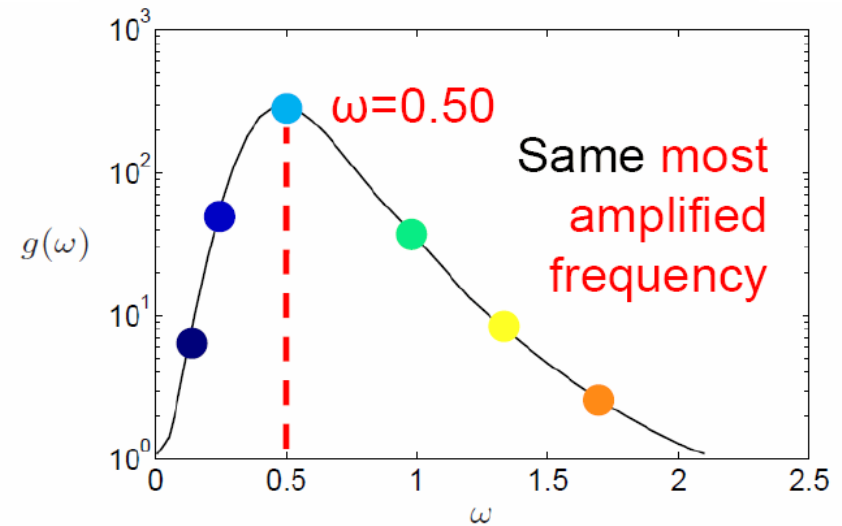
Parallel flow: $\mathbf{u}'(x, y, t) = \mathbf{u}(y) e^{i(kx - \omega t)}$

Spatial analysis $\omega \in \mathbb{R}, k \in \mathbb{C}$



Overall amplification: integral gain

$$g(\omega) = \exp \left(\int -k_i^{(S)}(\omega, x) dx \right)$$



Method 2: transfer function

Nonlinear problem (base flow)

$$\nabla \cdot \mathbf{U}_b = 0,$$

$$\mathbf{U}_b \cdot \nabla \mathbf{U}_b + \nabla P_b - Re^{-1} \nabla^2 \mathbf{U}_b = \mathbf{0},$$

+ linear perturbation

$$\nabla \cdot \mathbf{u}' = 0,$$

$$\partial_t \mathbf{u}' + \nabla \mathbf{u}' \cdot \mathbf{U}_b + \nabla \mathbf{U}_b \cdot \mathbf{u}' + \nabla p' - Re^{-1} \nabla^2 \mathbf{u}' = \mathbf{F}$$

Method 2: transfer function

Nonlinear problem (base flow)

$$\nabla \cdot \mathbf{U}_b = 0,$$

$$\mathbf{U}_b \cdot \nabla \mathbf{U}_b + \nabla P_b - Re^{-1} \nabla^2 \mathbf{U}_b = \mathbf{0},$$

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$$\nabla \cdot \mathbf{u}' = 0,$$

$$\partial_t \mathbf{u}' + \nabla \mathbf{u}' \cdot \mathbf{U}_b + \nabla \mathbf{U}_b \cdot \mathbf{u}' + \nabla p' - Re^{-1} \nabla^2 \mathbf{u}' = \mathbf{F}$$

for harmonic forcing $\mathbf{F}(t) = \mathbf{f} e^{i\omega t}$

look for solution $(\mathbf{u}', p')(\mathbf{x}, t) = (\mathbf{u}, p)(\mathbf{x}) e^{i\omega t}$

Method 2: transfer function

harmonic forcing

$$\nabla \cdot \mathbf{u} = 0,$$

$$i\omega \mathbf{u} + \nabla \mathbf{u} \cdot \mathbf{U}_b + \nabla \mathbf{U}_b \cdot \mathbf{u} + \nabla p - Re^{-1} \nabla^2 \mathbf{u} = \mathbf{f}$$

Resolvent operator (transfer function)

$$\mathbf{u} = \mathcal{R}(\omega) \mathbf{f}$$

Method 2: transfer function

harmonic forcing

$$\nabla \cdot \mathbf{u} = 0,$$

$$i\omega \mathbf{u} + \nabla \mathbf{u} \cdot \mathbf{U}_b + \nabla \mathbf{U}_b \cdot \mathbf{u} + \nabla p - Re^{-1} \nabla^2 \mathbf{u} = \mathbf{f}$$

Resolvent operator (transfer function)

$$\mathbf{u} = \mathcal{R}(\omega) \mathbf{f}$$

Harmonic gain

$$G^2(\omega) = \frac{||\mathbf{u}||^2}{||\mathbf{f}||^2} = \frac{(\mathcal{R}\mathbf{f} | \mathcal{R}\mathbf{f})}{(\mathbf{f} | \mathbf{f})} = \frac{(\mathcal{R}^\dagger \mathcal{R} \mathbf{f} | \mathbf{f})}{(\mathbf{f} | \mathbf{f})}$$

Optimal response to harmonic forcing

$$\frac{\partial \mathbf{u}}{\partial t} = L\mathbf{u} + \mathbf{f} \quad \text{insert:} \quad \begin{cases} \mathbf{f}(x, t) = \hat{\mathbf{f}}(x)e^{i\omega_o t} + \text{c.c} \\ \mathbf{u}(x, t) = \hat{\mathbf{u}}(x)e^{i\omega_o t} + \text{c.c} \end{cases} \quad \text{to obtain:} \quad \hat{\mathbf{u}} = (i\omega_o I - L)^{-1} \hat{\mathbf{f}} = R(i\omega_o) \hat{\mathbf{f}}$$

$$\text{optimize:} \quad G(i\omega_o) = \max_{\hat{\mathbf{f}}} \frac{\|\hat{\mathbf{u}}\|}{\|\hat{\mathbf{f}}\|} = \|R(i\omega_o)\| = \frac{1}{\epsilon_o} \quad \text{under the scalar product} \quad \langle \hat{\mathbf{u}}_a | \hat{\mathbf{u}}_b \rangle = \int_{\Omega} \hat{\mathbf{u}}_a^H \hat{\mathbf{u}}_b d\Omega$$

Singular value decomposition:

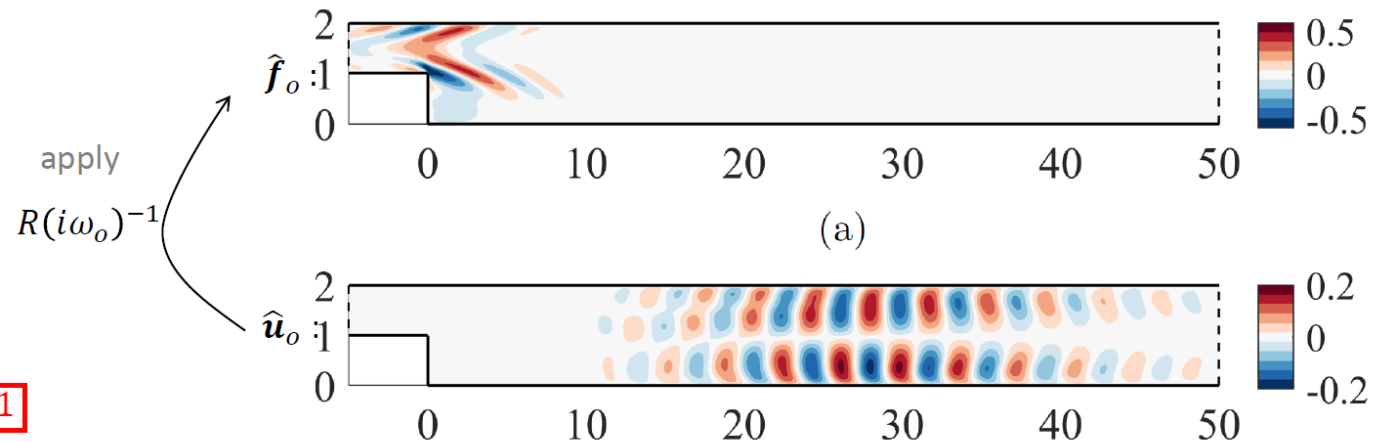
$$R(i\omega_o)^{-1} \hat{\mathbf{u}}_o = \epsilon_o \hat{\mathbf{f}}_o$$

With normalization

$$\langle \hat{\mathbf{u}}_o | \hat{\mathbf{u}}_o \rangle = \|\hat{\mathbf{u}}_o\|^2 = 1$$

$$\|\hat{\mathbf{f}}_o\| = 1$$

Strong nonnormality: $\epsilon_o \ll 1$



Optimal response to harmonic forcing

Bounds of resolvent norm

- **Diagonalize** the system matrix L

$$L = S\Lambda S^{-1}$$

Eigenvalue decomposition

S : Column eigenvector

Λ : Diagonal eigenvalues

$$\frac{1}{\text{dist}\{i\omega, \Lambda\}} \leq \|(i\omega - L)^{-1}\| = \|S(i\omega - \Lambda)^{-1}S^{-1}\| \leq \kappa(S) \frac{1}{\text{dist}\{i\omega, \Lambda\}}$$

$$\kappa(S) \gg 1$$

Non-Normal system: upper and lower bound differ

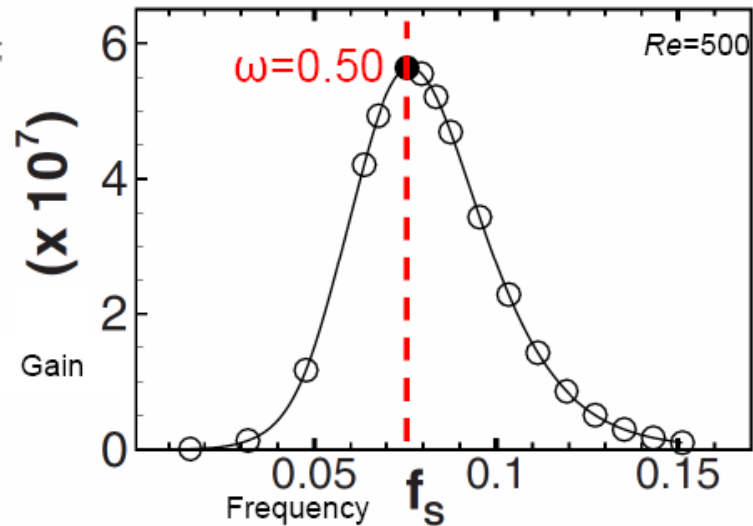
we can have a pseudo-resonance

Strong amplification also far from system eigenfrequency

Method 2: transfer function

Optimal harmonic gain [Marquet & Sipp, 2010;
Marquet *et al.*, 2010, priv. comm.]:

same preferred frequency,
large gain



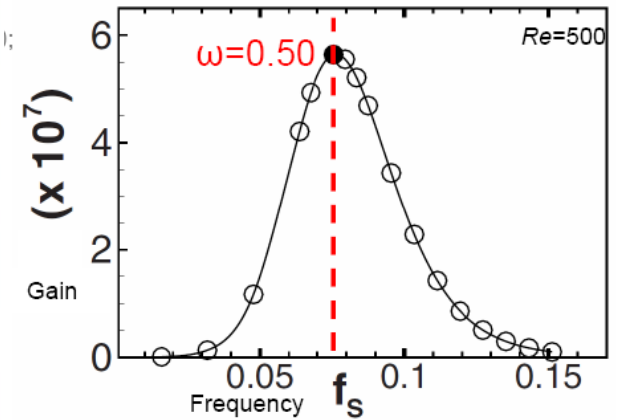
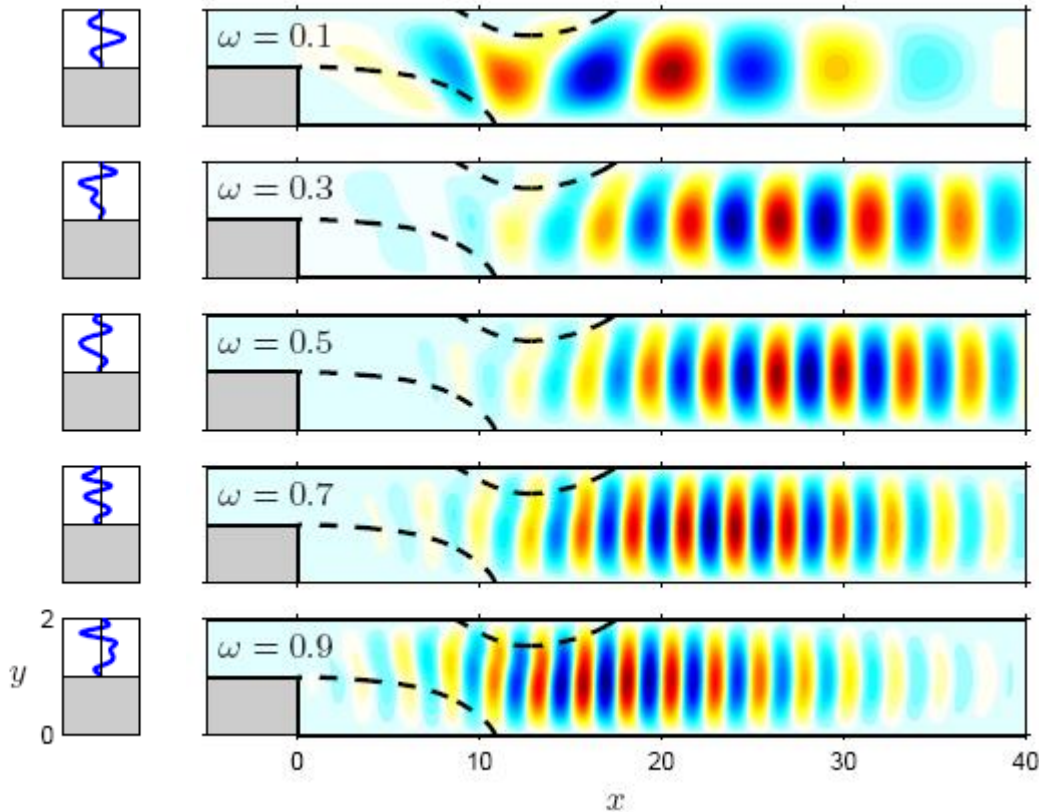
Optimal forcing



and optimal response very similar to transient growth:



Method 2: transfer function



Local weakly non parallel spatial stability analysis v.s. global resolvent analysis

